

# Experiment studies on vibro-acoustic behavior of a stiffened submerged Conical - Cylindrical Shell subjected to Force and acoustic Excitation

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## Abstract

An experimental model was made to investigate the influence of force and acoustic excitation on the vibration and underwater sound radiation of the stiffened conical-cylindrical shell. Meanwhile a coupled precise transfer matrix method (PTMM) and wave superposition method (WSM) was also proposed to analyze vibro-acoustic responses of combined shells. To test accuracy of present method, vibration and acoustic results of combined shells are firstly examined. As expected, results of present method are in excellent agreement with the ones in literature and model test. The experimental results show that free vibrations of the experimental test are consistent with the literature data and the present method's results. Forced vibration and acoustic test results are also well agreed with the numerical results from the coupled PTMM/WSM. The comparisons show that the coupled PTMM/WSM is reliable and credible to solve the vibro-acoustic response of combined shells. The analysis shows that the acoustic excitation is the key factor for radiated noise in low frequency range. However, the radiated noise resulted from force excitation is dominant in mid-high frequency band.

## Key words

A conical-cylindrical shell; vibration; sound radiation; Acoustic excitation; Experiment;

## Introduction\*

Widely used in aeronautical and naval industry, the ring-stiffened conical-cylindrical shell is one of the most normal structural of mechanical engineering, the radiated noise of which is a classic acoustic-structure interaction problem.<sup>1-2</sup> There is a lot of literature research on vibration and sound radiation of various shells (spherical shell<sup>3</sup>, cylindrical shell<sup>4</sup> and conical shell<sup>5</sup>) and some attentions on the structure dynamic analysis of conical-cylindrical shells have been made public.<sup>6-17</sup>

Regarding the numerical analysis of sound radiation of submerged elastic structure, several numerical methods such as finite element method (FEM), boundary element method (BEM) and coupled FEM/BEM are available to predict and analyze the acoustic problem. The traditional way of FEM<sup>18</sup> is to transform an infinite domain problem into a finite field problem by defining artificial boundary conditions; While because of its requirement for the Green function, BEM<sup>19</sup> is well-suited for the infinite domain problem; The vibration and sound radiation of submerged elastic structures can be predicted directly by coupled FEM and BEM.<sup>20</sup> Liu and Chen<sup>21</sup> used a coupled FEM/BEM to analyze the vibration response of submerged cylindrical shells in conjunction with the wave number theory. The effect of mass distribution and

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isolation in a submerged hull was investigated by developing a fully coupled finite element/boundary element (FE/BE) model.<sup>22</sup> The fluid-structure interaction problem can be solved by constructing the low order shape functions and generating the discrete element meshes. Due to these fundamental assumptions, the development of FEM and BEM is now limited in many prospects. As the frequency increases, the number of required elements and memory space will surge increase sharply, and computational efficiency reduces seriously in contrast.

The vibro-acoustic response of a few simple structures such as spherical shells<sup>3</sup>, cylindrical shells<sup>4</sup> can be calculated directly by the analytic method. Due to the cone vertex of conical shell<sup>5</sup>, tension-bending coupling term in the constitutive equation increases the mathematical complexity and calculation difficulty. There are many researchers paying attention to the vibration and sound radiation of the cylindrical shell, but paying attention to acoustic characteristics of combined shells is less. Caresta and Kessissoglou<sup>23</sup> analyzed a submarine's acoustic responses under harmonic force excitation. In order to improve the computational efficiency and frequencies, Wang *et al.*<sup>24</sup> developed a precise transfer matrix method (PTMM) and solved a set of first order differential equations to obtain vibro-acoustic responses of submerged stiffened combined shell and conical shells. Qu *et al.*<sup>25</sup> predicted the vibration and acoustic behaviors of circumferential and longitudinal rings stiffened submerged spherical-cylindrical-spherical shells. An improved variational method combined with a way of multi-segment partitioning was present to formulate the coupled stiffened shell model, whereas a Kirchhoff-Helmholtz integral formulation is used to simulate the exterior fluid.

The present work aims to analysis the noise and vibration results of experiment model, which was constructed to simulate a submerged stiffened conical-cylindrical shell under the acoustic and force excitation. To the authors' knowledge, there are few references concerning vibro-acoustic behavior of combined shells in force and acoustic excitation case. The authors tested the experiment model to obtain the combined shell's free vibration, forced vibration and acoustic response in air and water. The structural and acoustic responses were also determined by a coupled PTMM/WSM, which is suitable for assessing distributive systems' vibro-acoustic responses. Two models of combined shells from Refs. [11-12,23] were employed to compare with results from the coupled PTMM/WSM method verify its effectiveness, which is also proofed by comparison between the present method and experiments results. Meanwhile the vibration and sound radiation characteristics of the stiffened combined shell is investigated under different types of excitation.

## Experimental research

As shown in Figure 1a, the experimental model is a single ring-stiffened conical-cylindrical shell. And welding wasn't considered to accord with the theoretical model. The material properties of the conical-cylindrical shell are as follows: Length  $L_1=0.2113\text{m}$ ,  $L_2=0.5\text{m}$ ,  $L_o=0.5\text{m}$ , Radius  $R=0.25\text{m}$ . The combined shell thickness is  $t=4\text{mm}$ . The stiffener's sectional dimension is  $40\text{mm} \times 4\text{mm}$ , and its spacing is  $l=0.1\text{m}$ . The density is  $\rho=7850\text{kg}/\text{m}^3$ , the elastic modulus is  $E=206\text{GPa}$  and its Poisson's ratio is  $\nu=0.3$ . 15 mm thick caps are set at the ends of the experimental model. Equipped with a fixed exciter (BK 4809) and a non-directional dodecahedron sound source (BK 4292L) as primary exciting forces, the stiffened conical-cylindrical shell is approximately axisymmetric.

The vibration measurement system is composed of anechoic tank, hammer, vibration exciter, signal generator, power amplifier, data acquisition device, acceleration sensor, force sensor, and hydrophone. The anechoic tank is 8 m in length, 4 m in width and 3 m in depth. The wedge-shaped sound absorption cone is installed on the inner surfaces of the water pool. The anechoic tank and test model are shown in Figure.1c and Figure. 2a.

A patch (10 mm in diameter) screwed to a shaker is glued on the inner surface of the combined shell. The exciter is fixed on the both end caps to excite the point with cylindrical coordinates  $(x, r, \theta) = (0.25\text{m}, 0.25\text{m}, 0)$  in which  $x=0$  is at the bottom of the cylindrical shell segment. The force excitation type is sinusoidal, the amplitude of which is 1 N. There are two components between the shaker and the patch: one is an impedance head (PCB-208C02) for the measurement of acceleration and force, another is a threaded rod which allows a punctiform radial excitation acting on the shell. There is a flexible rope to connect the hoisting machine and the model. A hydrostatically balanced loading is employed to be located on the shell's top cover. Taking advantage of the axial symmetries of the combined shell, a series of measuring points along the circumferential and axial direction can be arranged to obtain the vibration acceleration. 10 acceleration sensors (PCB-352C03) are set to measure the radial vibration acceleration of the outer surface for several cases. The positions of these sensors as shown in Figure2b are located at Position1 (0.09m, 0.25m, 0), Position2 (0.17m, 0.25m, 0), Position3 (0.25m, 0.25m, 0), Position4 (0.33m, 0.25m, 0), Position5 (0.41m, 0.25m, 0), Position6 (0.625m, 0.1778m, 0), Position7 (0.25m, 0.25m,  $\pi/2$ ), Position8 (0.25m, 0.25m,  $\pi$ ), Position9 (0.625m, 0.1778m,  $\pi/2$ ) and Position10 (0.625m, 0.1778m,  $\pi$ ).

### Vibration solution for combined shells

Based on Flügge shell theory, the equations of revolutionary shell can be written in the matrix differential equations

$$\frac{d\mathbf{Z}(\xi)}{d\xi} = \mathbf{U}(\xi)\mathbf{Z}(\xi) + \mathbf{F}(\xi) - \mathbf{p}(\xi) \quad (1)$$

where  $\mathbf{Z}(\xi) = \{u_s, w_s, \varphi_s, N_s, V_s, M_s, S_{s\theta}\}^T$  is a state column vector. The quantities  $u_s, w_s, \dots$  marked with superscript score are dimensionless variables, respectively. These dimensionless state variables satisfy

$$(u_s, w_s, \varphi_s, N_s, V_s, M_s) = \sum_{\alpha=0}^1 \sum_n \left( h_o \vartheta_\alpha h_o \vartheta_\alpha \frac{h_o \vartheta_\alpha}{R}, \frac{K N_s^\alpha}{R^2}, \frac{K V_s^\alpha}{R^2}, \frac{K M_s^\alpha}{R} \right) \sin \left( n\theta + \frac{\alpha\pi}{2} \right) \quad (2a)$$

$$(v_s, S_{s\theta}) = \sum_{\alpha=0}^1 \sum_n \left( h_o \vartheta_\alpha \frac{K S_{s\theta}^\alpha}{R^2} \right) \cos \left( n\theta + \frac{\alpha\pi}{2} \right) \quad (2b)$$

where  $u_s, v_s$  and  $w_s$  are the respective displacement components in the axial, circumferential and radial directions.  $\alpha=0,1$  denotes symmetric and anti-symmetric mode.  $R, h_o$  is the radius and thickness of the cylindrical shell segment.  $K = Eh_o^3/12(1-\mu^2)$  is bending rigidity. The coefficient matrix  $\mathbf{U}(\xi)$  relates to  $\xi$  ( $\xi = s/R$  is the dimensionless variable), and the nonzero elements are given in Ref. [9].  $\mathbf{F}(\xi)$  and  $\mathbf{p}(\xi)$  are respectively the external driving forces and fluid load.

The conical-cylindrical shell can be separated into several parts, and the node coordinates are  $\xi_k, k=1, 2, 3, \dots, N-1$  respectively. So,  $\xi_k$  and  $\xi_{k+1}$  are the adjacent coordinates, and satisfies  $\xi_{k+1} = \xi_k + \Delta\xi$ . Hence, Eq (1) can be solved by the following equation

$$\mathbf{Z}(\xi_{k+1}) = \Phi_0(\Delta\xi)\mathbf{Z}(\xi_k) + \int_0^{\Delta\xi} e^{U\tau} \mathbf{r}(\xi_{k+1} - \tau) d\tau \quad (3)$$

where  $\Phi_0(l) = e^{U\Delta\xi}$  can be obtained by precise integration method.<sup>9</sup>

The effect of stiffeners on the shell is reflected on the variation of the state vector. Internal forces (moment, shear and axial force) are changed between left section  $\xi_k^L$  and right section  $\xi_k^R$  of the stiffener at a position  $\xi_k$ , at where the shell and the stiffener are connected together. The state vectors  $\mathbf{Z}(\xi_k)$  satisfy

$$\mathbf{Z}(\xi_k^R) = \mathbf{T}r_k \mathbf{Z}(\xi_k^L) \quad (4)$$

where  $\mathbf{Z}(\xi_k^L)$  and  $\mathbf{Z}(\xi_k^R)$  respectively denote state vectors of the left section and the right section.  $\mathbf{T}r_k$  is the point transfer matrix for  $k$ th stiffener.

The relation between the displacement  $u_c, v_c, w_c, \varphi_c$  of conical shell component and the displacement  $u, v, w, \varphi$  of the cylindrical shell component should keep continuous, as shown in Figure. 3. Then the displacement connection conditions at the junction can be deduced as

$$\begin{aligned} u &= u_c \cos \alpha - w_c \sin \alpha, v = v_c \\ w &= u_c \sin \alpha + w_c \cos \alpha, \varphi = \varphi_c \end{aligned} \quad (5)$$

Likewise, the relation between internal forces of conical shell component  $N_c^s, S_c^{s\varphi}, V_c^s, M_c^s$  and the cylindrical shell component  $N^s, S^{s\varphi}, V^s, M^s$  should maintain the balance, as shown in Figure. 3. The force compatibility condition at the junction can be derived as

$$\begin{aligned} N^s &= N_c^s \cos \alpha + V_c^s \sin \alpha, S^{s\varphi} = S_c^{s\varphi} \\ V^s &= -N_c^s \sin \alpha + V_c^s \cos \alpha, M^s = M_c^s \end{aligned} \quad (6)$$

Based on the internal forces and displacements connection conditions, the state vectors at the junction satisfy

$$\mathbf{Z}(s = L_{cyl}^L) = \mathbf{TP}^{con \rightarrow cyl} \mathbf{Z}(s = L_{con}^R) \quad (7)$$

where  $\mathbf{Z}(s = L_{cyl}^L)$  and  $\mathbf{Z}(s = L_{con}^R)$  respectively denote state vectors of the left section and the right section at the junction.  $\mathbf{TP}^{con \rightarrow cyl}$  denotes the  $8 \times 8$  point transfer matrix.

## Numerical solution of Acoustic response

Geometrically similar to the real boundary  $S$ , a virtual boundary  $S'$  inner the model is constructed to avoid singular integral, as shown in Figure 4. According to the potential theory, the fluid load  $p$  can be derived as

$$p(P) = \iint_{S'} \sigma(O) K(P, O) dS' \quad (8)$$

where  $O$  and  $P$  are the coordinates of the virtual source point and field point, respectively.  $\sigma(O)$  is the function of the virtual source strength density. And  $K(P, O)$  is the function of virtual source strength. The distance  $d(P, O)$  between point  $P$  and point  $O$  satisfies  $\sqrt{r_p^2 + r_o^2 - 2r_p r_o \cos \beta + (z_p - z_o)^2}$ .  $\beta = \theta_p - \theta_o$  denotes the relative angle. To avoid the non-unique problem at characteristic frequency, the distance  $\bar{d}$  is defined by complex  $(1 + i\gamma)d$ .<sup>26</sup> Based on a complex

Fourier series, the unknown distributed function  $\sigma(O)$  can be expanded as

$$\sigma(O) = \sum_{n=-\infty}^{+\infty} \sum_{m=-\infty}^{+\infty} c_{mn} e^{im\frac{2\pi}{L'}L_0} e^{in\theta_0}, n \in -\infty \sim +\infty \quad (9)$$

The source strength function  $K(P, O)$  can also use a complex Fourier series. Replace Eq. (8) by Eq. (9), divide  $M$  equal divisions from  $(0 \sim L')$  and  $N$  equal divisions from  $(0 \sim 2\pi)$

$$p(P) = \sum_{n=-\infty}^{+\infty} \sum_{m=-\infty}^{+\infty} c_{mn} K_{mn}(\beta_0) e^{in\theta_p} \quad (10)$$

where  $K_{mn}(\beta_0) = \frac{2\pi L'}{MN} \sum_{k_1=0}^{M-1} \sum_{k_2=0}^{N-1} K[\beta_0; k_1, k_2] r_{L_0} e^{ink_1 \frac{2\pi}{M}} e^{-ink_2 \frac{2\pi}{N}}$  is generalized fluid load for a prescribed wave number  $(m, n)$ .  $L'$  is the length of combined shell's meridian line. Because  $M$  and  $N$  divisions is the power series of 2,  $K_{mn}(\beta_0)$  can be calculated quickly and accurately by discrete Fourier transform method.

The normal derivative of the sound pressure can also use a complex Fourier series to expand as

$$\frac{\partial p(P)}{\partial n_p} = \sum_{n=-\infty}^{+\infty} \sum_{m=-\infty}^{+\infty} c_{mn} K'_{mn}(\beta_0; \theta_0) e^{in\theta_p} \quad (11)$$

where  $K'_{mn}(\beta_0; \theta_0) = \frac{2\pi L'}{MN} \sum_{k_1=0}^{M-1} \sum_{k_2=0}^{N-1} K'(\beta_0; k_1, k_2) r_{k_1} e^{im\frac{2\pi k_1}{M}} e^{-in\frac{2\pi k_2}{N}}$ .

Points  $\xi_i$   $i=1, 2, \dots, N$ , which are located on the meridian line of the combined shell, satisfy the boundary condition of Neumann. For a prescribed wave number  $n$ . Then the Neumann boundary condition can be derived in matrix as

$$\mathbf{K}'_n(\xi_i) \mathbf{c}_n = \rho_0 \omega^2 \mathbf{w}_n(\xi_i), n \in (-\infty \sim +\infty) \quad (12)$$

where  $\mathbf{c}_n = \{c_{-m,n} \quad L \quad c_{m-1,n} \quad c_{mn}\}^T$  and  $\rho_0$  is the surrounding fluid media's density.  $\mathbf{K}'_n = K'_{m,n}(\xi_i)$ ,  $i=1, 2, \dots, N$  and  $\mathbf{w}_n = \{w_n(\xi_1) \quad L \quad w_n(\xi_{N-1}) \quad w_n(\xi_N)\}^T$ .

The  $i$ th rib state vector  $\mathbf{Z}(\xi_i)$  should satisfy  $\mathbf{Z}(\xi_i^R) = \mathbf{T}_i \mathbf{Z}(\xi_i^L)$ .  $\mathbf{T}_i$  denotes the point transfer matrix for  $i$ th rib. After considering the effect of ribs, then we can get

$$\mathbf{Z}(\xi_{i+1}) = \mathbf{Z}_{i+1} \mathbf{Z}(\xi_i) + \mathbf{P}_{i+1}, i=1, \dots, N-1 \quad (13)$$

where  $\mathbf{T}_{j+1} = \exp[\mathbf{U}(\xi_{j+1} - \xi_j)]$ ,  $\mathbf{P}_{j+1} = \int_{\xi_j}^{\xi_{j+1}} \exp[\mathbf{U}(\xi_{j+1} - \tau)] \mathbf{r}(\tau) d\tau$ .

The governing equations of the combined shell can be integrated by gathering Eq. (13). Based on boundary conditions, find the line numbers of determinate state vectors in  $\mathbf{Z}(\xi_1)$ ,  $\mathbf{Z}(\xi_N)$ , and delete relevant rows in the coefficient matrix, the governing equations can be got as

$$\begin{bmatrix} -\mathbf{T}_2 & \mathbf{I}_8 & 0 & 0 & 0 & 0 \\ 0 & -\mathbf{T}_3 & \mathbf{I}_8 & 0 & 0 & 0 \\ 0 & 0 & -\mathbf{T}_4 & \mathbf{I}_8 & 0 & 0 \\ 0 & 0 & 0 & \dots & \mathbf{I}_8 & 0 \\ 0 & 0 & 0 & 0 & -\mathbf{T}_N & \mathbf{I}_8 \end{bmatrix} \begin{Bmatrix} \mathbf{Z}(\xi_1) \\ \mathbf{Z}(\xi_2) \\ \mathbf{Z}(\xi_3) \\ \mathbf{M} \\ \mathbf{Z}(\xi_N) \end{Bmatrix} = \begin{Bmatrix} \mathbf{P}_2 \\ \mathbf{P}_3 \\ \mathbf{P}_4 \\ \mathbf{M} \\ \mathbf{P}_N \end{Bmatrix} \quad (14)$$

The state vectors  $\mathbf{Z}_n^f(\xi)$  can be obtained under the generalized sound pressure  $K_{mn}$  or external load  $\mathbf{F}(\xi)$  directly. The determinant of the coefficient matrix in Eq (14) can be expanded to obtain the dispersion equation.

According to the linear superposition method, the radial displacement on the surface  $S$  can be described as

$$w_n(\beta_0) = w_n^f(\beta_0) + \sum_{m=-\infty}^{+\infty} c_{mn} w_{mn}^p(\beta_0) \quad (15)$$

where  $w_n^f(\beta_0)$ ,  $w_{mn}^p(\beta_0)$  are respectively the state vectors under the external load  $\mathbf{F}(\beta_0)$  and generalized sound pressure  $K_{mn}$ .

After replacing Eq. (12) by Eq. (15) and selecting  $q$  ( $q > 2m+1$ ) collocation points on the meridian of the surface  $S$ , several linear algebraic equations can be described as

$$\mathbf{U}_n \mathbf{c}_n = \mathbf{Q}_n, n \in (-\infty \sim +\infty) \quad (16)$$

where  $\mathbf{U}_{q \times 2m+1} = \mathbf{K}_{mn}'(\beta_{s_j}) - \rho_0 \omega^2 w_{mn}^p(\beta_{s_j})$ ,  $\mathbf{Q}_{q \times 1} = \rho_0 \omega^2 w_n^f(\beta_{s_j})$ . Unknown coefficient vector  $\mathbf{c}_n$  can be calculated directly by Moore-Penrose generalized inverse method. Finally, the vibration and acoustic responses of the submerged combined shell can be solved the coupled PTMM/WSM simply and effectively over a wide frequency range.

## Results and discussion

### Free vibration results and discussions

To verify the accuracy of the coupled PTMM/WSM method, A conical-cylindrical shell in Ref. [23] with free-free boundary conditions was employed, of which the parameters are as follows:  $L_{start} \sin \alpha / R = 0.08453$ ,  $L = 0.2\text{m}$ ,  $L/R = 1$ ,  $\alpha = 30^\circ$ ,  $R = 1\text{m}$ ,  $h = 0.002\text{m}$ ,  $\rho = 7800\text{kg/m}^3$ ,  $\mu = 0.3$ ,  $E = 210\text{GPa}$ . The comparison of the frequency parameter values  $\lambda = \omega R \sqrt{\rho(1-\nu)^2 / E}$  is given to illustrate the feasibility and effectiveness of the presented method. As shown in Table 1, the results calculated by the coupled PTMM/WSM are compared with corresponding results in Ref. [23]. The maximum error margin between the literature data and the present method is within +/- 2%.

Another conical-cylindrical shell model in Ref. [11] was also borrowed to assess the precision and accuracy of this method. The parameters of the combined shell are as followed:  $\alpha = 18^\circ$ ,  $L = 30\text{m}$ ,  $R = 1\text{m}$ ,  $h = 0.04\text{m}$ . The parameters of the ring stiffener are: width  $b = 0.06\text{m}$  and depth  $d = 0.08\text{m}$ . And there are 7 rings in each cylindrical shell segment. The material of shells and ring stiffeners are steel with  $\rho = 7800\text{kg/m}^3$ ,  $\mu = 0.3$ ,  $E = 210\text{GPa}$ . The results obtained by modified variational approach in literature<sup>11,12</sup> and experimental results are shown in Table 2. The maximum error margin between the results in Ref. [12] and the present method is +/- 2% except the value at ( $n=2$ ,  $m=1$ ).

Stuck to the measuring points, acceleration sensors relate to the data acquisition system. And a force hammer is used to thump structure model. The free vibration of the experiment model hoisted by a flexible rope is carried out by an impact hammer (PCB 086C03) and several accelerometers (PCB 353B15) to obtain the natural frequencies and structural modes by the vibration response analysis processing module (HBM Quantumx1615B). To verify the accuracy of vibration responses, the natural frequencies values calculated by the present method are compared with the experimental results. The first 10 orders of the natural frequencies values are given in Table 3, which include the numerical results and experimental results.

### ***Forced vibration results and discussion***

The forced vibration tests under the exciter and the non-directional dodecahedron sound source excitation is respectively carried out in air and water condition. During this process, a sinusoidal single frequency signal is generated as the input signal. To amplify the signal, the power amplifier is driven with the linear sweep frequency (100Hz-2500Hz). The data measured by acceleration sensors is transformed into vibration acceleration level by formula  $L_a = 20\lg(a/a_0)$ , where  $a$  is the test value and  $a_0$  is the reference acceleration value  $1e-6 \text{ m/s}^2$ .

In the air condition, the ten measured points' acceleration data was disposed to obtain the acceleration level in the frequency band. The coupled PTMM/WSM is also employed to calculate the vibration response of the experimental model. The boundary conditions of the both ends are supposed to be clamped. The vibration response under the force excitation is shown in Figure 5. Apparently the variation tendency of the both frequency response curves are consistent. In the 100~500Hz band, the vibration response curves predicted by the coupled PTMM/WSM coincided very well with experimental curves, except some loss of resonance peaks. It's because that the peak frequency 141Hz and 224 Hz is corresponding to the nature frequencies of the end cap. Although the both vibration response curves have a little difference at the resonance peak in the range of 500~2000Hz, the results from the present method are still consistent with experimental values in the air case. In the frequency up to 2500Hz, the resonance peak numbers of test model are more than that of the coupled PTMM/WSM, and acute fluctuation also appears in phase advance and peak position. In authors' view, the difference between two curves is partly because of the local vibration of the end caps and seal flanges. Also, the conditions of actual boundary cannot be perfectly like the clamped boundary. But in general, the results show that the present method is reliable, from which the results are also credible.

As shown in Figure 6, the forced vibration responses of the submerged conical-cylindrical shell were also carried out in frequency of 100-2500Hz. In both air case and water case, the first ten natural frequencies (see Table 3) are in good agreement with the peak frequencies of forced responses curves (see Figure 5 and Figure 6). Although there is some obvious difference between two curves in the 100~600Hz band, the vibration variation trend predicted by the coupled PTMM/WSM is basically consistent with experimental values. The existence of water leads to a strong structure-acoustic coupling interaction. The experimental results have some resonance peaks resulting from global vibration modes of entire test model and local vibration modes of end cap and the sealing flange. However, the numerical model does not contain these vibration modes. In the 600~1500Hz band, the results from the coupled PTMM/WSM are consistent with

experimental results. The experiment results and the numerical results still are a little different at the peak frequencies. In the 1500~2500Hz range, the peak numbers of experiment model are more than that of numerical model. There is an obvious difference between the both frequency response curves, partly because of the local vibration of end cap and model. As a group, the results from the present method are consistent with the vibration trend of the experimental results.

### ***Radiated noise results and discussion***

As shown in Figure 2, before submerging into the anechoic tank, the conical-cylindrical shell model and the two end caps are sealed with water sealant. The signal generator generated the input signal, and drove the shaker and the non-directional dodecahedron sound source to work. The surrounding fluid is water with the density of  $1000\text{kg/m}^3$  and the sound velocity of  $1500\text{m/s}$ .<sup>27</sup> Then the noise from the vibrating surfaces of the conical-cylindrical shell was radiated into the water. As seen in Figure 2b, there are 3 hydrophones (BK-8104) arranged 1.0 m far from the surface outside the shell to measure the sound pressure in the anechoic tank. In the case of force and acoustic excitation with the linear sweep frequency (100 Hz-2500Hz), the data is transformed into a sound pressure level  $L_p = 20\lg(p/p_0)$ , in which  $p$  is the experimental test data and  $p_0$  is the reference pressures  $1 \mu\text{Pa}$ .

Comparison between the experimental results and numerical results of measuring points (hydrophone 1, hydrophone 2 and hydrophone 3) under the force excitation is shown in Figure 7. Although the both vibration response curves have a little different at the resonance peak in the range of 500~2500Hz, the sound pressure results from coupled PTMM/WSM are consistent with experiment results. When air is full in the internal medium of the test model, the coupling of acoustic cavity mode and structural mode has little effect on the vibro-acoustic response of the conical-cylindrical shell. The sound radiation computed by the coupled PTMM/WSM is related to the structural vibration mode with force excitation. It suggests that the coupling of acoustic cavity mode and structural vibration mode can be neglected. The results from the coupled PTMM/WSM miss some resonance peaks compared with the experiment results in the frequency of 1000-1500 Hz. As the investigation is performed, the analysis result of research model is totally corresponding with the previous conclusion.

Comparison of the experimental results and numerical results on measured points under the acoustic excitation is shown in Figure 8. The comparison in the force and acoustic excitation case is also shown in Figure 9. Although the overall trend of numerical results from the coupled PTMM/WSM is consistent with that of experimental results, there is some difference of peak frequencies in low frequency band. That's because the experimental model has a reverberant field and structure-acoustic coupling. The structure-acoustic coupling between interior acoustic cavity modes and structural vibration modes has some effect on the sound radiation of the combined shell in low frequency band.

As shown in Figure 10, sound pressure values of all measured points are compared under different cases of excitations. There is obvious difference in peak numbers and frequencies between acoustic excitation case and force excitation case. The main difference in both cases is that some peak values only exist in the low frequency range of force excitation case while some peak values only exist in the middle frequency range of acoustic excitation case. This suggests that the sound pressure's peak value in the force excitation case is relate to the structural natural frequencies. The peak values of sound



pressure in acoustic excitation case are related to the natural frequencies of the shell and internal cavity. On the whole, it is shown that the acoustic excitation is the key factor for radiated noise in low frequency band. However, the radiated noise resulted from force excitation is dominant in mid-high frequency band. The variation tendency of sound pressure level under force excitation is in agreement with the results under force and acoustic excitation. In 300-1300Hz frequency range, in spite of little difference in peak values, sound pressure levels are of the same order under force and acoustic excitation.

## **Conclusion**

In this paper, the experimental test of a ring-stiffened conical-cylindrical shell on the vibration and sound radiation is investigated under different types of excitations. The test of vibration and acoustic responses under different excitations are carried out and the experimental data are analyzed. The coupled PTMM/WSM is used to solve the vibration acoustic response of conical-cylindrical shell in air case and water case and then compared with the experimental results. It can be concluded as follows:

- Compared the numerical results including the vibration and acoustic responses with the literature data and experimental results, it shows that coupled PTMM/WSM is feasible to solve the vibration and acoustic problem of the submerged combined shell under cases of excitations.
- The forced vibration of the submerged conical-cylindrical shell in the water case has little difference compared to the air case in the low frequency range. The vibration values of the conical section of the combined shell in water case are significantly less than that of air case. Fluid load is the primary factor that can affect the circumferential mode and local mode of conical section.
- The peak values of sound pressure in force excitation case are relate to the natural frequency of the shell. The peak values of sound pressure in the acoustic excitation case are relate to the natural frequency of the shell and interior cavity.
- The acoustic excitation is the key factor for radiated noise in low frequency band. However, the radiated noise resulted from force excitation is dominant in mid-high frequency band.

## **Declaration of conflicting interests**

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## **Table Captions list**

**Table 1.** Comparison of frequency parameters of the model in Ref. [23]

**Table 2.** Comparison of frequency parameters of model in Refs. [11-12]

**Table 3.** Comparison of natural frequencies of the conical-cylindrical shell in air case

**Table 1.** Comparison of frequency parameters of the model in Ref. [23]

$(n,m)$	Caresta [23]	Present	error%
(1,1)	0.720025	0.7191	0.32
(2,1)	0.010026	0.0102	1.15
(2,2)	0.031057	0.0308	0.93
(3,1)	0.025699	0.0256	0.9
(3,2)	0.081591	0.0815	1.98
(4,1)	0.046447	0.0464	0.75
(4,2)	0.144166	0.1441	1.84
(4,3)	0.358169	0.3581	1.35
(4,4)	0.464651	0.4646	0.2
(5,1)	0.072717	0.0727	0.48
(5,2)	0.199828	0.1998	0.81
(5,3)	0.352868	0.3528	0.07
(5,4)	0.388592	0.3885	1.3

**Table 2.** Comparison of frequency parameters of model in Refs. [11-12]

$nm$	Free-Free				Camped-Camped				
	Chen [11]	Qu [12]	Present	error%	Chen [11]	Qu [12]	Present	error%	
0	1	0.1593	0.1596	0.1595	0.06	0.0576	0.0577	0.0575	0.35
	2	0.2309	0.2303	0.2301	0.09	0.1518	0.1521	0.1520	0.07
	3	0.3162	0.3169	0.3161	0.25	0.1818	0.1822	0.1821	0.05
	4	0.4287	0.4293	0.4291	0.05	0.333	0.3337	0.3331	0.18
	5	0.4674	0.4684	0.4676	0.17	0.3448	0.345	0.3446	0.12
	6	0.5777	0.579	0.5768	0.38	0.4863	0.4873	0.4863	0.21
1	1	0.0818	0.0819	0.0820	0.12	0.0345	0.0345	0.0344	0.29
	2	0.1774	0.1776	0.1775	0.06	0.1023	0.1026	0.1024	0.19
	3	0.2766	0.2776	0.2765	0.40	0.1863	0.1867	0.1863	0.21
	4	0.3708	0.3719	0.3706	0.35	0.2791	0.2795	0.2792	0.11
	5	0.4596	0.4605	0.4602	0.07	0.3741	0.3744	0.3741	0.08
	6	0.5369	0.5373	0.5368	0.09	0.4617	0.462	0.4618	0.04
2	1	0.0235	0.0276	0.0237	14.13	0.0249	0.029	0.0256	11.72
	2	0.0394	0.0405	0.0398	1.73	0.0633	0.0653	0.0642	1.68
	3	0.0477	0.0484	0.0479	1.03	0.1136	0.1147	0.1136	0.96
	4	0.0771	0.0792	0.0781	1.39	0.1547	0.1553	0.1548	0.32
	5	0.1369	0.1383	0.1388	0.36	0.1915	0.1918	0.1917	0.05
	6	0.2022	0.2033	0.2022	0.54	0.243	0.2432	0.2431	0.04

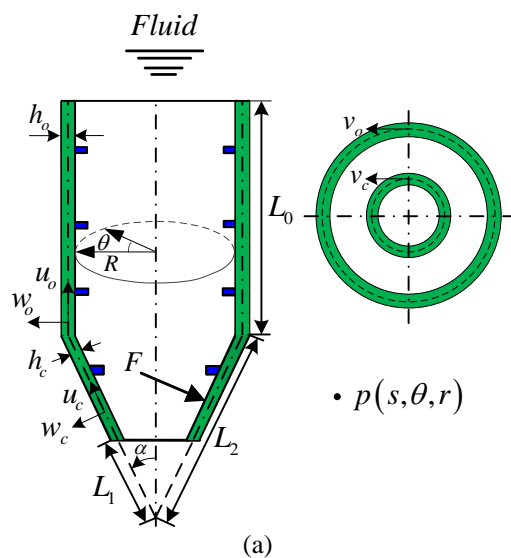
**Table 3.** Comparison of natural frequencies of the conical-cylindrical shell in air case

Order number	1 (2,1)	2 (3,1)	3 (1,2)	4 (1,3)	5 (1,4)
Experiment	647.6	847.4	948.5	1006.1	1179.0
Present	673.8	865.2	929.5	964.9	1249.9
%Error	3.9	2.1	2.0	4.3	5.7
Order number	6 (2,4)	7 (3,3)	8 (3,4)	9 (3,5)	10 (2,5)
Experiment	1278.4	1459.1	1516.8	1609.3	1785.7
Present	1297.6	1448.1	1479.1	1608.1	1778.1
%Error	1.5	0.8	2.5	0.1	0.4

## Figure Captions

- Figure 1.** Pictures of an axisymmetric stiffened conical-cylindrical shell: (a) the schematic diagram; (b) test models; (c) anechoic tank with a running gear
- Figure 2.** Experimental setup for the stiffened conical-cylindrical shell submerged in water: (a) The measurement excitation system; (b) The layout of measuring points
- Figure 3.** Sign convention for the forces, moments and displacements of the cone and cylinder
- Figure 4.** Surface boundary and virtual boundary of the conical-cylindrical shell
- Figure 5.** The forced vibration response of the conical-cylindrical shell in the air: (a) position1, (b) position2, (c) position3, (d) position4, (e)position5, (f)position6, (g)position7, (h) position8, (i) position9 and (j)position10
- Figure 6.** The forced vibration response of the conical-cylindrical shell in the water: (a) position1, (b) position2, (c) position3, (d) position4, (e)position5, (f)position6, (g)position7, (h) position8, (i) position9 and (j)position10
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- Figure 8.** Comparison of sound pressure level of the measurement point with acoustic excitations: (a) hydrophone 1, (b) hydrophone 2, (3) hydrophone 3
- Figure 9.** Comparison of sound pressure level of the measurement point with force and acoustic excitations: (a) hydrophone 1, (b) hydrophone 2, (3) hydrophone 3
- Figure 10.** Comparison of sound pressure level of the measurement point with different excitations: (a) hydrophone 1, (b) hydrophone 2, (3) hydrophone 3

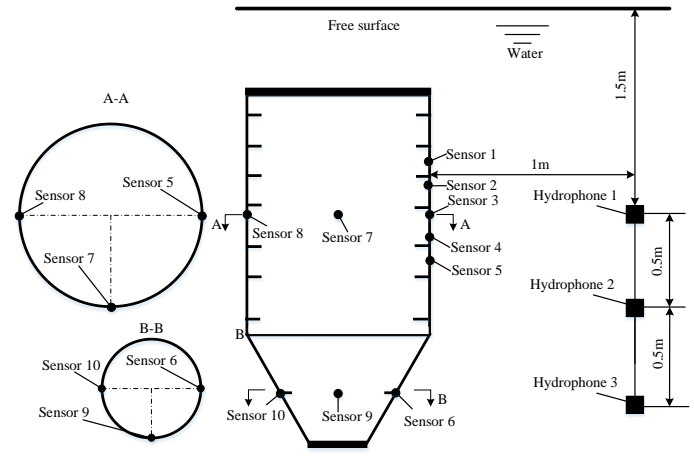




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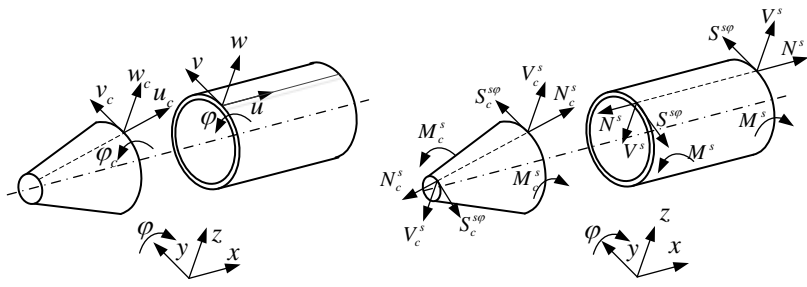


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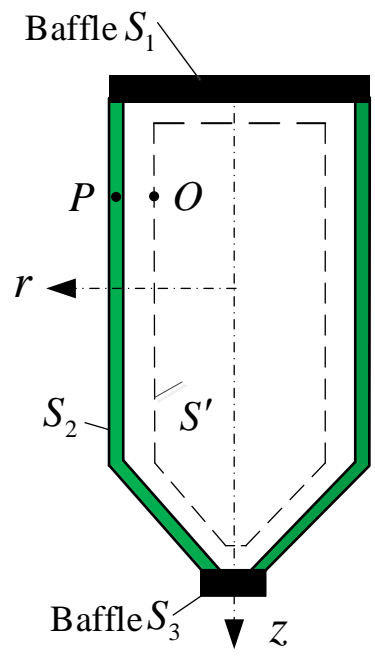


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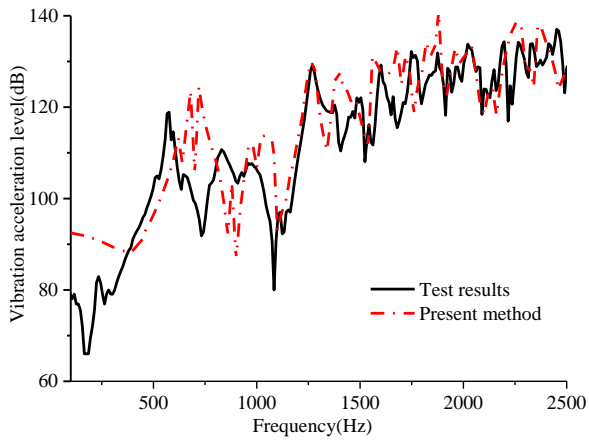
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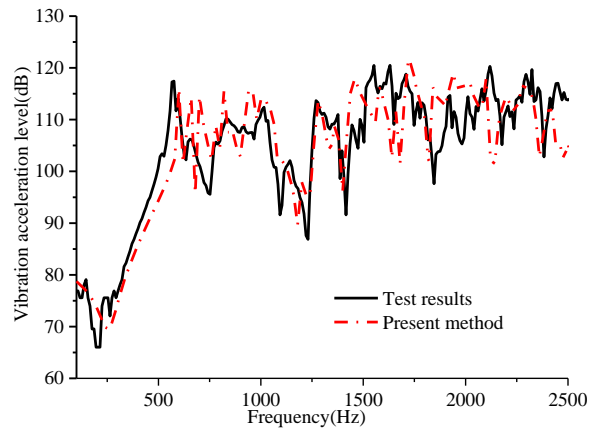
**Figure 3.** Sign convention for the forces, moments and displacements of the cone and cylinder



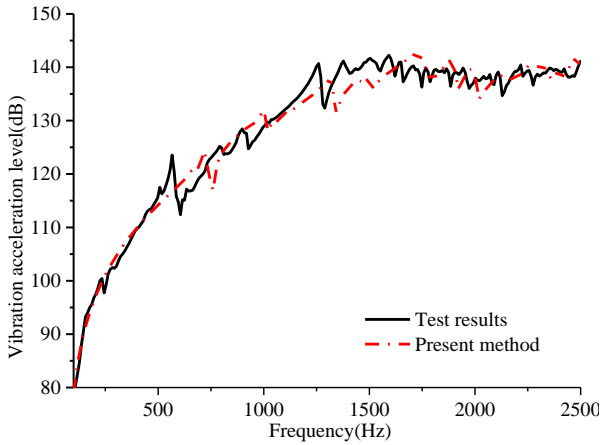
**Figure 4.** Surface boundary and virtual boundary of the conical-cylindrical shell



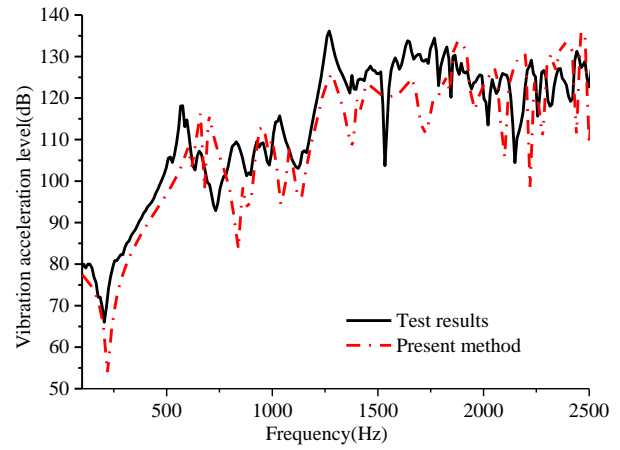
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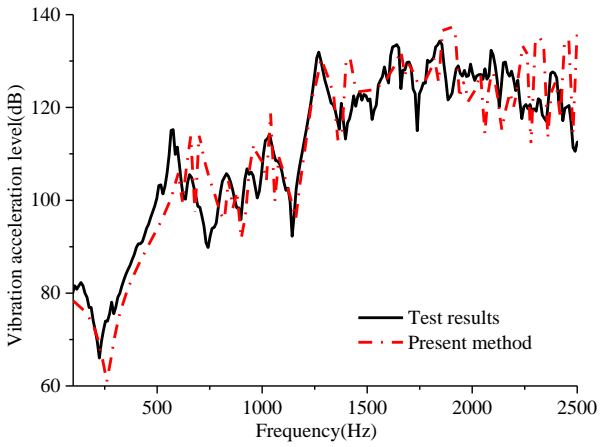
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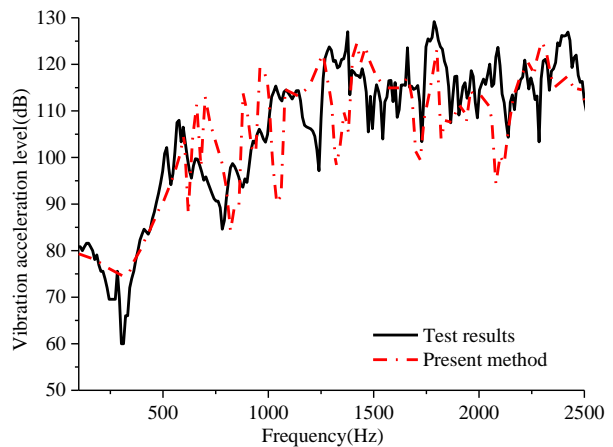
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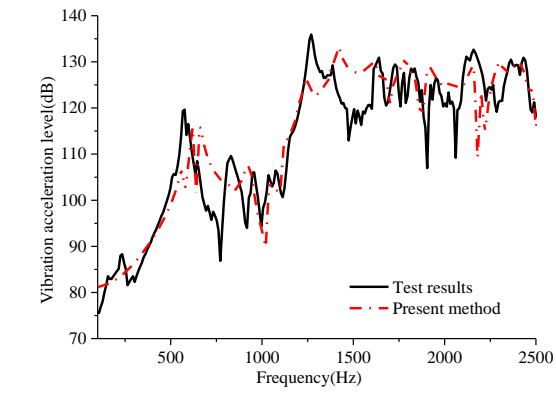
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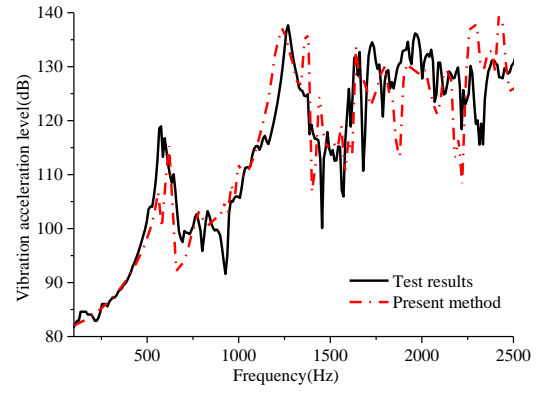
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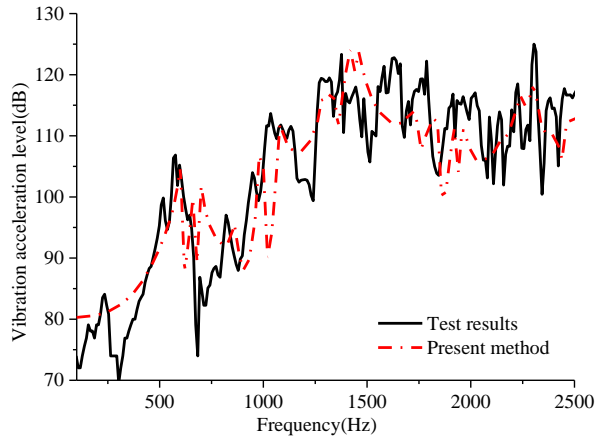
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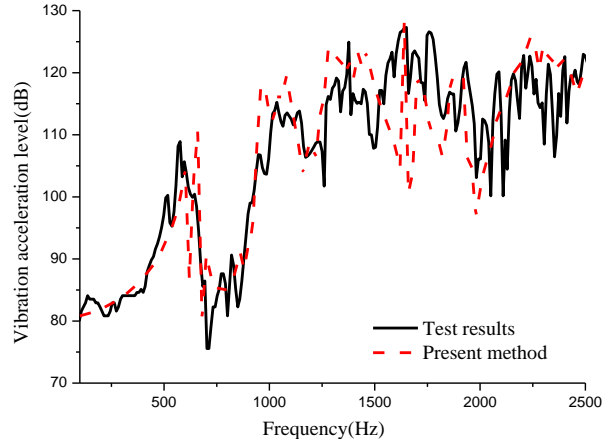
(g)



(h)

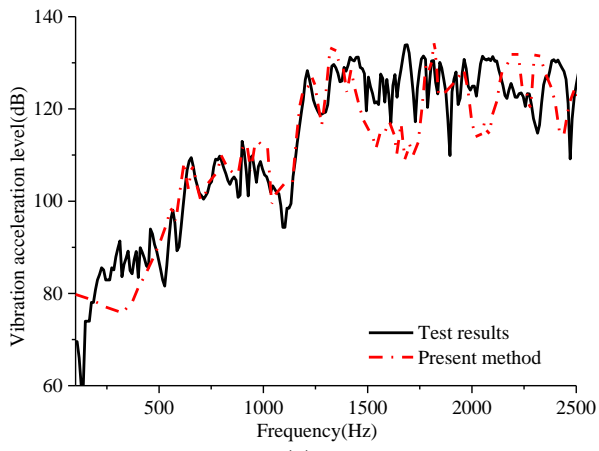


(i)

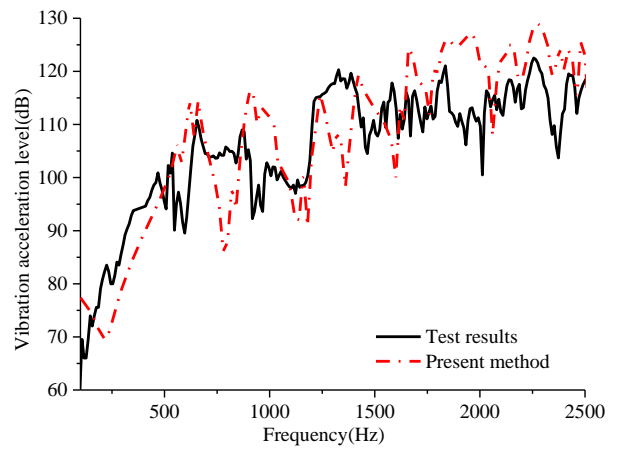


(j)

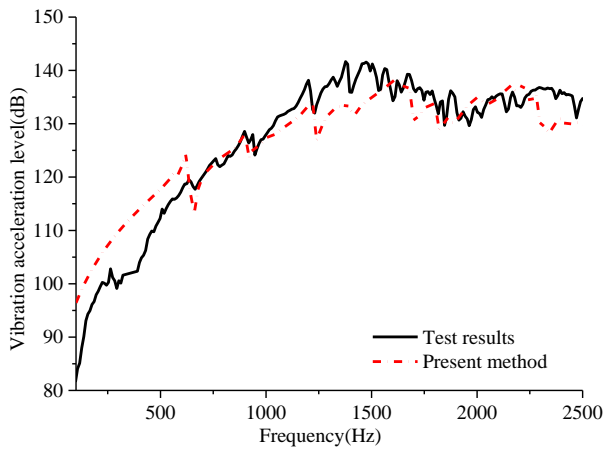
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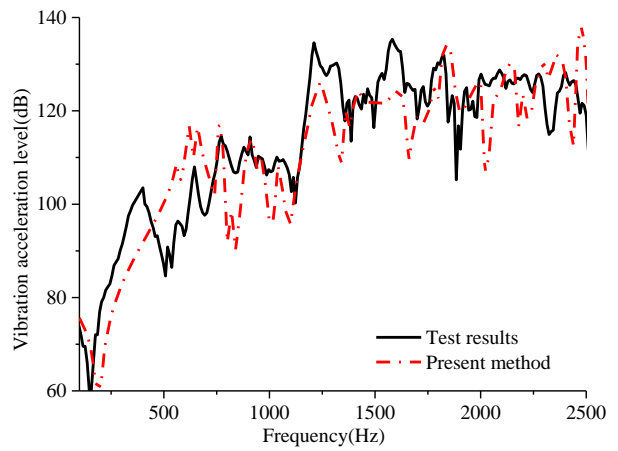
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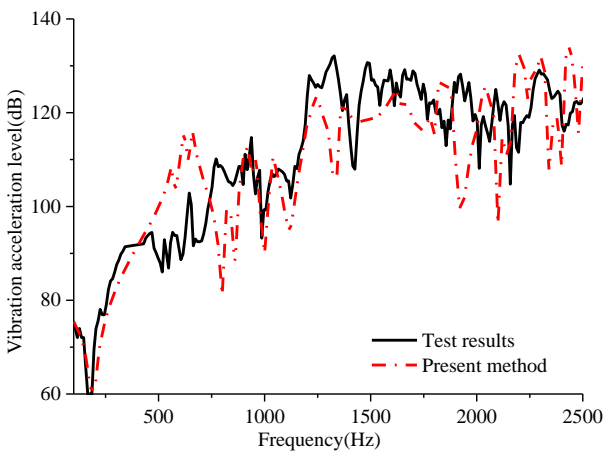
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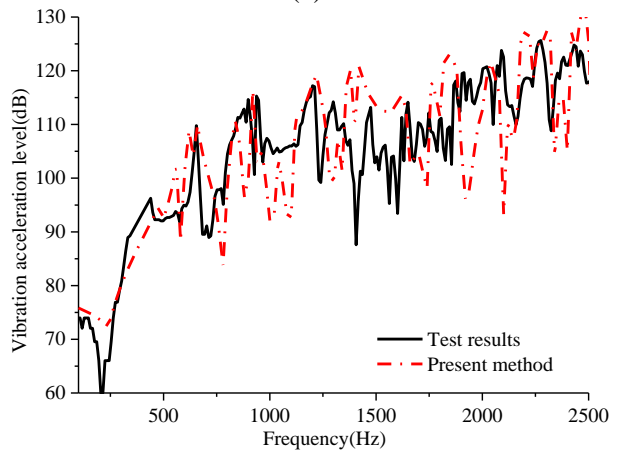
(c)



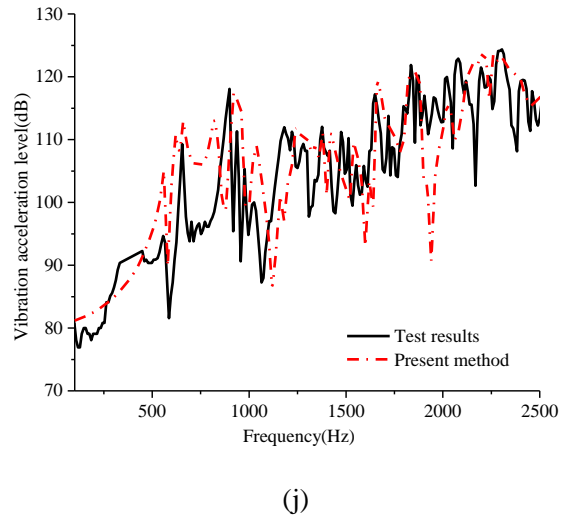
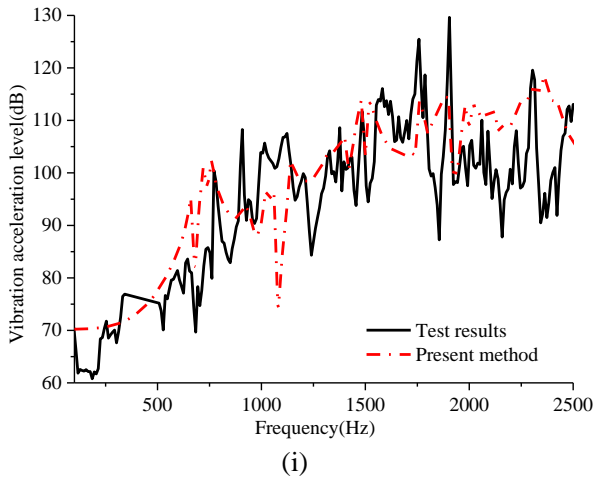
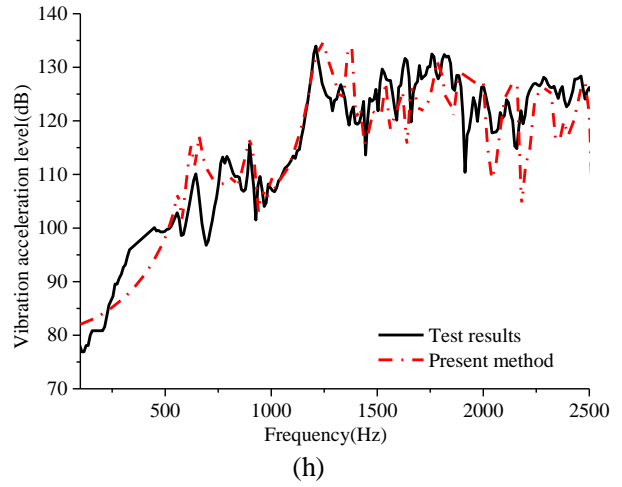
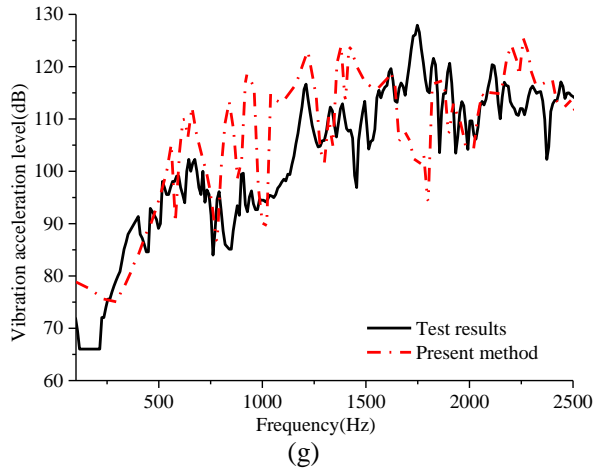
(d)



(e)

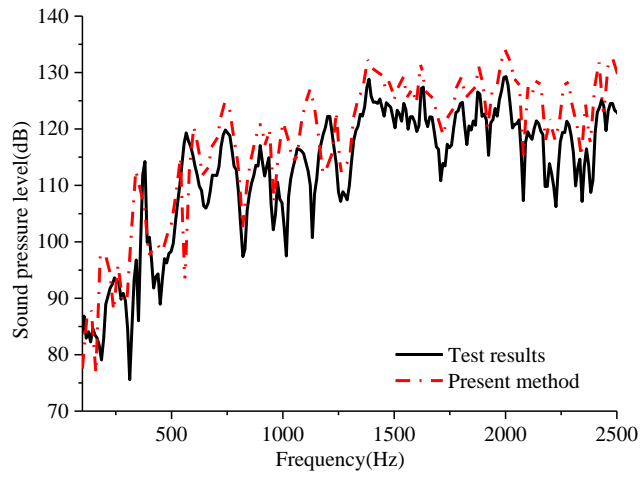


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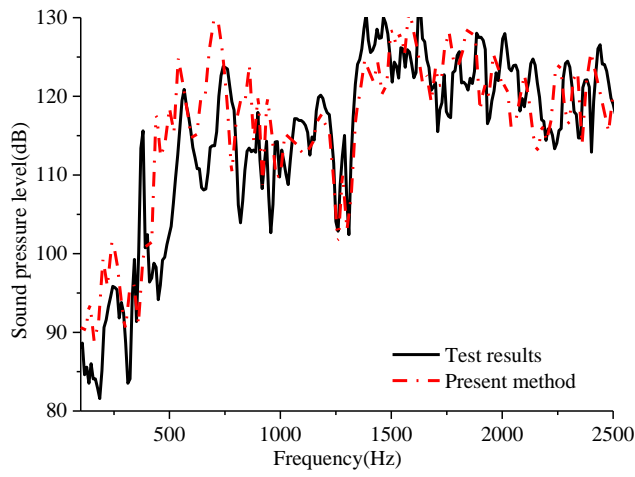


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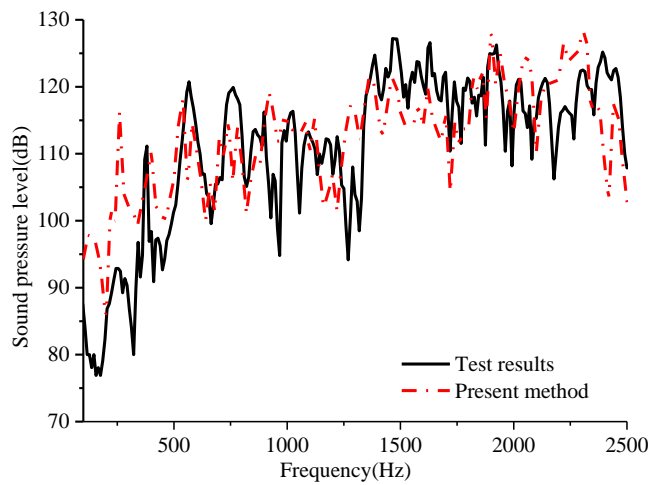




(a)

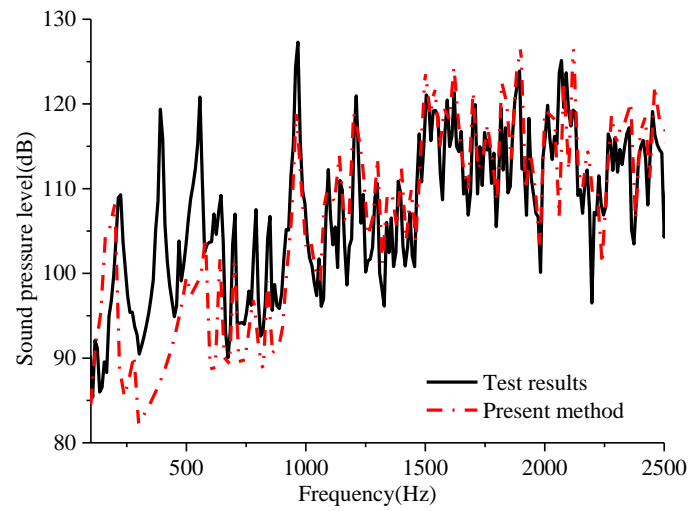


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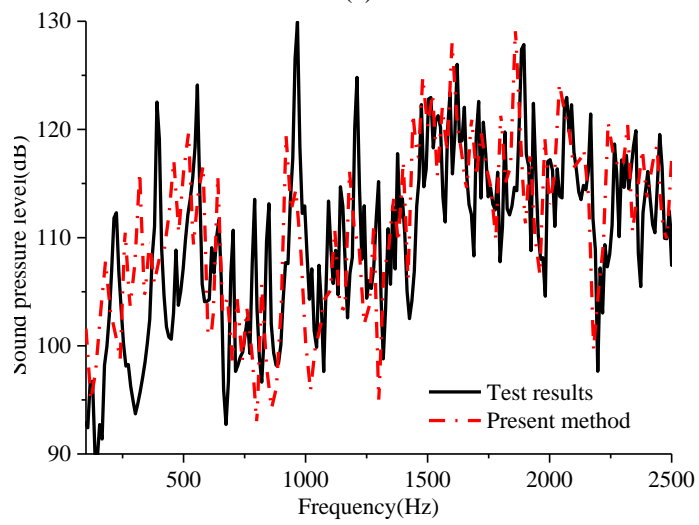


(c)

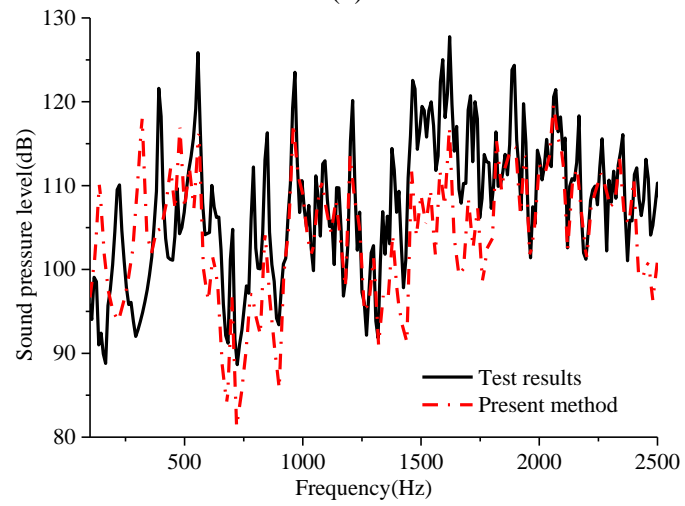
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(a)

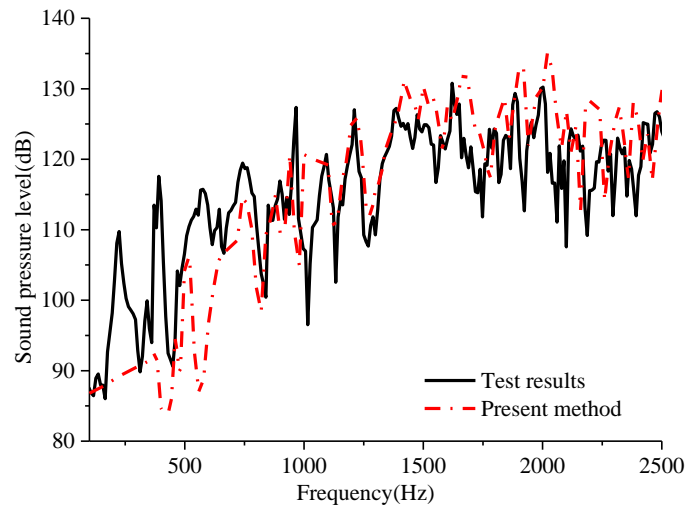


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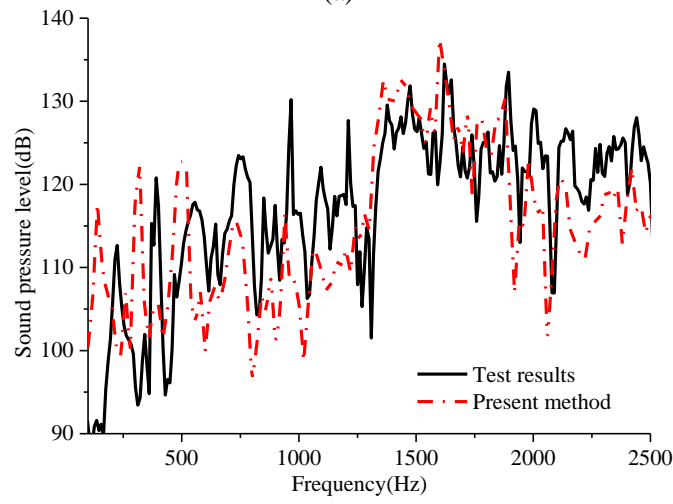


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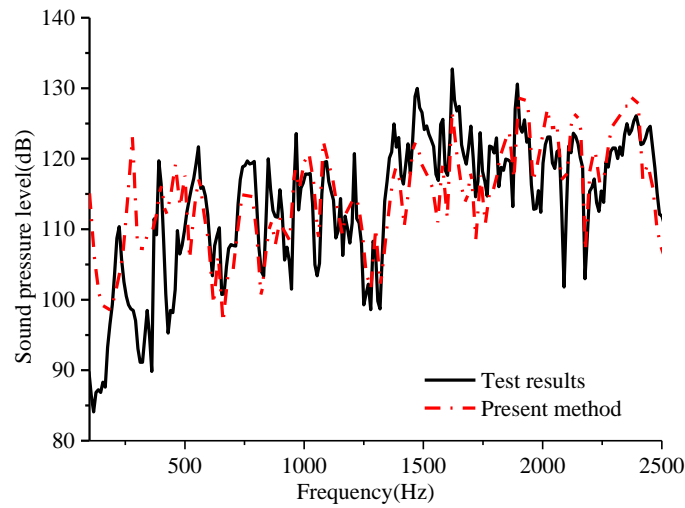
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(a)

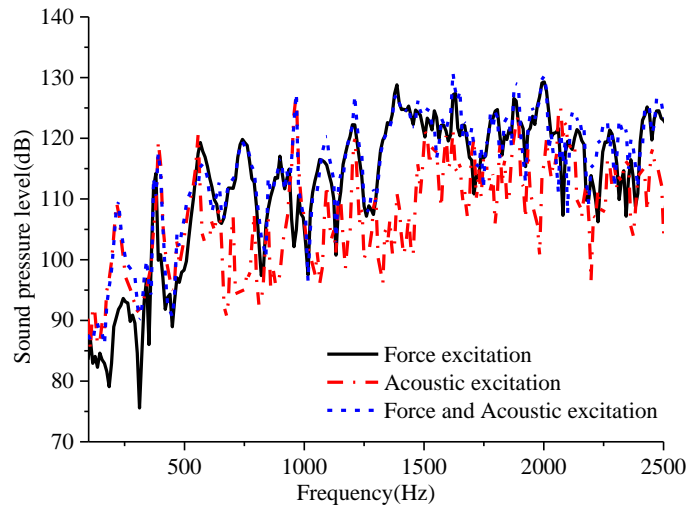


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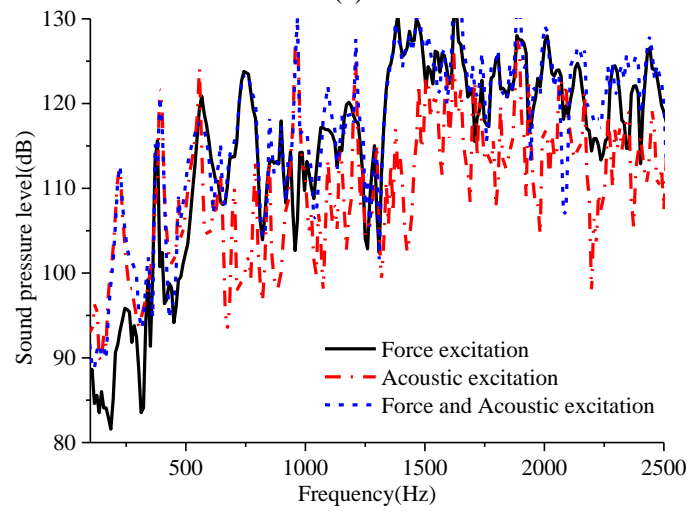


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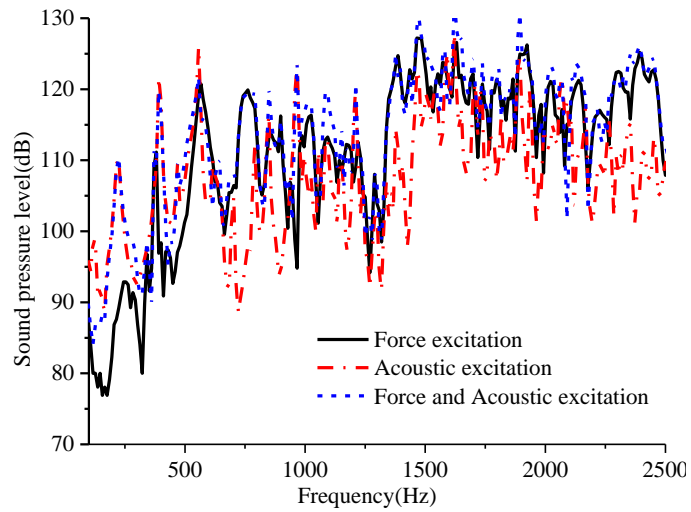
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(a)



(b)



(c)

**Figure 10.** Comparison of sound pressure level of the measurement point with different excitations: (a) hydrophone 1, (b) hydrophone 2, (c) hydrophone 3