

Peak loading and surface pressure fluctuations of a tall model building

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Abstract

Using a recently developed inflow generation approach (Kim et al., 2013), Large-Eddy Simulation (LES) of the flow over the Commonwealth Advisory Aeronautical Council (CAARC) standard tall building has been performed. The surface pressure fluctuations were calculated and are shown to be in reasonable agreement with the measurements in the literature. Numerical experiments were conducted to investigate the effects of the inflow turbulence intensity and integral length scales. Previous research, e.g. Surry and Djakovich (1995) and Huang et al. (2007), suggests a presence of periodicity in the lateral forces on tall buildings, indicating a correlation of the surface pressures on the building's lateral faces. This paper also focuses on the correlations of surface pressure, local wall-normal forces and cross-wind velocity.

Keywords: inflow condition; large-eddy simulation; peak pressure; correlation; conditional sampling.

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1. Introduction

At present, a wind engineering toolbox consists of wind tunnel testing of scaled models, field measurements, and mechanical load testing. In recent years, the application of computational fluid dynamics (CFD) has made the numerical evaluation of wind loading an appealing approach. However, despite the popularity of and improvement in computational methods, the problem of providing a fully developed turbulent flow at the inlet, at a reasonable computational expense, is a continuing issue. This is particularly crucial for large-eddy simulation approaches. A common method is to place a sub-domain with axially periodic boundary conditions upstream of the model. The sub-domain provides an inflow for the model. This method can achieve a reasonable simulation of a fully developed turbulent flow and deduction of the ‘peak loading’ on the surface of a bluff body situated in the flow, but can be an expensive process. More recent research has focused on an inlet condition which leads to a more accurate assessment for peak loading and at a cheaper computational expense (e.g. Tamura et al., 2008). The present research applies a recently proposed inflow generator for wind engineering applications (Kim et al., 2013). The generated synthetic inflow turbulence is divergence-free. It is very efficient and has been validated for accurate calculations of pressure fluctuations. The efficacy of this new inflow generator for flows around a tall building was discussed previously in Daniels et al. (2012), in which the effects of inflow turbulence parameters, domain size and mesh resolution were studied.

Flows over isolated square or rectangular buildings have been extensively studied. Analysis has mainly focused on the use of wind tunnel testing but

26 has more recently tended towards a numerical approach. One of the more
27 popular high rise building models is the one described by the Common-
28 wealth Advisory Aeronautical Council (CAARC, Moss and Wardlaw, 1970).
29 The CAARC standard tall building model has been extensively used for the
30 analysis of wind loading on tall buildings in wind tunnel studies. Most of
31 these studies use the CAARC building as a means of calibrating experimental
32 techniques. Only more recently has CFD been applied to the CAARC build-
33 ing and most of these studies focus on calibrating various CFD numerical
34 methods. The main source of fluid-structure induced vibrations is thought
35 to be the pressure fluctuations on the building surfaces. Recent research
36 on the CAARC building have focused on the flow patterns and the root-
37 mean-squared (*r.m.s.*) pressure fluctuation coefficients around the building
38 perimeter (e.g., Dagneu and Bitsuamlak, 2010; Daniels et al., 2012; Huang
39 et al., 2007). Obtaining an accurate estimation for the pressure fluctuations
40 is critical when considering structural vibrations and resulting fatigue con-
41 cerns. However, this can be challenging. In particular, the approaching flow
42 conditions (e.g. turbulence intensity and integral length scales) and the un-
43 derlying surface conditions may significantly increase the uncertainties of the
44 prediction.

45 Our recent work (Daniels et al., 2012) on the numerical analysis of the
46 pressure fluctuations over the CAARC building confirms observations in the
47 literature, e.g. Huang et al. (2007), that the lateral force on the CAARC
48 building exhibits a clear periodicity. This suggests that the surface pressure
49 on one lateral face of the CAARC building must be in some (anti-phase)
50 correlation with that on the opposing side. Using conditional sampling in a

wind tunnel, Surry and Djakovich (1995) attempted to identify a correlation of pressure peaks on the two lateral faces of a tall building. They found that the pressure peaks at corresponding taps on the opposite lateral faces were in anti-phase (i.e. shifted by 180 degree). This is anticipated because of the antisymmetric nature of vortex shedding. Bartoli and Ricciardelli (2010) investigate the correlations of surface pressure and forces on windward, leeward and two lateral faces of a medium-rise, rectangular plan building in wind tunnel scale, and thus to quantify the errors associated with the quasi-steady assumption for wind load evaluation.

To the best of our knowledge, quantitative analysis of the correlations of the pressure fluctuations on the two lateral faces of the CAARC building in atmospheric boundary layer flows have not been reported in the literature. The present study takes account of the effect of various inflow parameters and focuses on a systematic analysis of these correlations, as well as those between local wall normal forces on the lateral faces, and of the cross-wind velocity fluctuations in the wake.

2. Outline of CFD modelling

The Large-Eddy Simulation (LES) approaches embedded in the open source CFD code OpenFOAM 1.7.1 (OpenFOAM, 2010) was used throughout this work. The Smagorinsky sub-grid scale (SGS) model with the constant $C_s = 0.1$ was used. The governing equations and the standard parameters can be found in the literature (e.g. Xie and Castro, 2008). For all simulations the initial durations were sufficiently long (e.g. greater than $500t^*$, where $t^* = U_H t / D$) to allow the flow to reach full development and the sub-

75 sequent averaging durations were about $500t^*$ to obtain converged statistics
76 and were used for further analysis, e.g. conditional sampling.

77 2.1. Computational domain and mesh resolution

78 The chosen CAARC building for this study is the classical model having
79 a flat-topped prismatic shape, a rectangular cross-section and lateral flat
80 walls with no parapets or other geometric details; the dimensions are: length
81 (D , streamwise) = 30m, width = 45m, and height (H , vertical) = 180m.
82 The Reynolds number based on the free stream velocity at $y = 4H$ and the
83 height of the building was $Re = 3 \times 10^5$, which is within the range used in
84 several papers for the validation of wind-tunnel testing. Initially we used a
85 large domain (i.e. $20H \times 4H \times 10H$) which is similar as that in Dagneu and
86 Bitsuamlak (2010). A smaller domain (i.e. $10H \times 4H \times 8H$, Fig.1a) was also
87 designed. We noticed that the results obtained using the two domain sizes
88 were consistent, and hence most of the results reported in this paper were
89 obtained from the smaller domain.

90 Although the CAARC building is a simple rectangular cylinder, the com-
91 putational mesh generation is not straightforward because of the requirement
92 for a good quality mesh in the near wall regions. Also the total number of
93 cells must be kept relatively low in order to maintain an efficient compu-
94 tation. A structured mesh (Fig.1a) was chosen. The total number of cells
95 used was 7.2 million and 9.4 million for the smaller and larger domains, re-
96 spectively. The grid stretch ratios of the mesh outward from the CAARC
97 building are as follows: front, 1.08; back, 1.09; sides, 1.06; and top, 1.06.
98 The y_1^+ values are in the range used in the comparative papers of Huang
99 et al. (2007), and Dagneu and Bitsuamlak (2010) (case 1A), with $y_1^+ < 10$.

100 A wall resolution sensitivity test with $10 < y_1^+ < 20$ was performed, and
 101 is further discussed in Section 3.2. The Smargorinsky SGS model suffers
 102 from its excessive dissipation in high shear regions. So the C_s constant must
 103 be reduced in these regions by using a damping factor, e.g. the Van-Driest
 104 damping function (van Driest, 1956),

$$D = 1 - \exp\{-y^+/A^+\}, \quad (1)$$

105 where y^+ is the wall normal distance in wall units to the cell centre, and A^+
 106 is set 26 in OpenFOAM1.7.1. This damping function was applied to the floor
 107 of the domain and the surfaces of the CAARC building.

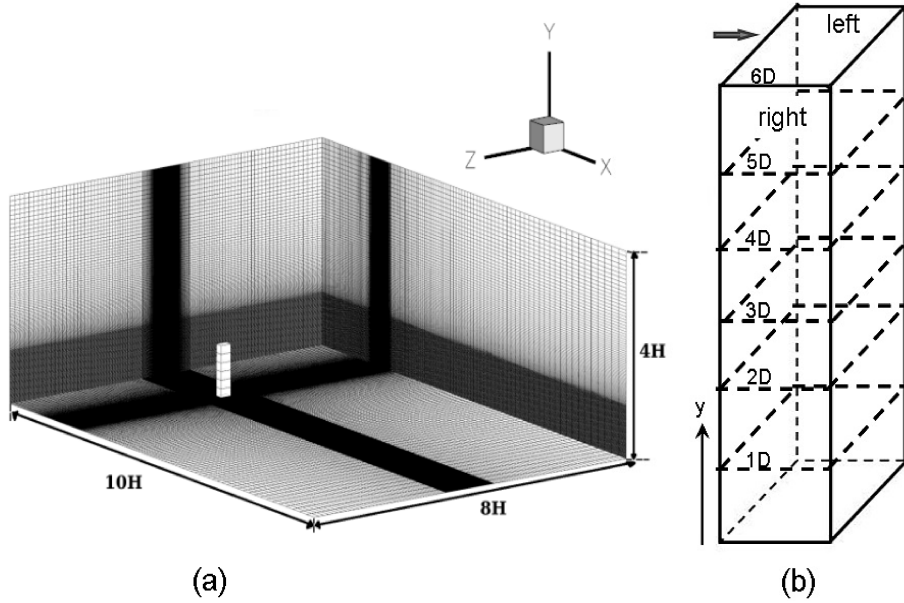


Figure 1: (a) Overall grid distribution, with coordinates x, y, z corresponding to stream-wise, vertical and lateral respectively. (b) A sketch of section partitions on the lateral faces.

108 2.2. Boundary conditions

109 The boundary conditions represent the settings used for the wind tunnel
110 test of Dagneu and Bitsuamlak (2010). The lateral sides and top of the
111 domain were set as symmetry-planes. No-slip conditions were used on the
112 walls of the CAARC building and floor of the domain.

113 A recently proposed inflow generator (Kim et al., 2013) (hereafter
114 XC_DF_07) was used for the inlet condition and is now briefly described.
115 Synthetic turbulence for inflow conditions formulated on a 2-D plane gen-
116 erally produces unphysically large pressure fluctuations in direct numerical
117 and large-eddy simulations (e.g. Xie and Castro, 2008). To reduce such ar-
118 tificial fluctuations a divergence-free inflow generator XC_DF_07 has been
119 developed recently with incompressible flow solvers. Firstly, the generated
120 synthetic turbulence using (Xie and Castro, 2008) is inserted on a transverse
121 plane ($x = x_0$) in the computational domain near, but crucially not *at*, the
122 inlet boundary after having solved the momentum equations using the PISO
123 procedure (Issa, 1985). Kim et al. (2013) suggests that XC_DF_07 works well
124 for x_0 greater than a quarter of the channel depth. In the current simula-
125 tions, we chose $x_0 = 3D$. Secondly, these velocities are then adjusted by
126 the velocity-pressure coupling procedure. This means that, on application
127 of the pressure-correction step, the imposed velocities on the plane where
128 the synthetic turbulence is introduced only act as intermediate velocities.
129 Applying synthetic turbulence on the inlet boundary itself, in contrast, fixes
130 those velocities as final velocities throughout one time step. Thirdly, once the
131 synthetic turbulence goes through the velocity-pressure coupling procedure,
132 the velocities are adjusted and satisfy the divergence-free condition. As ex-

133 pected, the final velocities are not generally exactly the same as the original.
134 Nevertheless the changes are found to be small. An important feature of this
135 method is that it does not require any additional computational cost.

136 The inlet condition requires the user to define profiles of the integral
137 length scales (ILS), turbulence intensity (TI), Reynolds stresses and mean
138 velocity. The velocity profile and streamwise velocity fluctuations ($\overline{u'u'}$) were
139 calculated empirically from the mean velocity and turbulence intensity de-
140 rived from the Dagneu and Bitsuamlak (2010) wind tunnel results. The
141 mean velocity was modelled using a power law exponent of 0.16 with ref-
142 erence velocity $U_H = 11.7m/s$ at the building height. The other Reynolds
143 stress components ($\overline{v'v'}$, $\overline{w'w'}$, and $\overline{u'v'}$) were calculated using appropriate re-
144 lations suggested by Xie and Castro (2009). The streamwise integral length
145 scale of turbulence was deduced from the collected wind tunnel data from
146 Obasaju (1992). And following Xie and Castro (2009), the integral length
147 scales in the streamwise, vertical and lateral directions were set $4D$, D and
148 D respectively.

149 *2.3. Performance of the inflow methods*

150 Both Huang et al. (2007) and Dagneu and Bitsuamlak (2010) studies used
151 a spectral synthesizer available in ANSYS Fluent to generate the velocity
152 fluctuations (Smirnov et al., 2001). In this method, the fluctuating velocity
153 components are computed by synthesizing a divergence-free velocity vector
154 field from the summation of the Fourier harmonics on the basis of the input
155 turbulence boundary conditions. However, this method leads to a rapid
156 energy decay in the inertial-subrange of turbulence, creating an early cut-off
157 frequency as detailed in Huang et al. (2010).

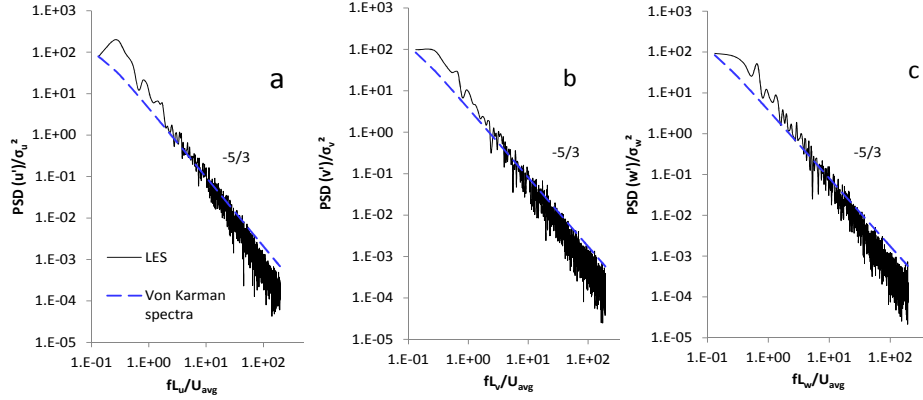


Figure 2: Power spectral density of velocity fluctuations at $x = 3.17D$, $y = D$ on the central plane. (a) u' , (b) v' , (c) w' .

158 In order to improve this, Huang et al. (2010) apply a Dirac delta func-
 159 tion to the Smirnov et al. (2001) method to construct velocity spectra with
 160 appropriate shapes. The Huang et al. (2010) method pays a high price on
 161 CPU time for the turbulence generation. With the default settings in Huang
 162 et al. (2010), it is 250 times more expensive than the Smirnov et al. (2001)
 163 method. Huang et al. (2010) comment that this method is, however, more
 164 suitable than the earlier ones for parallel computations, which are naturally
 165 much more efficient.

166 It is to be noted that the major feature of the Xie and Castro (2008)
 167 method is its very high efficiency and its flexibility. Its improved version –
 168 the new divergence-free method (Kim et al., 2013) – has a similar efficiency.
 169 The velocity spectra presented in Huang et al. (2010) show a substantial im-
 170 provement compared with those using the Smirnov et al. (2001) method. Our

171 methods (Xie and Castro, 2008; Kim et al., 2013) show a similar performance
 172 as the Huang et al. (2010) method in this respect. Fig.2 shows a power spec-
 173 tral density of velocity fluctuations at a station (i.e. at $x = 3.17D, y = D$
 174 on the central plane) immediately downstream of the x_0 plane where the
 175 synthetic turbulence is imposed. These are consistent with those in Xie and
 176 Castro (2008) and Kim et al. (2013). The von Karman wind spectra (ESDU-
 177 85020, 2001) are also included in Fig.2 for comparison. In this model, the
 178 spectra for the three velocity components are described,

$$\frac{PSD(u')}{\sigma_u^2} = \frac{4n_u}{f(1 + 70.8n_u^2)^{5/6}}; \quad n_u = fL_u/U_{avg} \quad (2)$$

$$\frac{PSD(\xi')}{\sigma_\xi^2} = \frac{4n_\xi(1 + 755.2n_\xi^2)}{f(1 + 283.2n_\xi^2)^{11/6}}; \quad n_\xi = fL_\xi/U_{avg}; \quad \xi = v \text{ or } w \quad (3)$$

179 where L is the integral length scale, f is the frequency, σ^2 is the variance, and
 180 U_{avg} is the local average velocity. The LES spectra show an evident slope
 181 $-5/3$. However, at very high frequencies the LES spectra show a slightly
 182 steeper slope, which might be due to the limited resolution.

183 It is crucial to check the flow development from the inlet to immediately
 184 upstream of the CAARC building, which might be affected by the coarse
 185 resolution in the near inlet region. We extensively checked the mean and
 186 turbulence intensity profiles. We found that the turbulence flow converged
 187 very quickly, e.g. from $3D$ downstream of x_0 - where the inflow synthetic
 188 turbulence was inserted.

189 3. Data analysis and discussion

190 3.1. Mean and *r.m.s.* fluctuating pressure

191 While there is a wide range of wind tunnel tests which might be used to
192 validate our results, such tests have to be chosen carefully. Early experimen-
193 tal testing, such as the data collected from five institutions by Melbourne
194 (1980), generally shows some inconsistencies between the pressure measure-
195 ments. Huang et al. (2007) made a detailed comparison of the Melbourne
196 (1980) data and deduced that the effects of the inflow conditions on the pres-
197 sure results is significant. Therefore for the purpose of validation for LES,
198 we choose wind tunnel data (Huang and Gu, 2005; Dagneu and Bitsuamlak,
199 2010) having similar inlet characteristics to those employed herein.

200 The mean pressure and *r.m.s.* of pressure fluctuation coefficients are re-
201 spectively defined as $C_p = (p_m - p_{ref}) / (0.5\rho U_H^2)$ and $c'_{rms} = (p_{rms}) / 0.5\rho U_H^2$,
202 where U_H is the undisturbed velocity at the CAARC building height at the
203 inlet, p_{ref} is the mean pressure taken at 3H above and H upstream of the
204 CAARC building, p_{rms} is the *r.m.s.* of the pressure fluctuations and ρ is the
205 air density. A Large-Eddy Simulation (LES) in Huang et al. (2007) with
206 the inflow method from Smirnov et al. (2001), a Reynolds Averaging Navier-
207 Stokes (RANS) RNG k- ϵ calculation (Dagneu and Bitsuamlak, 2010), and
208 wind tunnel tests (Dagneu and Bitsuamlak, 2010; Huang and Gu, 2005)
209 were all used to evaluate the mean pressure and root-mean squared (*r.m.s.*)
210 surface pressure fluctuations; results are compared with the present compu-
211 tations (using XC_DF_07). The mean and *r.m.s.* fluctuating pressure on the
212 CAARC building are plotted in Figs. 3 and 4 respectively. These include the
213 pressure results for cases in which the inlet turbulence intensity was altered

214 by either doubling or reducing to one hundredth the base value, in these cases
 215 the integral length scales (L_u , L_v , and L_w) remained the same (base). As well
 216 as this, the effect of the integral length scales were studied by halving the
 217 base values while maintaining the same (base) turbulence intensity. In the
 218 case of the RANS calculation, the averaging technique failed to capture the
 219 separation bubble on the corners of the CAARC building; hence the large
 220 underestimation of mean pressure at $x/D = 1.5$ in Fig.3. The present LES-
 221 predicted mean pressure coefficient over the building perimeter at $y = 2/3H$
 222 is in good agreement with all of the results, apart from the RANS data.

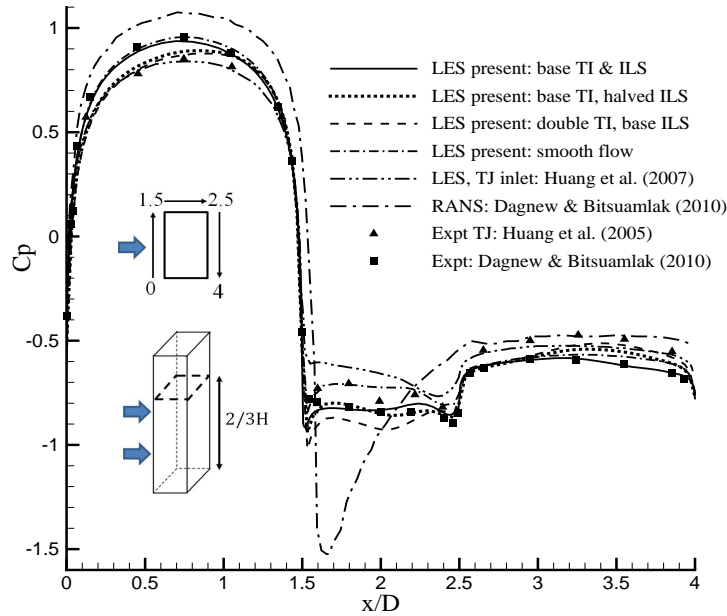


Figure 3: Mean pressure coefficients over the CAARC building perimeter ($y=2/3 H$).

223 For the *r.m.s.* pressure fluctuations (Fig.4), the XC_DF_07 inlet condition
 224 captures the high fluctuations on the rear corners of the CAARC building
 225 but shows some discrepancies in the magnitude of the fluctuations on the

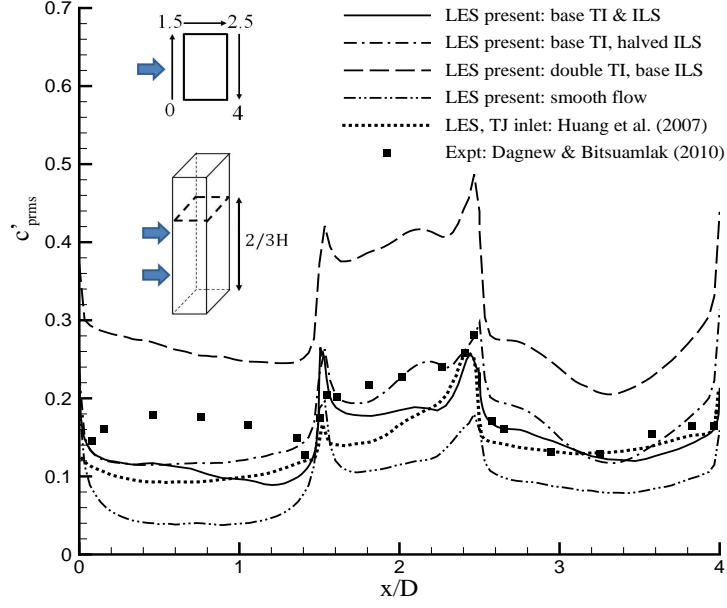


Figure 4: Root-mean-squared pressure fluctuations around the perimeter of the CAARC building at $y = 2/3H$.

lateral sides. Such discrepancies could be due to differences in the specified mean velocity power-law exponent, and turbulence intensity at the inlet. Huang et al. (2007) suggested that the pressure fluctuations on the CAARC building are less sensitive to the velocity profile than the turbulence intensity, but the mean pressure is more sensitive to the mean velocity profile than the turbulence intensity. Such observations agree with this present study. Although the mean velocity and turbulence intensity for the inlet condition were taken from Dagnew and Bitsuamlak (2010), the approximations for the Reynolds stress components are slightly different, which could explain the differences in the *r.m.s.* surface pressure fluctuations. The mean pressure around the CAARC building is in agreement with Dagnew and Bitsuamlak (2010)'s wind tunnel test, suggesting that the mean velocity profile has the

238 dominant effect on the mean pressure.

239 The LES results in Fig. 4 generally agree with the wind tunnel data.
240 However there is a slight contradiction concerning where the pressure fluc-
241 tuations on the front face are at a maximum, and there is debate about this
242 in previous studies of the CAARC building. Goliger and Milford (1988) sug-
243 gested that the wide range of turbulence intensities and experimental errors
244 presented by many wind tunnel tests may be the reason for the variations in
245 the pressure fluctuations on the front face of the CAARC building. Never-
246 theless, the distribution of fluctuating pressure on a square-section cylinder
247 (Bearman and Obasaju, 1982), which has the lowest pressure fluctuation
248 at the centre of the windward face, provides some evidence to support the
249 present results.

250 Research on the effects of integral scale on the pressure fluctuations over
251 bluff bodies have dealt almost exclusively with low-rise buildings (e.g. Holdø
252 et al. (1982)). Xie and Castro (2008) investigated numerically the effects
253 of the integral length scales of the flow over an array of staggered cubes
254 of uniform height. Their results show that the integral length scale has a
255 small effect on the flow statistics (i.e. the Reynolds stresses), as long as the
256 estimated integral length scales are within an appropriate range. No surface
257 pressure fluctuations were reported in Xie and Castro (2008). Figs. 3 and 4
258 show the changes of the pressure results by halving the base integral length
259 scales. The effect of the integral length scales are small for both mean and
260 *r.m.s.* fluctuating pressures, especially when compared with the effect due
261 to the variation of the turbulence intensity in Fig.4. The small variation of
262 C_p and c'_{prms} due to the changes of the integral length scales in Figs.3 and 4

seems within the numerical uncertainties, e.g. averaging error. This confirms the point in Xie and Castro (2008) and also suggests that the integral length scale has a small effect on the surface pressure fluctuations as long as they are within an appropriate range.

3.2. *Correlations between aerodynamic forces and surface pressure fluctuations*

Each lateral face of the CAARC building was partitioned into 6 sections in the vertical direction (Fig.1b). Pressure fluctuation coefficients at two stations of each section were sampled. The total wall-normal force on each section was also sampled.

Surface pressure fluctuations on the two sections at $y = 3D$ of the two lateral faces are clearly in anti-phase (i.e. shifted by 180 degree), as shown in Figs.5a and b. The total wall-normal force fluctuations on the left lateral section at $y = 2.5D$ (Fig.5c) are also in anti-phase with those on the right lateral section shown in Fig.5d. The lateral velocity fluctuations in the wake at various distances from the leeward face of the building (Figs.5e and f) are almost in phase. Figs.5a-f collectively show the same dominant frequency, with the Strouhal number $fD/U_H \sim 0.076$, where f is the frequency, D is the building length in streamwise direction and U_H is the reference velocity which is the inlet velocity at building height. This is consistent with the dominant frequency of the total lateral force on the CAARC building (Daniels et al., 2012), and is also in agreement with the literature, e.g. Huang et al. (2007). These again confirm that the vortex shedding from the lateral faces is the genuine mechanism (Surry and Djakovich, 1995) driving the oscillation of the aerodynamic forces.

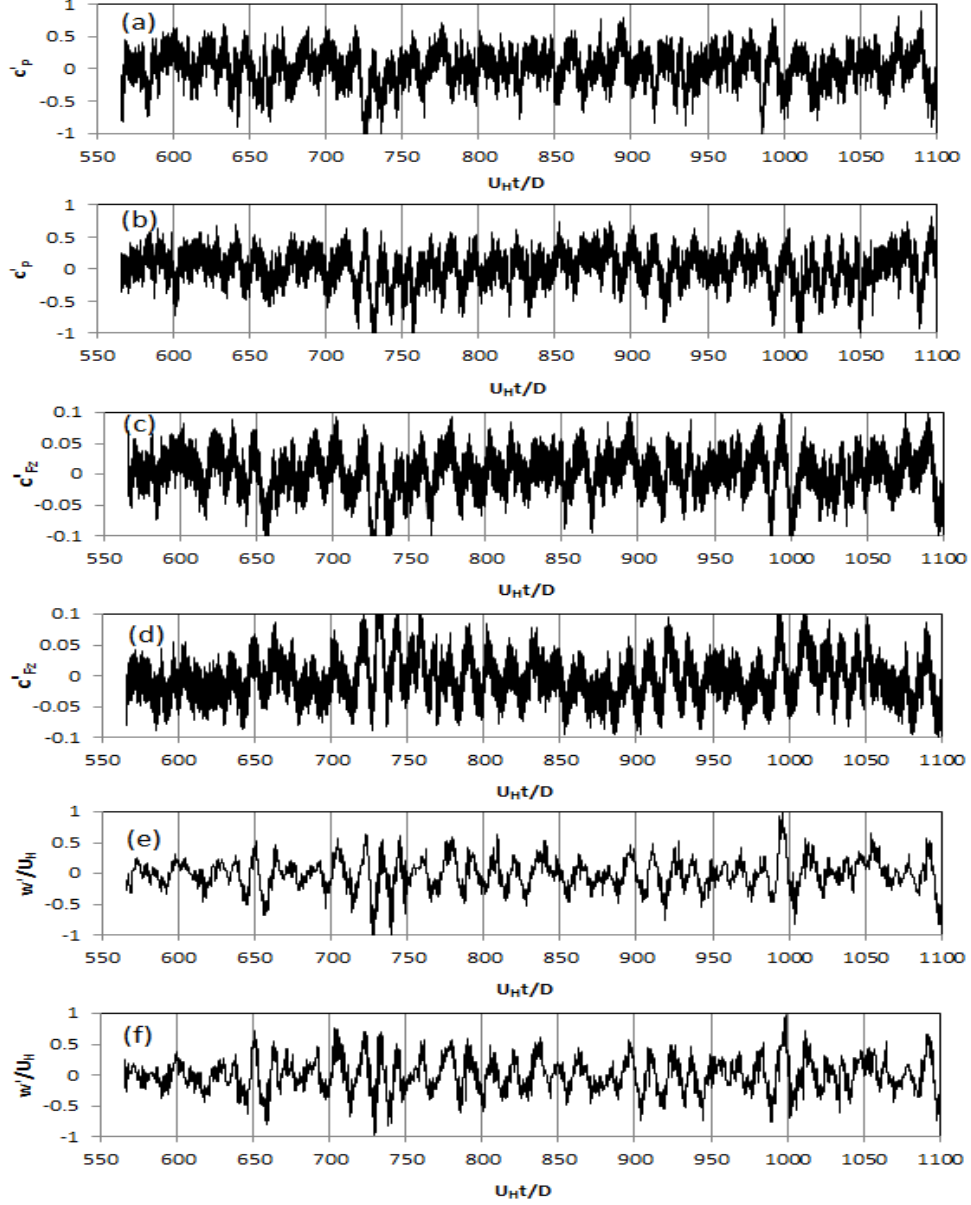


Figure 5: (a) A time series of the pressure coefficients on left lateral face at $y = 3D$ at the centre of the section, (b) as for (a) but on the right lateral face; (c) a time series of the coefficient of the wall-normal force fluctuation on the left lateral face section at height $y = 2.5D$, (d) as for (c) but on the right lateral side; (e) a time series of the dimensionless spanwise velocity fluctuation w' at $0.17D$ from the leeward side and at $y = 3D$ on the central plane, (f) as for (e) but at $0.5D$ from the leeward side.

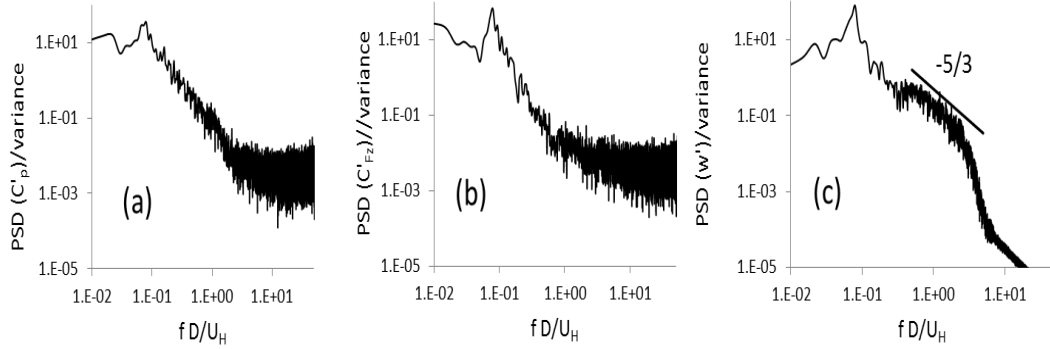


Figure 6: Power spectral density of (a) pressure fluctuation coefficient on the left lateral face at $y = 3D$, and at a distance $0.5D$ from the leading corner (Fig.5a), (b) wall-normal force fluctuation coefficient on the left lateral face section at height $y = 2.5D$ (Fig.5c), and (c) spanwise velocity fluctuations at $y = 3D$ and at a distance $0.5D$ from the leeward side on the central plane (Fig.5f).

288 The total wall-normal forces on sections at $y = 2.5D$ were highly corre-
 289 lated (i.e. correlation coefficients exceeding 0.9) with the surface pressures on
 290 the same section. This is clearly shown in Figs.5a and c. This might not be
 291 surprising because the total wall normal force is the integration of the surface
 292 pressure. The wake velocities in Figs.5e and f also show a high correlation
 293 with the surface pressures (compare Figs.5a and b with Figs.5e and f). Also
 294 a phase lag between the wake velocities and the surface pressures is evident.
 295 This will be further discussed in §3.3.

296 Fig.6a shows the power spectral density of the pressure fluctuation coef-
 297 ficient on the left lateral face at $y = 3D$, and at a distance $0.5D$ from the
 298 leading corner (the corresponding time series is Fig.5a). The spectra shows
 299 an evident peak at $fD/U_H \sim 0.076$, confirming the dominant vortex shedding
 300 frequency stated earlier. The spectra of the time series in Fig.5b sampled

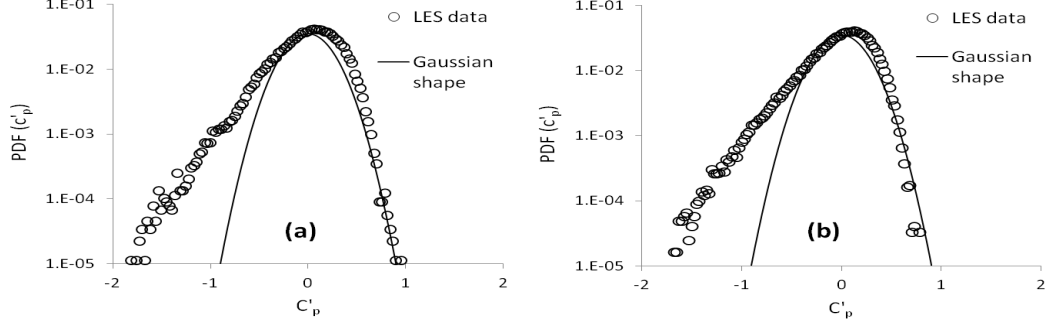
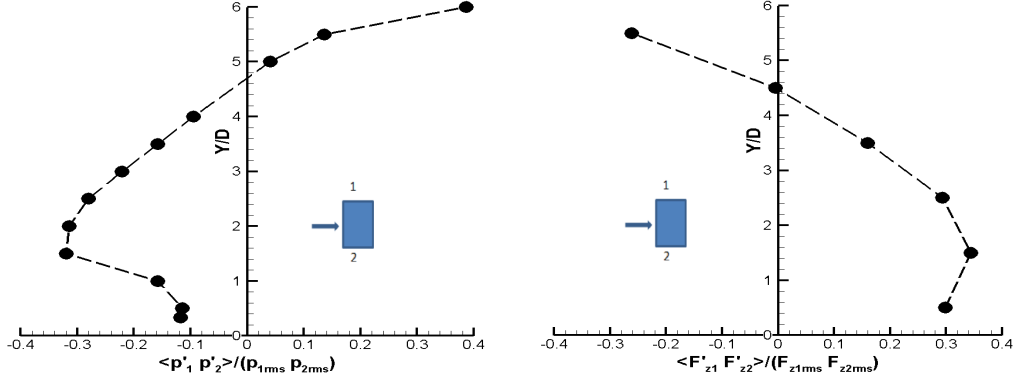


Figure 7: Probability Density Functions (PDFs) of surface pressure fluctuation coefficient on the left lateral face at a distance $0.5D$ from the leading corner. (a) at $y = 3D$ (Fig.5a), (b) at $y = 4D$.

301 at the right lateral face is almost identical as that at the left lateral face
 302 in Fig.6a. Fig.6b shows the power spectral density of the wall-normal force
 303 fluctuation coefficient (the time series is Fig.5c) on the left lateral face sec-
 304 tion at height $y = 2.5D$, which also shows a dominant frequency ($fD/U_H \sim$
 305 0.076). This is expected as the section wall-normal force is highly correlated
 306 with the local surface pressure. Fig.6c shows the power spectral density of
 307 the spanwise velocity fluctuations (Fig.5f) at $y = 3D$ and at a distance $0.5D$
 308 from the leeward side on the central plane. It again confirms the dominant
 309 frequency ($fD/U_H \sim 0.076$) in the wake. An inertial sub-region is evident
 310 in Fig.6c in the wake region.

311 It has been reported in the literature (eg. Gioffre et al., 2001) that the
 312 Probability Density Function (PDF) of surface pressure fluctuations is non-
 313 Gaussian. The present numerically predicted surface pressure fluctuations on
 314 the lateral and the leeward faces are likewise non-Gaussian; Fig.7 shows two
 315 examples. Fig.7a shows a PDF of surface pressure fluctuation coefficient (see



(a) Correlations between the corresponding surface pressures at $0.5D$ from the leading corner on the two lateral faces.

(b) Correlations between the corresponding total wall-normal forces of the two lateral face sections (see Fig.1b).

Figure 8: Correlations of the pressures and wall-normal forces on the lateral faces.

the time series in Fig.5a) on the left lateral face at a distance $0.5D$ from the leading corner and at $y = 3D$. The LES data show a much longer negative tail in the PDF than that of a Gaussian distribution. Fig.7b shows a PDF of the pressure fluctuation at $y = 4D$ on the same lateral face; this is quite similar to that in Fig.7a.

Fig.8a shows the correlations between the corresponding surface pressure fluctuations at $x = 0.5D$ from the leading corner on the two lateral faces, and Fig.8b shows the correlations between the corresponding total wall-normal force fluctuations of the two lateral face sections (see Fig.1 b), where subscripts 1 and 2 refer to left and right lateral faces respectively. Figs.8a and 8b show similar but oppositely signed correlation. The maximum of the absolute values of the correlations at the medium height is only around 0.3. This is certainly because the high inflow turbulence intensity reduces such correlations. Note that it is not possible to distinguish between a change in

330 coherence of the fluctuations and a change in phase using simple correlations
 331 only at zero lag. Instead, we address the issue of phase using conditional
 332 sampling in §3.3.

333 The correlation in Fig.8a increases to zero at $y \sim 4.5D$ and then to
 334 nearly +0.4 at the top of building. This is because large structures shedding
 335 from the top face generate in-phase pressure fluctuations on the top face and
 336 the two lateral faces near the top. We also checked time series of pressure
 337 fluctuations on the windward and leeward faces near the top. They all had a
 338 clear in-phase tendency with the pressure fluctuations on the two lateral faces
 339 near the top. (There might be other mechanisms to be investigated, but we
 340 have not pursued this further.) Again it is not possible to distinguish between
 341 a change in coherence of the fluctuations and a change in phase using such
 342 simple correlations. Near the ground, the correlations both seem to approach
 343 zero. This might be due to the very small scale eddies in the near surface
 344 region. The effect of the first wall grid resolution was also investigated. We
 345 changed the average first grid resolution from 20 wall units to 10 wall units,
 346 but found no significant effects.

347 *3.3. Deductions from conditional sampling: peak surface pressure, aerody-* 348 *namie forces and wake velocities*

349 Conditional sampling and averaging has been widely used for quanti-
 350 tative analysis in various problems, e.g. identifying coherent structures in
 351 quasi-periodic and periodic flows (Antonia, 1981) and determining peak sur-
 352 face pressure fluctuations (Surry and Djakovich, 1995; Lam and Zhao, 2002).
 353 Usually a number of simultaneous time histories of various variables at var-
 354 ious stations are considered. Only one time history of one variable at one

station is chosen as the triggering signal. If the value of this variable at a certain time (i.e. triggering time) exceeds the threshold, then all of the simultaneous time histories before and after the triggering time within a small constant duration are sampled. This corresponds to one event. The triggering time for every event is translated to zero. Finally all of the time histories of the events are averaged. This technique significantly reduces the size of the massive dataset and identifies the extreme events more clearly.

In the context of wind loading problems the technique is particularly useful for investigating the correlation of the peak pressures at different locations on the building surfaces, as well as the correlation of the peak pressure and the velocity components. It is also very useful for studying the correlation of the peak surface pressure and the peak loading on the buildings. We present here our specific technique and some of the results.

As in §3.2, each lateral face of the building was evenly partitioned into 6 sections in the vertical direction (Fig.1b). Surface pressure fluctuation coefficients at stations on the lateral faces, with heights $y = 5D, 4D, 3D, 2D$ and $1D$ respectively, and a distance $0.5D$ from the windward face of the building, were recorded for every time step for a duration $426t^*$ (i.e. 1162s), where $t^* = U_H t / D$. The wall normal force coefficients on all the sections were recorded simultaneously for the same duration. Lateral velocity components at stations on the central plane in the wake region were also recorded simultaneously.

The pressure fluctuation coefficient $c'_p = (p - p_m) / (0.5\rho U_H^2)$ at $y = 3D$ and a distance $0.5D$ from the leading corner on the left lateral face (Fig.1b) was chosen as the triggering signal. Alternatively, it would be straightforward

380 to use the local force as the trigger signal (Lam and Zhao, 2002). In order
381 to get sufficient number of events for averaging, the triggering threshold was
382 set to be -0.53 . We focused more on the negative peak c'_p because it was of
383 greatest interest (Surry and Djakovich, 1995). Using the conditional sampling
384 technique, the total number of sampled events n was 21. It is to be noted
385 that the period of the vortex shedding was about $13t^*$ (i.e. 36s).

386 The conditionally averaged time histories of the pressure fluctuation co-
387 efficients at heights $y = 5D, 4D, 3D, 2D$ and $1D$ on the both lateral faces
388 are shown in Fig. 9. The peak c'_p shows an evident anti-phase correlation
389 with that at the same height on the opposing lateral face. This was also
390 observed by Surry and Djakovich (1995), whereas on the same lateral face
391 all of the peaks at various height were in an excellent correlation. Surry and
392 Djakovich (1995) also observed that the peaks first occurred at upper levels
393 then sequentially towards the bottom of the building. Fig.9 shows that the
394 high peaks at $y = D$ occur approximately $2t^*$ (i.e. 6s) later than those at
395 $y = 5D$.

396 Fig.10 (a) shows the sum of the conditionally averaged time histories of
397 the wall-normal force coefficients on the two opposite lateral face sections.
398 These are the lift coefficients $C_l = L/(0.5\rho U_H^2 A)$, where L is the total lateral
399 forces on the two lateral sections at the same height, ρ is the air density,
400 U_H is the reference velocity and A is the surface area. Fig.10(a) shows the
401 same pattern as that in Fig.9. The lift coefficients at all heights are highly
402 correlated. The high peak at $y = 0.5D$ occurs approximately $2t^*$ (i.e. 6s)
403 later than those at $y = 5.5D$. This is consistent with the same observation
404 shown in Fig.9. Fig.10(a) also shows that the peaks of lift coefficients are

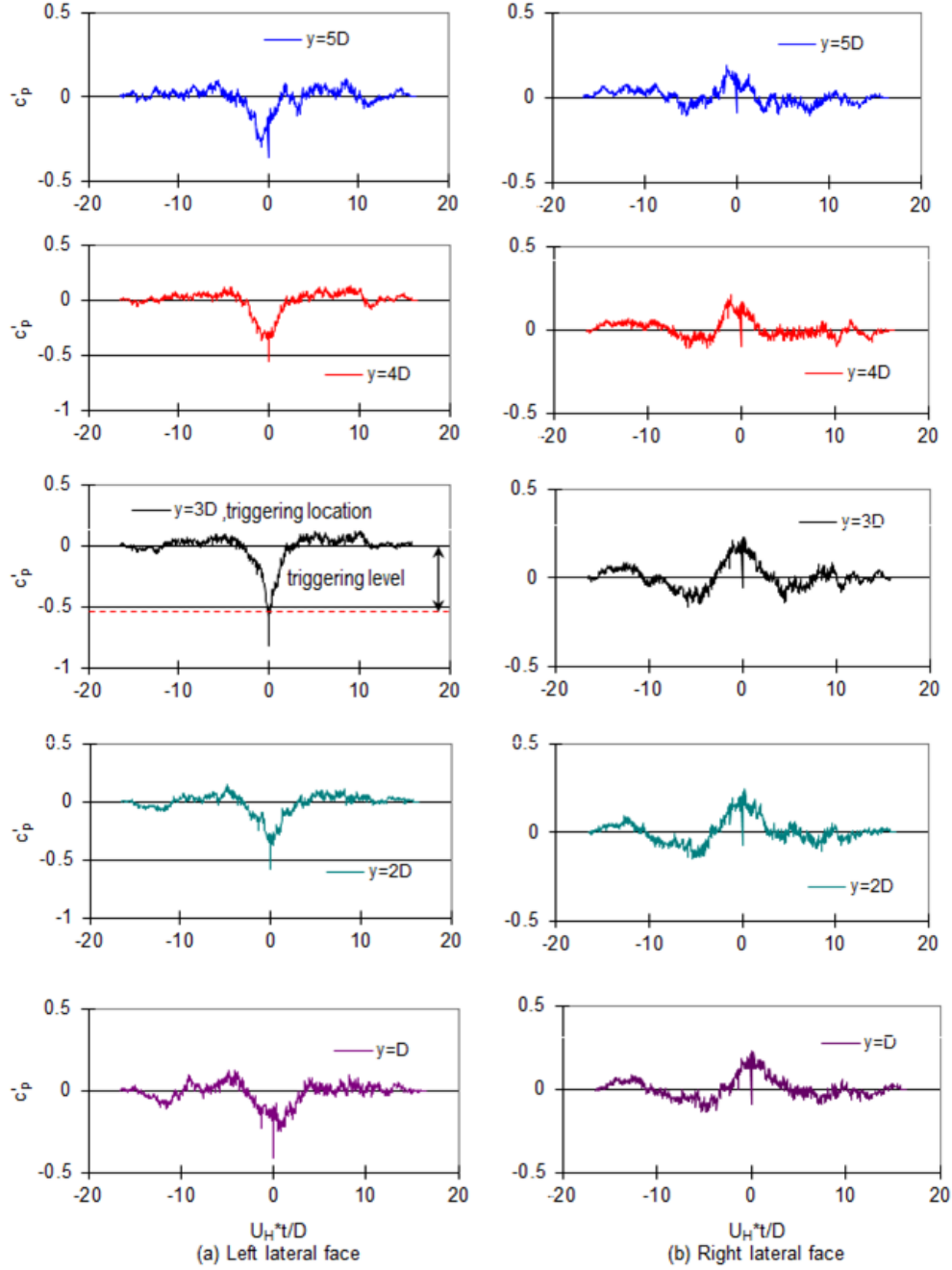


Figure 9: Conditionally averaged ($n = 21$) time series of pressure fluctuation coefficients at heights $y = 5D$, $4D$, $3D$, $2D$ and $1D$. (a) Left lateral face. (b) Right lateral face.

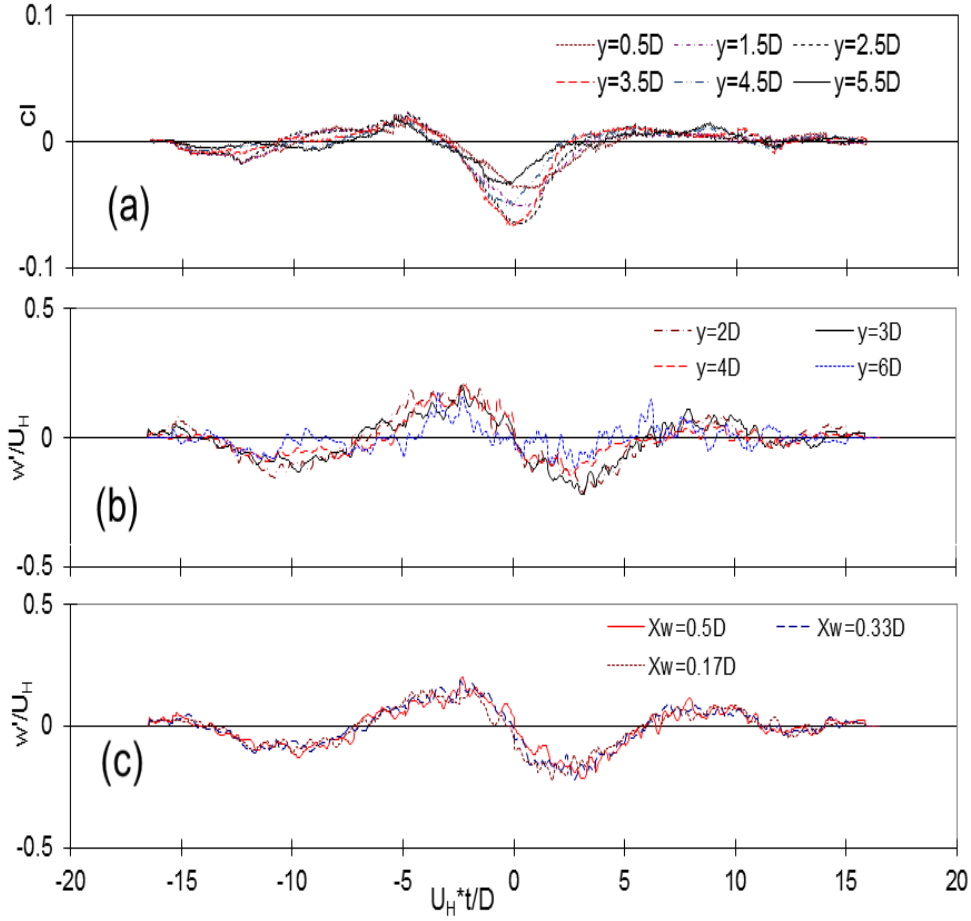


Figure 10: Conditionally averaged ($n = 21$) time series. (a) Lift coefficients at heights $y = 5.5D, 4.5D, 3.5D, 2.5D, 1.5D$ and $0.5D$. (b) Lateral velocity in the wake; distance to the leeward face $x_w = 0.5D$; $y = 6D, 4D, 3D$ and $2D$. (c) Lateral velocity in the wake; distance to the leeward face $x_w = 0.5D, 0.33D$ and $0.17D$; $y = 3D$.

405 in excellent correlation with the peaks of pressure fluctuations at the same
 406 heights on the left lateral face shown in Fig.9.

407 Fig.10(b) plots conditionally averaged time histories of lateral velocity
 408 fluctuations in the wake region at a distance from the leeward face $x_w = 0.5D$
 409 and at four heights $y = 6D, 4D, 3D$ and $2D$. Fig.10(c) plots averaged time
 410 histories at $x_w = 0.5D, 0.33D, 0.17D$ and $y = 3D$. All of the time histories

411 at various stations are closely in-phase. Nevertheless, the pattern of the high
 412 peaks in Fig.10 (b,c) is different from that in Figs.9 and 10(a). Fig.10 (b,c)
 413 shows that w' reduces to zero at $t^*=0$. Note that the extreme events of the
 414 surface pressures and lift coefficients occur at around the triggering time $t^*=0$
 415 and that the mean lateral velocity $w_m = 0$. The negative peak of w' occurs
 416 at about $t^*=3$, which is approximately at a phase angle 90° , taking $t^*=0$ as
 417 the time for phase angle 0° .

418 Surry and Djakovich (1995) placed monitors at various distances from the
 419 windward side of the building to sample the lateral velocity. They observed
 420 a time delay between the peak velocity and the peak surface pressure on the
 421 building lateral faces. They commented that this delay was consistent with
 422 a convected disturbance traveling at a certain speed. They also speculated
 423 that the peak velocity was associated with the previous vortex shedding from
 424 the opposite leading corner. Similarly, the time delay between the negative
 425 peak surface pressure and the negative peak lateral velocity in the wake is
 426 probably associated with the current vortex shedding from the leading edge
 427 of the left lateral face (see Fig.11a-c) and travelling to the near-leeward face
 428 region.

429 Fig.11 shows typical instantaneous velocity vectors and contours of pres-
 430 sure fluctuation coefficients at phases -90° , 0° and 90° , again assuming that
 431 $t^*=0$ is corresponding to phase angle 0° . While attached to the left lateral
 432 faces (Fig.11b), the vortex enhances the velocity magnitude at the outside
 433 edge of the separation bubble, and subsequently produces a negative peak
 434 surface pressure fluctuation. When the surface pressure exceeds the pre-
 435 scribed threshold, the conditional sampling is triggered. The dimensionless

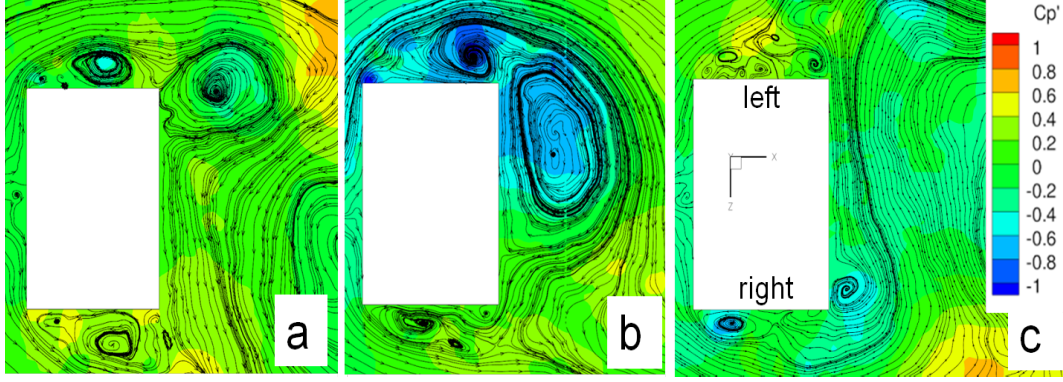


Figure 11: Typical instantaneous velocity vectors and contours of pressure fluctuation coefficients at $y = 3D$. (a) at phase -90° . (b) at phase 0° . (c) at phase 90° .

436 apparent travel time is consistent with the time delay $t^*=2.5$, given that
 437 the travelling velocity is less than U_H and the travelling distance is about
 438 $2D$, which is the distance from the centre of vortex at the left lateral face
 439 in Fig.11b to the centre (not shown due to space) of vortex in the wake in
 440 Fig.11c. Figs.11a, b and c respectively show a positive peak, a very small
 441 value and a negative peak of the w velocity in the near-leeward-face region
 442 on the central plane. The surface pressures on both lateral sides in Figs.11a
 443 and 11c are small and of the same order (also see Fig.9). These results also
 444 confirm that the surface pressures are correlated to the wake velocities, as
 445 we noticed in §3.2.

446 4. Conclusions

447 A recently proposed synthetic turbulent inlet condition for Large-Eddy
 448 Simulation (LES) has been used to predict the mean pressure and root-
 449 mean-squared (*r.m.s.*) of the pressure fluctuation over a CAARC building
 450 model surface. The new inlet condition has proven to be very successful in

451 modelling the mean pressure. As for the *r.m.s.* of the pressure fluctuations,
452 the present study has captured the detailed fluctuations on the corners of the
453 CAARC building. Numerical experiments show that the inflow turbulence
454 intensity has a significant effect on the surface pressure fluctuations, whereas
455 the inflow turbulence integral length scales have a negligible effect, as long
456 as they are within an appropriate range.

457 High correlations were observed between the surface pressures and the lo-
458 cal wall-normal forces on the two lateral faces of the building. The dominant
459 frequencies of the surface pressures, the local-wall normal forces and the wake
460 lateral velocity are essentially identical, and in good agreement with that of
461 the total lateral force. A conditional sampling and averaging technique was
462 used to investigate the correlations of peak surface pressure on the two lat-
463 eral faces and the lateral velocity component in the wake region. The peak
464 surface pressure fluctuations were found (as expected) to be in anti-phase
465 with those at the same height on the opposing lateral face, no doubt because
466 of the antisymmetric nature of vortex shedding from the lateral faces. It was
467 also found that there was a phase delay (of approximately 90°) between the
468 negative peak of the triggering surface pressure and the negative peak lateral
469 velocity (towards the triggering location) in the wake region near-the leeward
470 face. This was associated with the travel time towards the wake region of
471 the current vortex shed from the leading corner.

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