

Aerodynamics & Flight Mechanics Research Group

The Rotor in Axial Flight

S. J. Newman

Technical Report AFM-11/02

January 2011

UNIVERSITY OF SOUTHAMPTON
SCHOOL OF ENGINEERING SCIENCES
AERODYNAMICS AND FLIGHT MECHANICS RESEARCH GROUP

The Rotor in Axial Flight

by

S. J. Newman

AFM Report No. AFM 11/02

January 2011

© School of Engineering Sciences, Aerodynamics and Flight Mechanics Research Group



COPYRIGHT NOTICE

(c) SES University of Southampton All rights reserved.

SES authorises you to view and download this document for your personal, non-commercial use. This authorization is not a transfer of title in the document and copies of the document and is subject to the following restrictions: 1) you must retain, on all copies of the document downloaded, all copyright and other proprietary notices contained in the Materials; 2) you may not modify the document in any way or reproduce or publicly display, perform, or distribute or otherwise use it for any public or commercial purpose; and 3) you must not transfer the document to any other person unless you give them notice of, and they agree to accept, the obligations arising under these terms and conditions of use. This document, is protected by worldwide copyright laws and treaty provisions.



Introduction

The following analysis is an appraisal of Actuator Disc Theory and Blade Element Theory applied to a rotor in axial flight. The solutions to the methods require scrutiny to choose which is correct for the particular flight condition. The report discusses the manner in which the solutions interact and proposes a scheme to assemble them together in a unified whole.

The analysis is then extended to Annulus Theory.

Nomenclature

T	Thrust
R	Rotor Radius
V_T	Tip Speed
V_C	Climb Velocity
V_D	Descent Velocity
V_i	Rotor Downwash Velocity
N	No of Blades
C	Blade Chord (Constant)
ρ	Air Density
a	Lift Curve Slope
s	Rotor Solidity
θ_0	Collective Pitch
κ	Overall Blade Twist (Linear)
x	Non- Dimensional Spanwise Variable
μ_z	Vertical Advance Ratio
λ_i	Non Dimensional Rotor Downwash
C_T	Thrust Coefficient
BET	Blade Element Theory
ADT	Actuator Disc Theory
AT	Annulus Theory



Diagrams

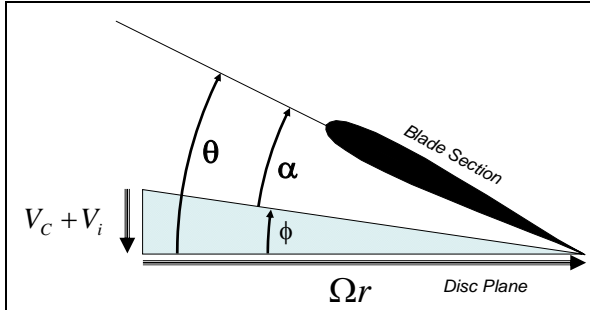


Diagram 1 - BET for Climb

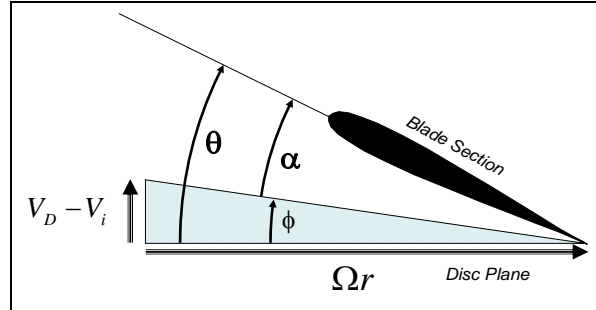


Diagram 2 - BET for Descent

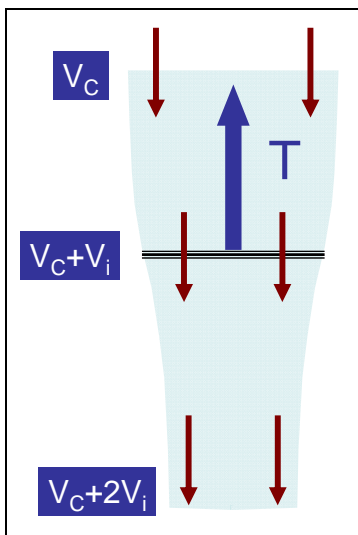


Diagram 3 - ADT for Climb

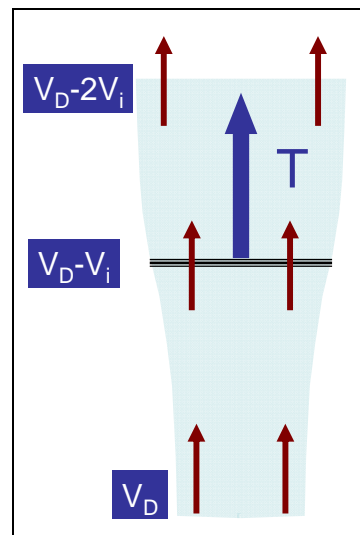


Diagram 4 - ADT for Descent

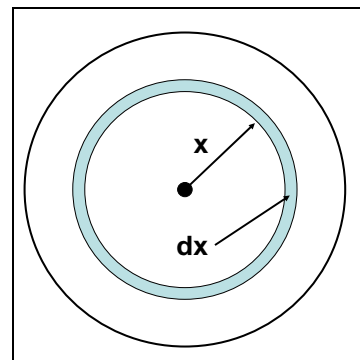
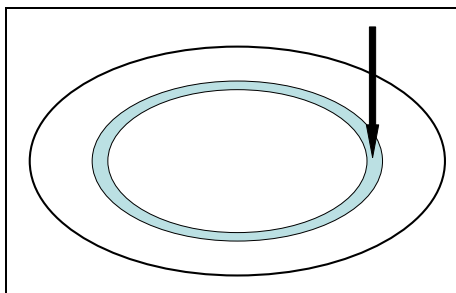


Diagram 5 - Annulus Theory



Blade Element Theory with Momentum Theory

Climb Root

BET gives:

$$T = \frac{1}{2} \rho V_T^2 \cdot NcR \cdot a \int_0^1 x^2 \left(\theta_0 - \kappa x - \frac{\mu_z + \lambda_i}{x} \right) \cdot dx \quad (1.)$$

which on normalisation becomes:

$$\begin{aligned} \frac{C_T}{sa} &= \frac{\theta_0}{3} - \frac{\kappa}{4} - \frac{\mu_z + \lambda_i}{2} \\ &= \frac{\theta_{75}}{3} - \frac{\mu_z + \lambda_i}{2} \end{aligned} \quad (2.)$$

ADT gives the following expression for rotor thrust:

$$\begin{aligned} T &= \rho A (V_z + V_i) \cdot 2V_i \\ &= 2\rho A V_T^2 (\mu_z + \lambda_i) \lambda_i \end{aligned} \quad (3.)$$

which on normalisation becomes:



$$C_T = 4(\mu_z + \lambda_i)\lambda_i$$

$$\frac{C_T}{4} = \mu_z \lambda_i + \lambda_i^2 \quad (4.)$$

Combining equations (2) & (4) results in the quadratic equation:

$$\lambda_i^2 + \lambda_i \left\{ \mu_z + \frac{sa}{8} \right\} - \frac{sa}{4} \left\{ \frac{\theta_{75}}{3} - \frac{\mu_z}{2} \right\} = 0 \quad (5.)$$

i.e.:

$$A\lambda_i^2 + \lambda_i B + C = 0 \quad (6.)$$

where the coefficients are defined as:

$$A = 1$$

$$B = \mu_z + \frac{sa}{8} \quad (7.)$$

$$C = -\frac{sa}{4} \left\{ \frac{\theta_{75}}{3} - \frac{\mu_z}{2} \right\}$$

if we introduce the product sa as a basic normalisation variable, the above equations reduce to:



$$\begin{aligned}
 \overline{\mu_z} &= \frac{\mu_z}{sa} \\
 \overline{\lambda_i} &= \frac{\lambda_i}{sa} \\
 \overline{\theta_{75}} &= \frac{\theta_{75}}{sa} \\
 \overline{C_T} &= \frac{C_T}{(sa)^2} \\
 &= \frac{\overline{\theta_{75}}}{3} - \frac{\overline{\mu_z} + \overline{\lambda_i}}{2}
 \end{aligned} \tag{8.}$$

$$\overline{\lambda_i}^2 + \overline{\lambda_i} \left\{ \overline{\mu_z} + \frac{1}{8} \right\} - \frac{1}{4} \left\{ \frac{\overline{\theta_{75}}}{3} - \frac{\overline{\mu_z}}{2} \right\} = 0 \tag{9.}$$

$$\overline{A} \overline{\lambda_i}^2 + \overline{\lambda_i} \overline{B} + \overline{C} = 0 \tag{10.}$$

$$\begin{aligned}
 \overline{A} &= 1 \\
 \overline{B} &= \overline{\mu_z} + \frac{1}{8} \\
 \overline{C} &= -\frac{1}{4} \left\{ \frac{\overline{\theta_{75}}}{3} - \frac{\overline{\mu_z}}{2} \right\}
 \end{aligned} \tag{11.}$$

It is now possible to solve this problem within a domain where rotor scale factors have been completely removed.



The solutions of (10) & (11) are:

$$\begin{aligned}\bar{\lambda}_i &= \frac{-\left(\bar{\mu}_z + \frac{1}{8}\right) \pm \sqrt{\left(\bar{\mu}_z + \frac{1}{8}\right)^2 + \left\{\frac{\bar{\theta}_{75}}{3} - \frac{\bar{\mu}_z}{2}\right\}}}{2} \\ &= \frac{-\left(\bar{\mu}_z + \frac{1}{8}\right) \pm \sqrt{\left(\bar{\mu}_z - \frac{1}{8}\right)^2 + \frac{\bar{\theta}_{75}}{3}}}{2}\end{aligned}\quad (12.)$$

These are denoted **C+** & **C-**

In order for either solution to exist, the following condition must be met:

$$\bar{B}^2 - 4\bar{A}\bar{C} \geq 0 \quad (13.)$$

i.e.:

$$\left(\bar{\mu}_z - \frac{1}{8}\right)^2 + \frac{\bar{\theta}_{75}}{3} \geq 0 \quad (14.)$$

which is a parabolic region defined by:

$$\bar{\theta}_{75} = -3 \cdot \left(\bar{\mu}_z - \frac{1}{8}\right)^2 \quad (15.)$$

Provided that this region is avoided, the remaining part of the domain will give rise to a positive or negative solution.



For a positive solution we have from inspection of (12):

C+

$$\left(\bar{\mu}_z + \frac{1}{8}\right) \leq 0 \tag{16.}$$

$$\bar{\mu}_z \leq -\frac{1}{8}$$

OR

$$\left| \sqrt{\left(\bar{\mu}_z - \frac{1}{8}\right)^2 + \frac{\bar{\theta}_{75}}{3}} \right| \geq \left| \bar{\mu}_z + \frac{1}{8} \right| \tag{17.}$$

which on squaring out gives:

$$\bar{\theta}_{75} \geq \frac{3}{2} \bar{\mu}_z \tag{18.}$$

the domain now sub-divides as shown in Figure 1:



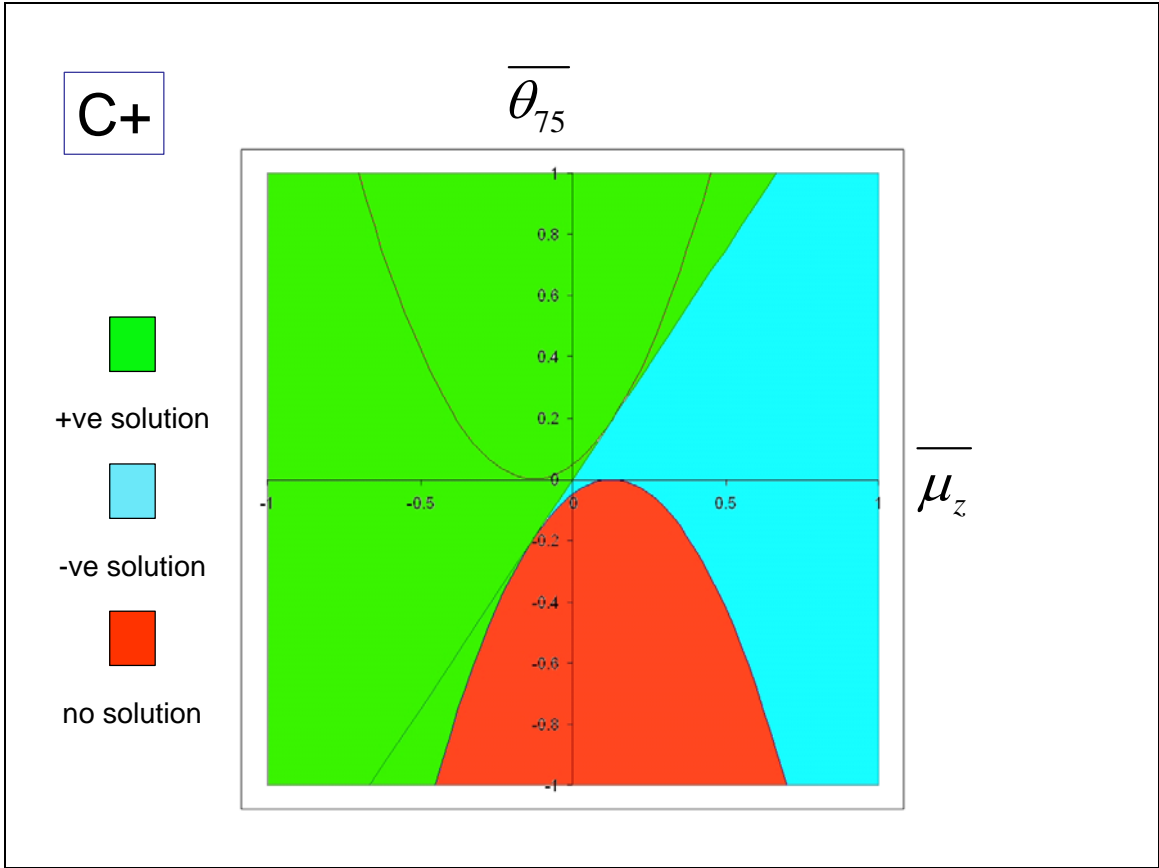


Figure 1 - C+ Solution Space

C-

a positive root will be obtained if:

$$\left(\overline{\mu}_z + \frac{1}{8} \right) \leq 0 \tag{19.}$$

$$\overline{\mu}_z \leq -\frac{1}{8}$$

AND



$$\left| \sqrt{\left(\overline{\mu}_z - \frac{1}{8}\right)^2 + \frac{\overline{\theta}_{75}}{3}} \right| \leq \left| \overline{\mu}_z + \frac{1}{8} \right| \tag{20.}$$

which on squaring out gives:

$$\overline{\theta}_{75} \leq \frac{3}{2} \overline{\mu}_z \tag{21.}$$

the domain now sub-divides as shown in Figure 2:

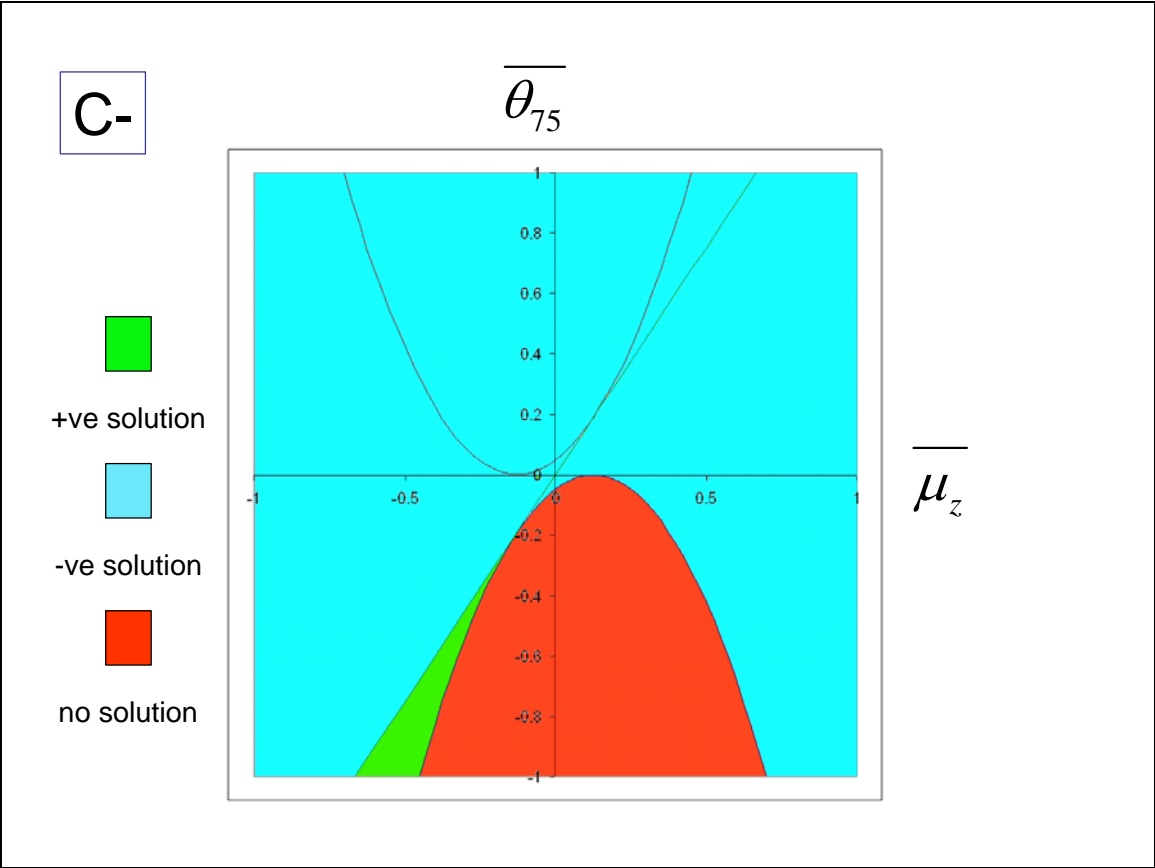


Figure 2 – C- Solution Space



Descent Root

BET gives:

$$T = \frac{1}{2} \rho V_T^2 \cdot NcR \cdot a \int_0^1 x^2 \left(\theta_0 - \kappa x - \frac{\mu_z + \lambda_i}{x} \right) dx \quad (22.)$$

which on normalisation becomes:

$$\begin{aligned} \frac{C_T}{sa} &= \frac{\theta_0}{3} - \frac{\kappa}{4} - \frac{\mu_z + \lambda_i}{2} \\ &= \frac{\theta_{75}}{3} - \frac{\mu_z + \lambda_i}{2} \end{aligned} \quad (23.)$$

ADT gives the following expression for rotor thrust:

$$\begin{aligned} T &= \rho A (-V_z - V_i) \cdot 2V_i \\ &= 2\rho A V_T^2 (-\mu_z - \lambda_i) \lambda_i \end{aligned} \quad (24.)$$

which on normalisation becomes:

$$\begin{aligned} C_T &= -4(\mu_z + \lambda_i) \lambda_i \\ \frac{C_T}{4} &= -\mu_z \lambda_i - \lambda_i^2 \end{aligned} \quad (25.)$$



Equations (23) & (25) combine to give:

$$\lambda_i^2 + \lambda_i \left\{ \mu_z - \frac{sa}{8} \right\} + \frac{sa}{4} \left\{ \frac{\theta_{75}}{3} - \frac{\mu_z}{2} \right\} = 0 \quad (26.)$$

i.e.

$$A\lambda_i^2 + \lambda_i B + C = 0 \quad (27.)$$

where the coefficients are defined by:

$$A = 1$$

$$B = \mu_z - \frac{sa}{8} \quad (28.)$$

$$C = \frac{sa}{4} \left\{ \frac{\theta_{75}}{3} - \frac{\mu_z}{2} \right\}$$

normalisation using sa gives:

$$\bar{\lambda}_i^2 + \bar{\lambda}_i \left\{ \bar{\mu}_z - \frac{1}{8} \right\} + \frac{1}{4} \left\{ \frac{\bar{\theta}_{75}}{3} - \frac{\bar{\mu}_z}{2} \right\} = 0 \quad (29.)$$

i.e.:

$$\bar{A}\bar{\lambda}_i^2 + \bar{\lambda}_i \bar{B} + \bar{C} = 0 \quad (30.)$$



where:

$$\begin{aligned}\bar{A} &= 1 \\ \bar{B} &= \bar{\mu}_z - \frac{1}{8} \\ \bar{C} &= +\frac{1}{4} \left\{ \frac{\bar{\theta}_{75}}{3} - \frac{\bar{\mu}_z}{2} \right\}\end{aligned}\tag{31.}$$

It is now possible to solve this problem within a domain where rotor scale factors have been completely removed.

The solutions of (30) & (31) are:

$$\begin{aligned}\bar{\lambda}_i &= \frac{-\left(\bar{\mu}_z - \frac{1}{8}\right) \pm \sqrt{\left(\bar{\mu}_z - \frac{1}{8}\right)^2 - \left\{ \frac{\bar{\theta}_{75}}{3} - \frac{\bar{\mu}_z}{2} \right\}}}{2} \\ &= \frac{-\left(\bar{\mu}_z - \frac{1}{8}\right) \pm \sqrt{\left(\bar{\mu}_z + \frac{1}{8}\right)^2 - \frac{\bar{\theta}_{75}}{3}}}{2}\end{aligned}\tag{32.}$$

These are denoted **D+** & **D-**

In order for a solution, the following condition must be met:

$$\bar{B}^2 - 4\bar{A}\bar{C} \geq 0\tag{33.}$$



i.e.:

$$\begin{aligned} \left(\bar{\mu}_z - \frac{1}{8}\right)^2 - \frac{\bar{\theta}_{75}}{3} + \frac{\bar{\mu}_z}{2} &\geq 0 \\ \left(\bar{\mu}_z + \frac{1}{8}\right)^2 - \frac{\bar{\theta}_{75}}{3} &\geq 0 \end{aligned} \quad (34.)$$

This is a parabolic region, but defined by:

$$\bar{\theta}_{75} = 3 \cdot \left(\bar{\mu}_z + \frac{1}{8}\right)^2 \quad (35.)$$

Provided that this region is avoided, the remaining part of the domain will give rise to a positive or negative solution.

For a positive solution we have from inspection of (32):

D+

$$\begin{aligned} \left(\bar{\mu}_z - \frac{1}{8}\right) &\leq 0 \\ \bar{\mu}_z &\leq \frac{1}{8} \end{aligned} \quad (36.)$$

OR



$$\left| \sqrt{\left(\overline{\mu}_z + \frac{1}{8}\right)^2 - \frac{\overline{\theta}_{75}}{3}} \right| \geq \left| \overline{\mu}_z - \frac{1}{8} \right| \quad (37.)$$

which on squaring out gives:

$$\overline{\theta}_{75} \leq \frac{3}{2} \overline{\mu}_z \quad (38.)$$

the domain now sub-divides as shown in Figure 3:

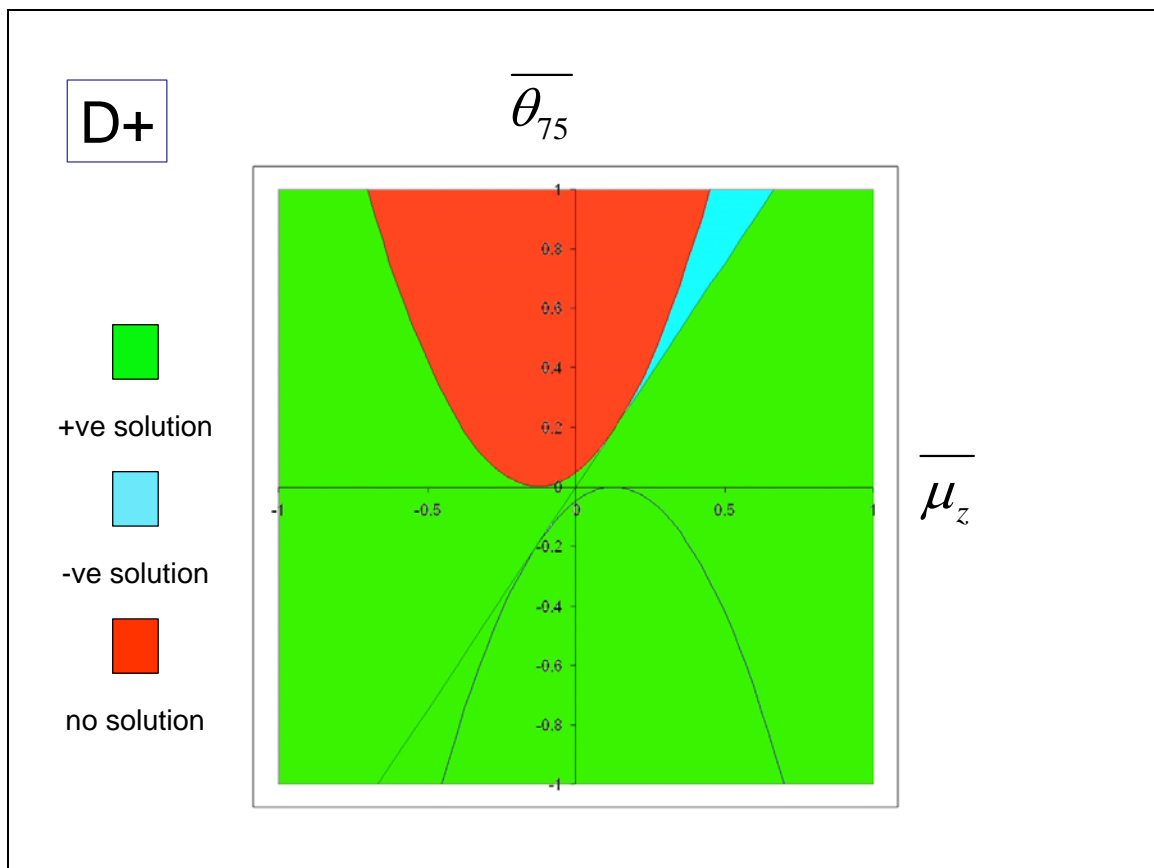


Figure 3 - D+ Solution Space



D-

a positive root will be obtained if:

$$\left(\bar{\mu}_z - \frac{1}{8}\right) \leq 0$$

$$\bar{\mu}_z \leq \frac{1}{8}$$
(39.)

AND

$$\left| \sqrt{\left(\bar{\mu}_z + \frac{1}{8}\right)^2 - \frac{\bar{\theta}_{75}}{3}} \right| \leq \left| \bar{\mu}_z - \frac{1}{8} \right|$$
(40.)

which on squaring out gives:

$$\bar{\theta}_{75} \geq \frac{3}{2} \bar{\mu}_z$$
(41.)

the domain now sub-divides as shown in Figure 4:



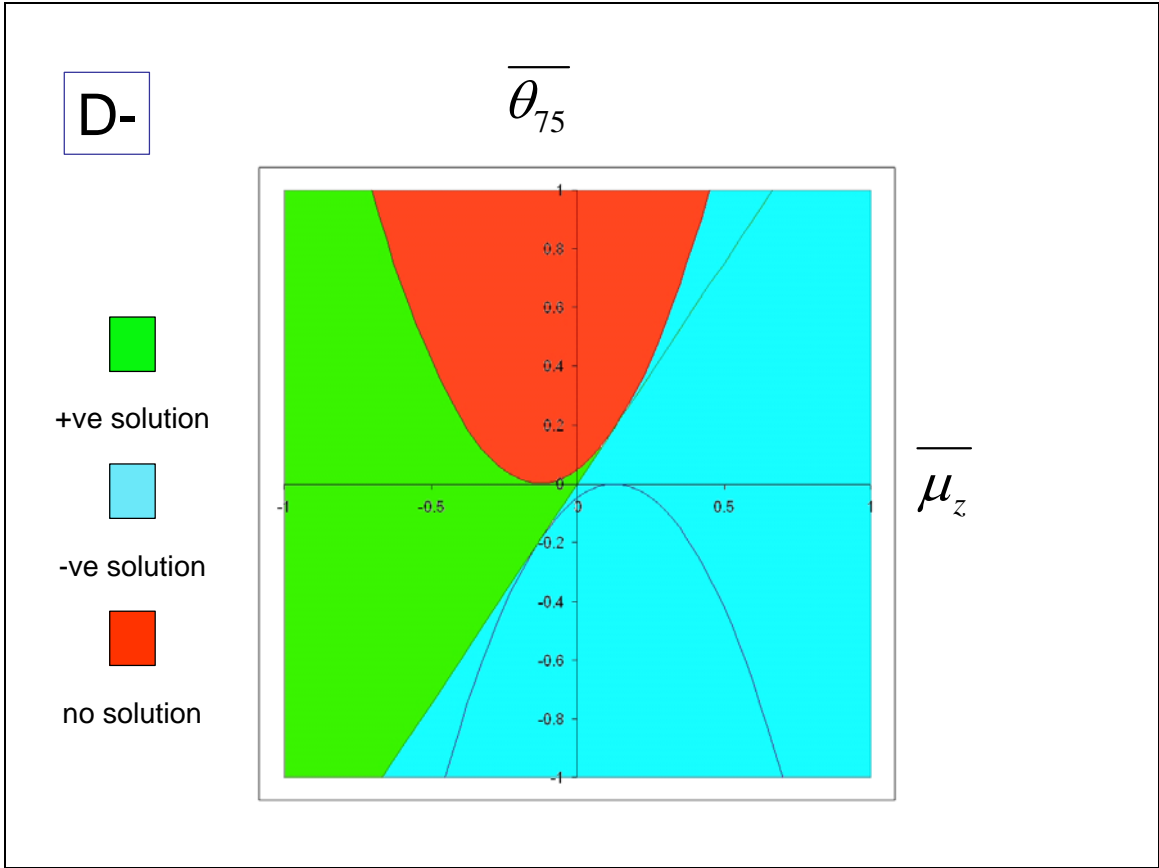


Figure 4 – D- Solution Space



Combined Solution

It is now possible to determine a calculation strategy permitting climb and descent with positive & negative pitch angles.

The following procedure is proposed:

The C+ & D- solutions form the basis. (*The C- & D+ solutions are not appropriate.*) These are shown together in Figure 5:

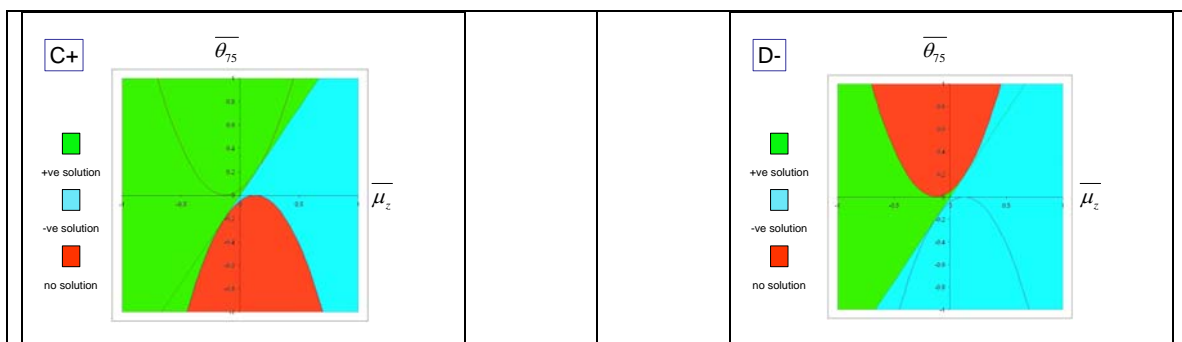


Figure 5 – Comparison of C+ & D- Solution Spaces

for any point in the domain of $\bar{\mu}_z$ & $\bar{\theta}_{75}$ a choice of C+ or D- solution is required.

The tables show the choice order.

Select 1 st Choice if it exists
Otherwise
Select 2 nd Choice



The choices are as follows:

θ_{75}	μ_z	1 st Choice	2 nd Choice
=>0	=>0	C+	C+
	<0	D-	C+
<0	=>0	C+	D-
	<0	D-	D-

This can be simplified to:

μ_z	1 st Choice	2 nd Choice
=>0	C+	D-
<0	D-	C+

Extension to Annulus Theory

The thrust over an elemental annulus is given by BET:

$$dT = \frac{1}{2} \rho (\Omega r)^2 \cdot N c dr \cdot a \left(\theta - \frac{\mu_z + \lambda_i}{x} \right) \quad (42.)$$



Momentum Theory - Climb

from momentum theory over the annulus:

$$dT = \rho(V_C + V_i) \cdot 2\pi r dr \cdot 2V_i \quad (43.)$$

Equating (42) & (43) gives:

$$(V_C + V_i)V_i = \frac{1}{8} \frac{Nc}{\pi} \cdot \left(\frac{V_T^2 x}{R} \right) \cdot a \left(\theta - \frac{\mu_z + \lambda_i}{x} \right) \quad (44.)$$

From which we have:

$$(\mu_z + \lambda_i)\lambda_i = \frac{sa}{8} (\theta x - \mu_z - \lambda_i) \quad (45.)$$

Hence:

$$\lambda_i^2 + \lambda_i \left(\mu_z + \frac{sa}{8} \right) - \frac{sa}{8} (\theta x - \mu_z) = 0 \quad (46.)$$

Normalizing on sa gives finally:



$$\bar{\lambda}_i^2 + \bar{\lambda}_i \left(\bar{\mu}_z + \frac{1}{8} \right) - \frac{1}{8} (\bar{\theta} x - \bar{\mu}_z) = 0 \quad (47.)$$

Momentum Theory - Descent

The thrust over an elemental annulus is given by BET:

$$dT = \rho(V_D - V_i) \cdot 2\pi r dr \cdot 2V_i \quad (48.)$$

Noting that:

$$V_D = -V_C \quad (49.)$$

from momentum theory over the annulus:

$$dT = -\rho(V_C + V_i) \cdot 2\pi r dr \cdot 2V_i \quad (50.)$$

Equating gives:

$$(V_C + V_i)V_i = -\frac{1}{8} \frac{Nc}{\pi} \cdot \left(\frac{V_T^2 x}{R} \right) \cdot a \left(\theta - \frac{\mu_z + \lambda_i}{x} \right) \quad (51.)$$

From which we have:



$$(\mu_z + \lambda_i)\lambda_i = -\frac{sa}{8}(\theta x - \mu_z - \lambda_i) \quad (52.)$$

Hence:

$$\lambda_i^2 + \lambda_i\left(\mu_z - \frac{sa}{8}\right) + \frac{sa}{8}(\theta x - \mu_z) = 0 \quad (53.)$$

Normalizing on sa gives finally:

$$\bar{\lambda}_i^2 + \bar{\lambda}_i\left(\bar{\mu}_z - \frac{1}{8}\right) + \frac{1}{8}(\bar{\theta}x - \bar{\mu}_z) = 0 \quad (54.)$$

Comparison of Actuator Disc Theory (ADT) & Annulus Theory (AT)

The quadratic equations for ADT & AT are presented below:

Climb

ADT

$$\bar{\lambda}_i^2 + \bar{\lambda}_i\left\{\bar{\mu}_z + \frac{1}{8}\right\} - \frac{1}{4}\left\{\frac{\bar{\theta}_{75}}{3} - \frac{\bar{\mu}_z}{2}\right\} = 0 \quad (55.)$$

AT



$$\bar{\lambda}_i^2 + \bar{\lambda}_i \left(\bar{\mu}_z + \frac{1}{8} \right) - \frac{1}{8} (\bar{\theta}_x - \bar{\mu}_z) = 0 \quad (56.)$$

Descent

ADT

$$\bar{\lambda}_i^2 + \bar{\lambda}_i \left\{ \bar{\mu}_z - \frac{1}{8} \right\} + \frac{1}{4} \left\{ \frac{\bar{\theta}_{75}}{3} - \frac{\bar{\mu}_z}{2} \right\} = 0 \quad (57.)$$

AT

$$\bar{\lambda}_i^2 + \bar{\lambda}_i \left(\bar{\mu}_z - \frac{1}{8} \right) + \frac{1}{8} (\bar{\theta}_x - \bar{\mu}_z) = 0 \quad (58.)$$

Inspection of these equation pairs shows the following equivalence between the local pitch angle (AT) and the pitch at 75% radius (ADT):

$$\bar{\theta}_x = \frac{2}{3} \cdot \bar{\theta}_{75} \quad (59.)$$

In other words, the AT solution can be achieved using the same choice of solution developed for the ADT.

The thrust variation can then be evaluated by:



$$T = \frac{1}{2} \rho V_T^2 \cdot NcR \cdot a \int_0^1 x^2 \left(\theta - \frac{\mu_z + \lambda_i}{x} \right) \cdot dx \quad (60.)$$

whence:

$$\frac{C_T}{sa} = \int_0^1 x^2 \left(\theta - \frac{\mu_z + \lambda_i}{x} \right) \cdot dx \quad (61.)$$

where finally:

$$\begin{aligned} \bar{C}_T &= \frac{C_T}{(sa)^2} \\ &= \int_0^1 x^2 \left(\bar{\theta} - \frac{\bar{\mu}_z + \bar{\lambda}_i}{x} \right) \cdot dx \end{aligned} \quad (62.)$$

