

# Autonomous landing of underwater vehicles using high resolution bathymetry

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## Abstract

The ability to land on the seafloor expands the envelope of tasks that underwater vehicles can carry out during survey and inspection. However, even though remotely operated vehicles routinely land during their operations, autonomous underwater vehicles (AUVs) lack the sensing and data processing capabilities needed to identify safe, stable landing sites. Here, an algorithm is developed that uses mm-resolution bathymetry to detect regions where an AUV of known geometry can safely and stably land on the seafloor. The algorithm uses physical models that consider vehicle geometry, seafloor slope, roughness, friction and currents. It can identify the most suitable of multiple candidate sites based on a landing cost function. The performance of the algorithm is evaluated using seafloor bathymetry data that was obtained using an AUV equipped with a high resolution laser mapping system on the slopes of the Takuyo Daigo seamount in the Northwest Pacific. The algorithm successfully identified multiple landing sites along a 500m transect on the slopes of the surveyed seamount. The study demonstrates that safe, reliable AUV landing operation is feasible in actual seafloor environments.

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## Index Terms

Autonomous landing, Seafloor observations, Seafloor mapping, Structured light, Autonomous Underwater Vehicle

## I. INTRODUCTION

The use of unmanned underwater vehicles for exploration of mineral deposits [1], such as manganese crusts [2], manganese nodules [3] and seafloor massive sulfides [4] has gained momentum in recent years. While high resolution bathymetric maps of the seafloor generated using acoustic [5] or visual [6] mapping systems are useful for recognizing visible and morphological seafloor features [7], [8], measurement of chemical composition [9] or frictional coefficient [10] of seafloor deposits require direct contact for obtaining measurements. There have been significant developments over the past decade that provide in-situ methods to make measurements of the chemical and geological properties of the seafloor, such as underwater microscopy [11], gamma radiation measurements [12], [13], laser induced breakdown spectroscopy (LIBS) [14], laser Raman spectroscopy [15], [16] and seafloor stiffness and frictional coefficients [17]. The development of these new classes of analytical sensors that requires direct contact motivates the development of landing capabilities for Autonomous Underwater Vehicles (AUVs) to deliver these capabilities in a more scalable manner. The underwater terrain however, can change abruptly on spatial scales that cannot be observed from the surface. Therefore the reliable use of in-situ instruments such as those described, and the safety of the underwater vehicle requires real-time detection of suitable landing sites. Although remotely operated vehicle (ROV) pilots routinely identify safe landing sites and perform manipulations or in-situ chemical measurements with the instruments described, AUVs currently lack the sensing and data processing capabilities needed for these tasks.

In this research, we developed a framework to enable an underwater vehicle to autonomously identify safe landing sites based on in-situ measurements. The design concept of an underwater vehicle is proposed, identifying the hardware modifications needed for landing of AUVs on the seafloor. This includes a landing skid, and a mm-resolution laser mapping system used to detect safe landing areas. The conditions for safe landing are identified and used to develop an algorithm that uses the mm-resolution bathymetry information to identify landing areas considering the geometry and righting moment of the AUV. The algorithm detects safe landing sites along different headings, selecting the most suitable landing site based on a cost function that takes

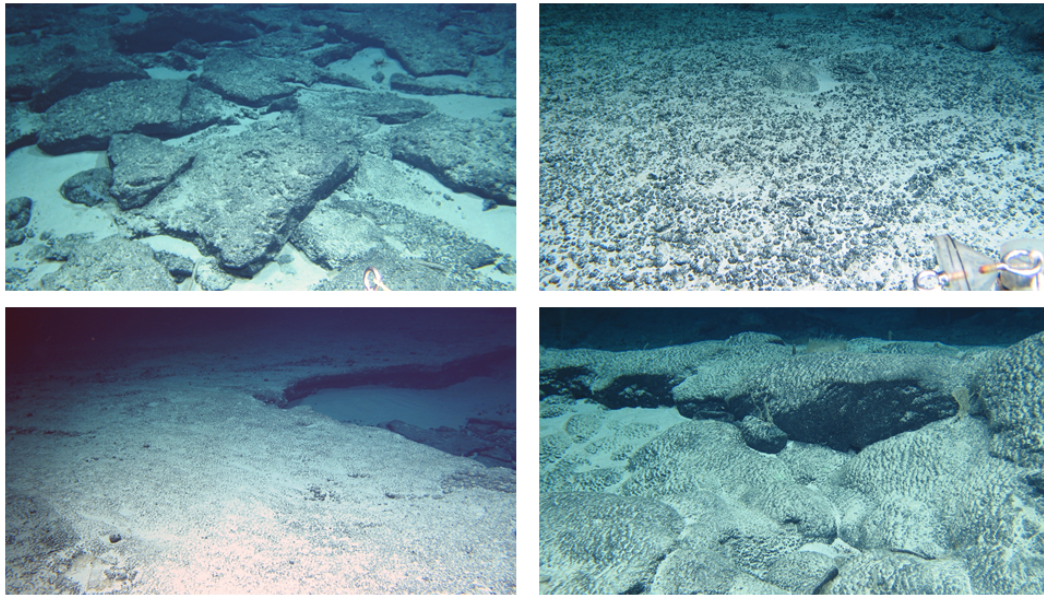


Fig. 1. Different seafloor terrain at Takuyo Daigo seamount [18]. Clockwise, from top left: Broken slabs in sand, nodules, pillowry crusts and continuous flat crusts

into account the slope and rugosity of the seafloor. The performance of the algorithm is verified by analysing more than  $750 \text{ m}^2$  of mm-resolution seafloor bathymetry obtained by an AUV along a 500 m transect on the slopes of the Takuyo Daigo seamount (located in the Northwest-Pacific) at an average depth of 1400 m. The results demonstrate the feasibility of safe autonomous landing in real seafloor terrains.

The complexity of seafloor topology (see Fig. 1) prohibits vehicles from simply landing at random locations and requires the identification and intelligent choice of landing sites. At the same time, the spatial scales relevant to AUV landing are too small to be observed by ship-board acoustic multibeam. Therefore, in order for an AUV to identify landing sites during its survey, it should be equipped with a mapping system capable of generating bathymetry with sufficiently high resolution, e.g. using light sectioning [19], [20], [21]. Even though landing site detection for aerial vehicles has been studied in [22], [23], the conditions essential for safe landing in underwater environments has not been sufficiently investigated. Simulations for control, navigation and dynamics of AUVs with landing capabilities have been reported in [24], [25]. However, these previous works do not develop the sensing and data processing methods needed to automatically identify areas where a vehicle can land safely. Regarding methods to analyse seafloor terrains, Fourier analysis based segmentation and 3D alignment was described in

[26], [27]. Other methods for surface classification using wavelets [28] have also been described, but have not been applied to landing site identification. Early works by our group demonstrated a landing algorithm using Fourier analysis to separate flat ground surface from objects on the seafloor [29]. The algorithm rejected all protruding objects as non-landable areas, and only considered flat regions for landing. This work builds on our previous studies, identifying the geometric conditions where it is possible to land on protruding objects and to land on slopes, considering the righting moment of the vehicle

The remainder of this paper is organized as follows; Section II describes the hardware requirements for landing, including the conceptual design of an underwater vehicle capable of landing and its high resolution mapping system for generating bathymetry with mm-resolution. In Section III, the different steps of the algorithm to identify landing sites are described and demonstrated by simulating its performance on seafloor data obtained using an equivalent high resolution mapping system. Section IV applies the algorithm to more than 750 m<sup>2</sup> of seafloor bathymetry obtained using an AUV during an underwater survey. Section V presents the conclusions of this work.

## II. LANDING HARDWARE REQUIREMENTS

### A. Vehicle hardware

The hardware requirements for vehicles to perform landing operations are sufficiently different to standard vehicles to warrant specific consideration. In this work, we propose a vehicle concept with negative buoyancy during operation. This minimises the use of vertical thrusters during landing operations, allowing the vehicle to remain stationary and vibration free whilst landed and saves power. The negative buoyancy also means that the vehicle can land passively, without the use of its thrusters during its final stage of descent in order to minimize agitation of loose sediments.

The features of the vehicle can be seen in the Fig. 2. Independent heave, surge, sway and heading control are needed to allow the vehicle to operate at low speeds manoeuvres and hover when necessary. Two horizontal thrusters provide surge and heading control. Two thrusters oriented away from the centre of the vehicle and inclined at 22.5° with the vertical, control sway and heave. The inclined thrusters direct thrust away from the area directly below the vehicle to minimizing the disturbance of sand and sediments during landing. A nylon landing



96 skid distributes the vehicle's weight and provides a stable footing when the vehicle has landed.  
 97 This also provides enough clearance to protect the sensors on the the vehicle.

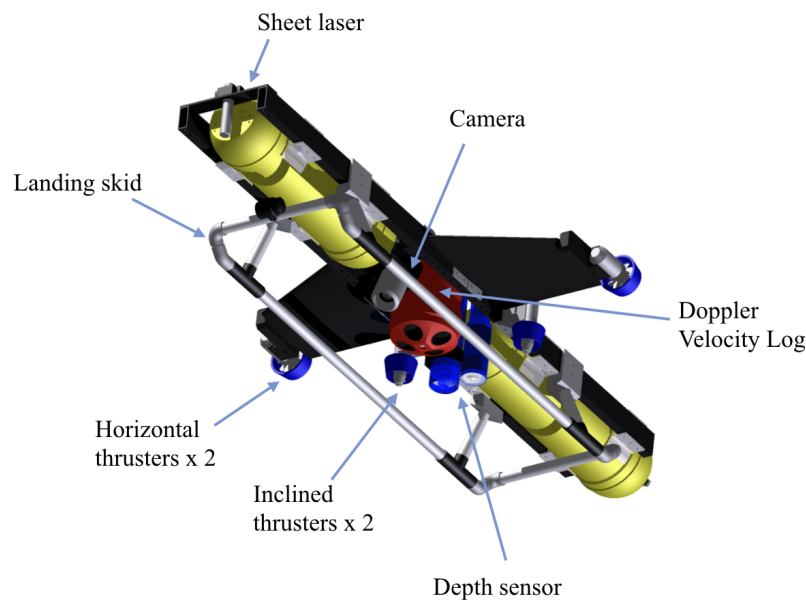


Fig. 2. Concept design of a landing vehicle

98 The vehicle is designed to be negatively buoyant to allow for landing. Although variable  
 99 buoyancy engines are available [30], these typically have a capacity of 1 L, and the limited  
 100 change in buoyancy imposes limits on the conditions under which a vehicle can remain securely  
 101 landed. Here, a hydrodynamic solution is chosen using a fixed wing NACA651412 profile to  
 102 offset the negative buoyancy during forward motion as this approach can compensate for a large  
 103 change in buoyancy. This profile produces lift at zero angle of attack and so minimising the  
 104 vehicle's drag. During slow manouvres the vehicle is still able to hover, and methods such as  
 105 the two drop weight method can be used for diving and fail-safe surfacing [31]. The vehicle has  
 106 a standard navigation suite, consisting of a 1.2 MHz Doppler Velocity Log (DVL), raised more  
 107 than 30 cm of the bottom of the vehicle to provide bottom lock even when landed, a compass  
 108 based Attitude Heading Reference System (AHRS), and pressure depth sensor. In addition to  
 109 standard localisation, the pressure sensor and DVL range can also be used to measure vertical  
 110 motion during landing and confirm the vehicle remains stationary on the seafloor once landed.

111 For short distances of travel, the relative motion of an underwater vehicle can be estimated  
 112 through dead reckoning with an error accumulation of 2 to 5% of the distance travelled for  
 113 relatively modestly priced solutions, and 0.1% for high-end systems. These would provide a

relative position error of 0.5 m to 1.25 m, and 2.5 cm when revisiting a specific location within a 25 m long map segment considered in this work for landing site analysis, where further increases in accuracy can be achieved by implementing SLAM or terrain aided navigation methods. While accurate positioning is an essential, fundamental aspect of AUV behaviour, the solutions required here are not specific to landing operations and further details on the various approaches can be found in [32].

### B. High resolution mapping system

A high resolution mapping system using light sectioning is used to generating millimeter resolution bathymetry, as illustrated in Fig. 3 [21]. The system comprises of a sheet laser projecting a line on the seafloor from a mapping altitude  $a$  and a camera offset by a distance  $b$  from the laser with vertical mounting angle  $\phi_m$ . The sheet laser projects a line on the seafloor whose projection is captured by the camera with horizontal and vertical opening angles  $\phi_h$  and  $\phi_v$  respectively. By detecting the laser line in the image captured by the camera, it is possible to determine the relative coordinates of each recognised point in the laser line. These can then be used to generate continuous bathymetry measurements in the earth-fixed coordinate system based on the pose of the vehicle.

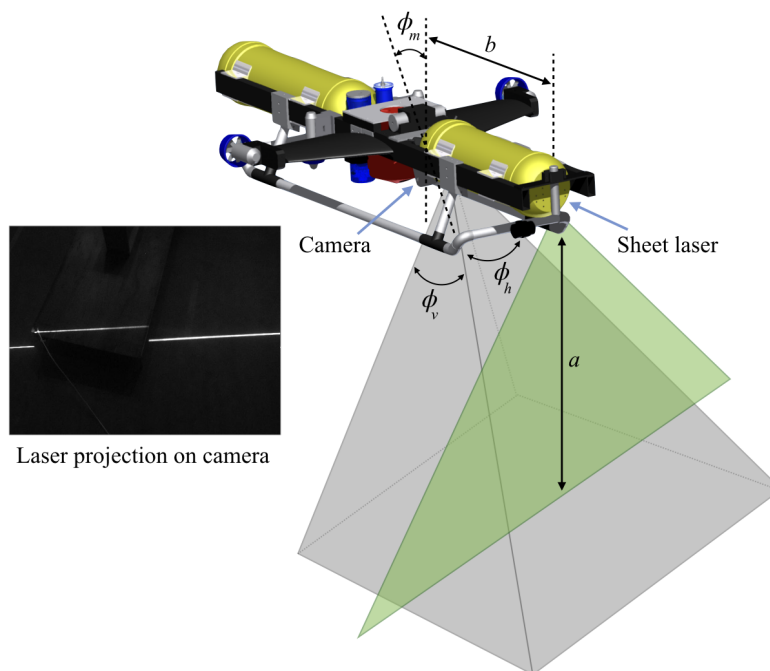


Fig. 3. Setup and mechanism of the high resolution mapping system

### III. AUTONOMOUS LANDING ALGORITHM

This section describes the conditions that need to be satisfied in order to achieve safe landing. These are illustrated using high resolution bathymetric data collected by the vehicle BOSS-A at the Takuyo Daigo seamount during the KR16-01 cruise of R/V Kairei [18], [20]. The specifications of the laser mapping system used to collect the data are given in Table I. These landing conditions are used to develop an autonomous landing algorithm that consists of the following steps:

- *Surface mapping*: Convert high resolution point clouds generated by the mapping system into a bathymetry surface with uniform lateral resolution.
- *Landing area detection*: Identify landing areas within sections of bathymetry that satisfy the criteria for safe landing developed in this work.
- *Site identification*: Within the detected landing areas, identify candidate landing sites that are large enough for a vehicle of defined geometry to fit along a certain heading.
- *Site selection*: Landing site properties are extracted for all the candidate sites and a cost function is defined to assess their suitability. The site with minimum landing cost is selected as the final landing site.

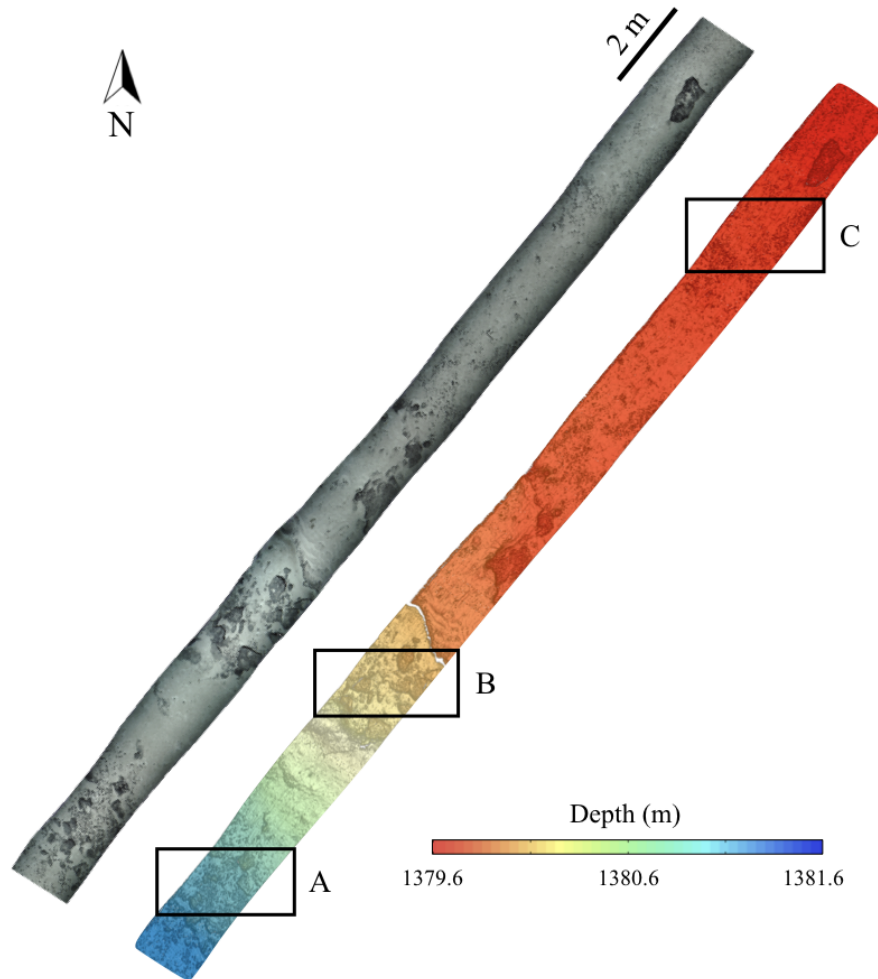
TABLE I  
PROPERTIES OF THE MAPPING SYSTEM

Property	Value
Mapping altitude $a$	2 m
Baseline between camera and laser $b$	1.03 m
Vertical mounting angle of camera $\phi_m$	20°
Horizontal opening angle of camera $\phi_h$	60.2°
Vertical opening angle of camera $\phi_v$	50.4°
Along-track resolution	4 mm
Cross-track resolution	3 mm
Vertical resolution	6 mm

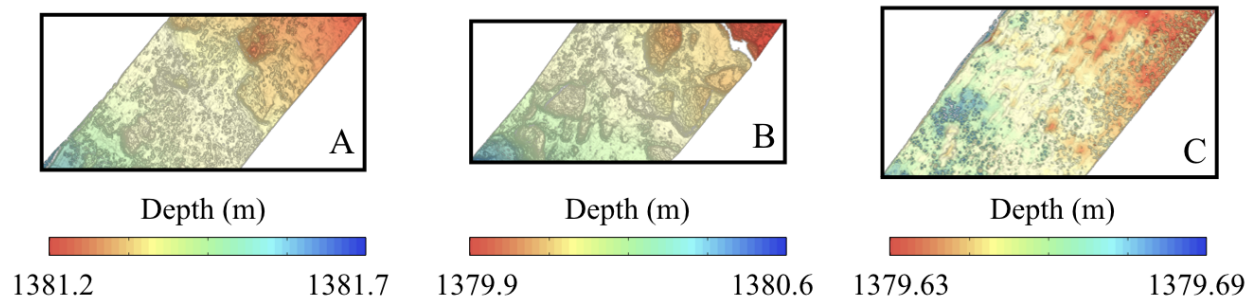
#### A. Surface mapping

Fig. 4a shows a 25 m section of laser bathymetry mapped at a heading of 230° using the method described by Bodenmann et al., [21]. Each point in the point cloud has a known position in the north, east and depth directions. The unstructured points are resampled to a uniform lateral

150 grid resolution of  $g_{res} = 10$  mm, as shown in (see Fig. 4b). This resolution is chosen as it is  
 151 sufficiently high to resolve any surface protrusions that may affect landing.



(a) Top view orthographic projection of a 3D seafloor color reconstruction (left) and its corresponding hill shaded depth map (right).



(b) Detailed views of bathymetric data resampled to a uniform lateral grid resolution of  $g_{res} = 10$  mm, corresponding to the three areas A,B and C in Fig. 4a.

Fig. 4. Seafloor bathymetry mapped using the high resolution laser mapping system mounted on a vehicle

## 152 B. Landing area detection

153 1) *Identifying landing conditions:* In order to detect safe landing areas, a number of physical  
 154 conditions need to be locally satisfied in the bathymetry. The effects of slope on landing are  
 155 analyzed considering the vehicle's righting moment, seafloor friction and currents. The effects  
 156 of protrusions on landing are analyzed considering their height. The vehicle parameters needed  
 157 to judge safe landing are illustrated in Fig. 5, with specific values used in this study shown in  
 158 Table II.

TABLE II  
PHYSICAL PROPERTIES OF UNDERWATER VEHICLE

Property	Description	Value
$l_u$	length of landing vehicle	1.7 m
$b_u$	width of landing vehicle	0.5 m
$h_u$	height of landing vehicle	0.40 m
$F_G$	force of gravity	637 N (mass 65 Kg)
$F_B$	force of buoyancy	608 N (mass 62 Kg)
$F_R$	net downward force	29 N
$d_g$	vertical distance to $C_G$	0.25 m
$d_m$	vertical distance between $C_G$ and $C_B$	0.05 m

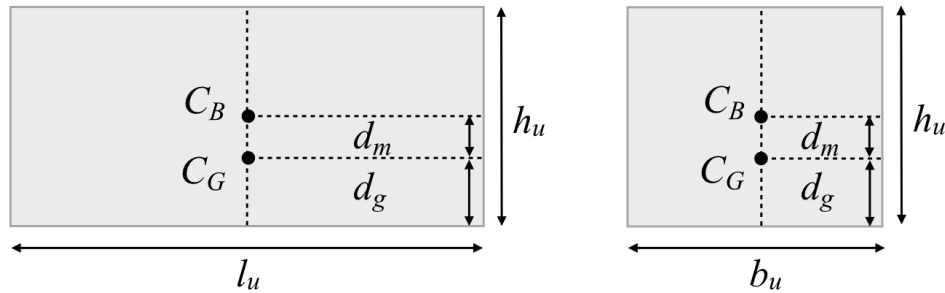


Fig. 5. Side and front view of a vehicle with parameters used to determine landing conditions

159 *Sloping surfaces:* The criteria for landing success is considered as when the vehicle can  
 160 remain stationary and in full contact with the seafloor. Here, we determine the minimum slope  
 161 on which a vehicle of known geometry and righting moment can meet this condition. The analysis  
 162 is performed along different orientations of the vehicle with respect to the slope  $\psi$ , to find the  
 163 maximum slope  $\theta_c$  where successful landing is possible (see Fig. 6).

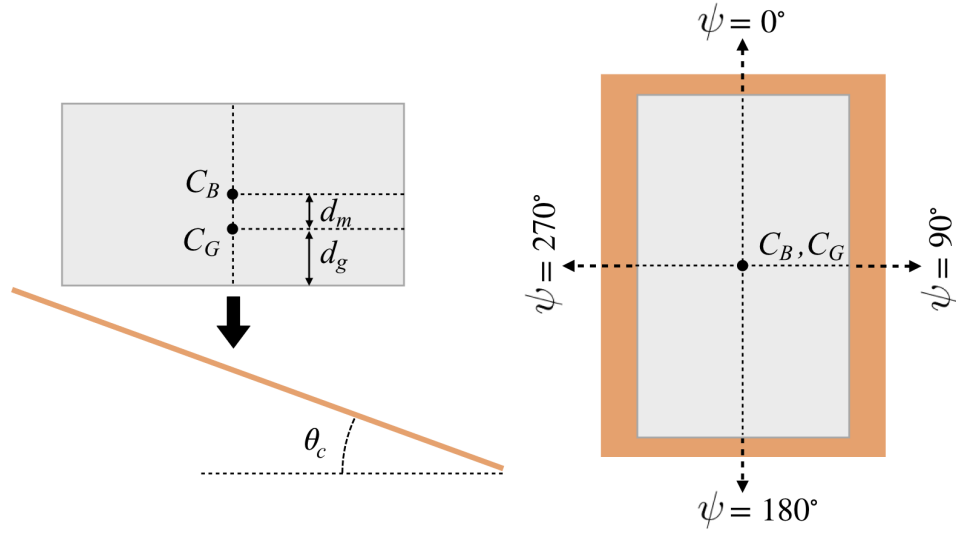


Fig. 6. Side and top view of a vehicle landing on a slope

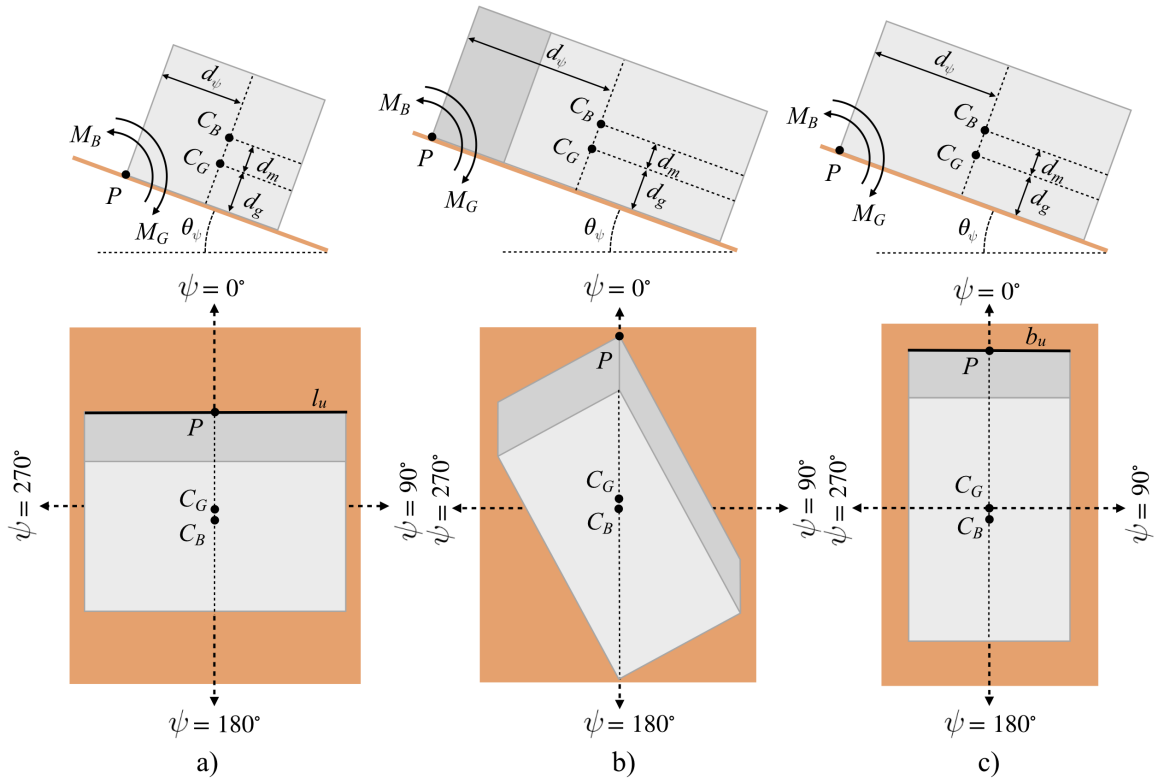


Fig. 7. a) Landing on the slope along the longer edge  $l_u$  b) Landing on the slope with the diagonal axis across the slope c) Landing on the slope along the smaller edge  $b_u$

While landing, the vehicle first makes contact with the slope along its smaller edge  $b_u$  for orientations  $0^\circ$  and  $180^\circ$  and longer edge  $l_u$  for orientations  $90^\circ$  and  $270^\circ$ . For all other orientations, the vehicle makes contact on one of its corners. To land successfully, the vehicle should settle flush with the slope of the seafloor. Once part of the vehicle makes contact with a sloped seafloor, the vehicle rotates along the plane formed by  $C_G$ ,  $C_B$  and point  $P$  as shown in Fig. 7,  $P$  being the point where the vehicle makes contact with the seafloor. Since the maximum tilt angle of the vehicle is determined by its righting moment, the maximum angle of rotation is given by the equation

$$\theta_\psi = \tan^{-1} \left[ \frac{(d_\psi \times F_R)}{(d_m \times F_B) - (d_g \times F_R)} \right], \quad (1)$$

where the distance  $d_\psi$  between the point of contact  $P$  and the centre line formed by  $C_G$  and  $C_B$  is determined by the orientation of the vehicle  $\psi$ , with respect to the slope. When landing along an edge, or along a diagonal axis across the slope, as seen in Fig. 7, the vehicle can make full contact with the surface while tilting along a single axis. For all other orientations, the vehicle rotates along a single axis until one of its edges makes contact with the slope after which it rotates about that edge to make full contact with the surface.

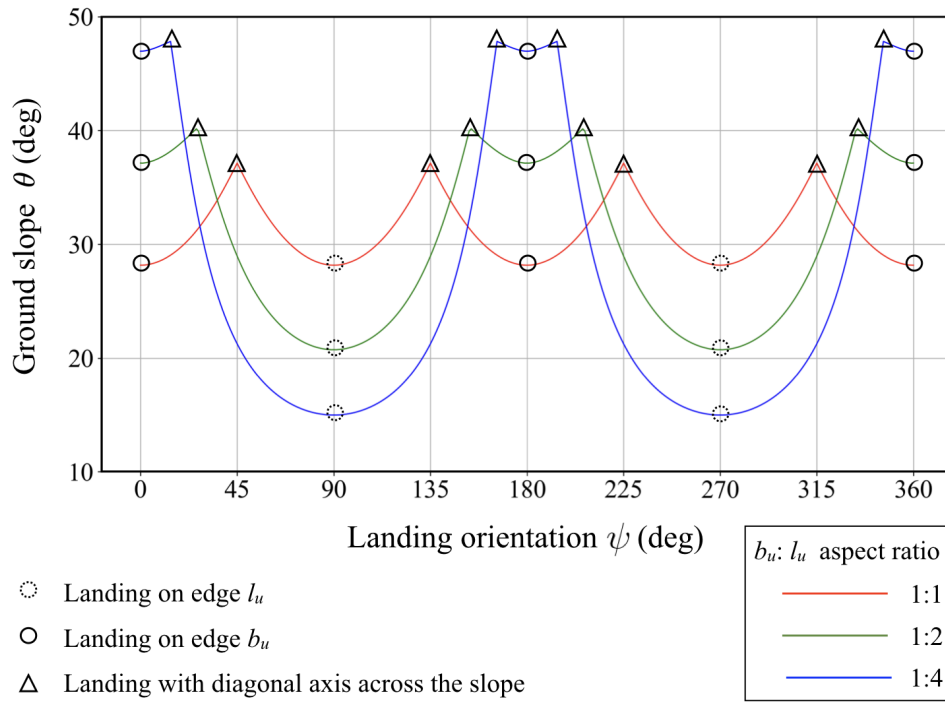


Fig. 8. Maximum landing slope  $\theta_c$  calculated for different landing orientations and aspect ratios of the vehicle



Fig. 8 shows the maximum slope  $\theta_c$  where the landing criteria can be met along relative vehicle orientations  $\psi$  from  $0^\circ$  to  $360^\circ$ . Simulations are performed for vehicle aspect ratios ( $l_u/b_u$ ) of 1, 2 and 4 respectively, where the height of the vehicle, distances  $d_m$  and  $d_g$  and net downwards force  $F_R$  remain unchanged. Since the minimum value of  $\theta_c$  occurs when then vehicle lands on its longer edge  $l_u$ , the maximum slope  $\theta_c$  on which the vehicle can land can be determined by letting  $d_\psi = 0.5 \times b_u$  in Equation 1, giving

$$\theta_c = \tan^{-1} \left[ \frac{(0.5 \times b_u \times F_R)}{(d_m \times F_B) - (d_g \times F_R)} \right]. \quad (2)$$

For the vehicle parameters in Table II, this gives a maximum slope on which the vehicle can land at any orientation of  $\theta_c = 17.7^\circ$ .

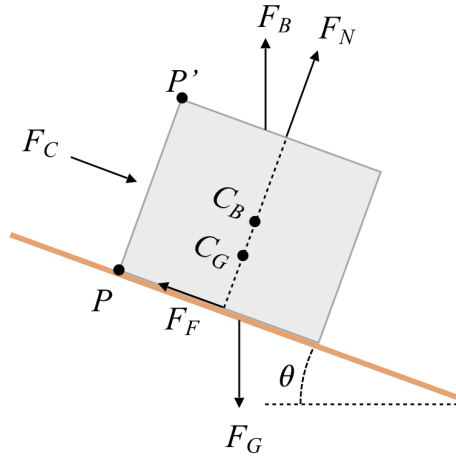


Fig. 9. Forces acting on the vehicle after landing on the sloping surface

For the vehicle to remain stationary on a slope once it has landed, the frictional force  $F_F$  needs to be greater than or equal to the sum of the component of gravity  $F_G$  acting along the slope in the downwards direction and the force due to seafloor currents  $F_C$  pushing the vehicle down the slope, corresponding to  $0 < \theta < \pi/2$  (when  $\sin \theta$  is positive). Fig. 9 illustrates this worst case scenario, where for the purpose of this simulation we neglect the effects of hydrodynamic lift. The steepness of the slope that the vehicle can remain stationary on depends on the frictional coefficient  $\mu$  between the vehicle and seafloor. The relationship between the velocity of seafloor currents  $v$  and the maximum slope on which the vehicle does not slip with its longer edge  $l_u$  across the slope is calculated for a drag coefficient for rectangular shapes  $C_d = 1.2$  [33][34] and

197 density  $\rho = 1025 \text{ Kg/m}^3$ . The balancing forces for the vehicle to remain stationary after landing  
 198 can be represented as,

$$199 \quad F_C - F_F + (F_G - F_B) \times \sin \theta \leq 0 \quad (3)$$

200

$$201 \quad F_F = \mu \times (F_G - F_B) \times \cos \theta \quad (4)$$

202 Giving the force acting on the vehicle due to seafloor currents as,

$$203 \quad F_C \leq (F_G - F_B) \times (\mu \cos \theta - \sin \theta) \quad (5)$$

204 The admissible velocity  $v_c$  of seafloor currents can be then calculated as:

$$205 \quad v_c \leq \sqrt{\frac{2 \times F_C}{C_d \times \rho \times l_u \times h_u}} \quad (6)$$

206 The frictional coefficient of the seafloor typically varies between 0.1 and 0.6 [35] with seafloor  
 207 currents in the deep ocean typically  $< 0.2 \text{ m/s}$  [36]. The effects of friction and current on  
 208 landing are calculated for slopes between  $0^\circ$  and  $35^\circ$ . The slope values calculated can be seen  
 209 in Fig. 10 where the area under each curve represents the conditions where the vehicle remains  
 210 stationary.

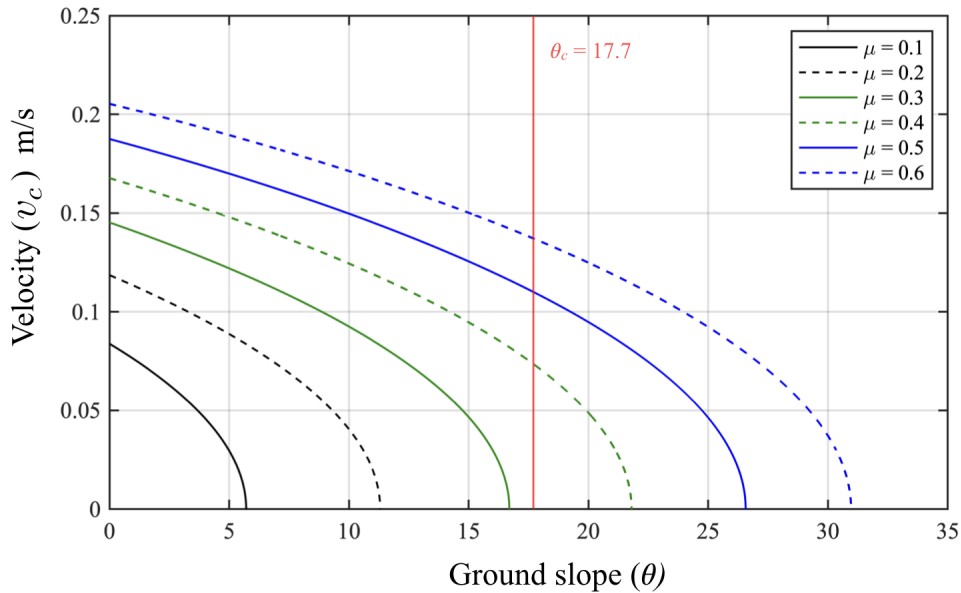


Fig. 10. Analysis of seafloor currents and friction on the ground slope

211 For a slope of  $\theta_c = 17.7^\circ$  determined for the vehicle in Table II, it can be seen that the vehicle  
 212 will remain stable for currents under  $0.14 \text{ m/s}$  and coefficient of friction more than  $0.32$ , which  
 213 is reasonable for most deep-sea applications.

214 *Seafloor protrusions:* Seafloor roughness can have an impact on landing. In particular, pro-  
 215 trusions can cause a vehicle to remain partially suspended due to its righting moment. Here we  
 216 consider the criteria for safe landing to be where the vehicle can settle on the seafloor without  
 217 being limited by its righting moment.

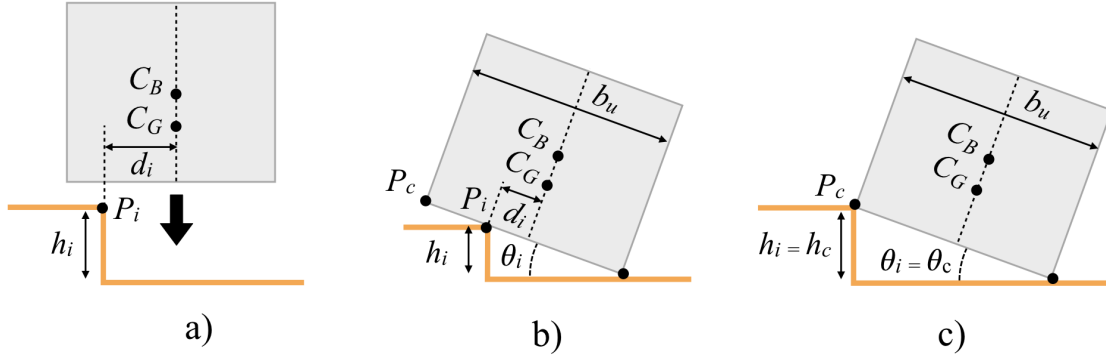


Fig. 11. a) Vehicle landing on an protrusion on the seafloor b) Landing on a protrusion between the edge of the vehicle and the  $C_G$ - $C_B$  centre line c) Extreme condition of landing on a protrusion at the edge of the vehicle

218 Fig. 11a shows a situation where the vehicle lands on the edge of a protrusion  $P_i$  at a distance  
 219  $d_i$  from the  $C_G$ . For any protrusion  $P_i$  between  $P_c$  and the  $C_G$ - $C_B$  centre line (Fig. 11b), the  
 220 maximum angle  $\theta_i$  to which the vehicle can tilt after making contact is determined by its righting  
 221 moment, given by the equation 1 as,

$$222 \quad \theta_i = \tan^{-1} \left[ \frac{(d_i \times F_R)}{(d_m \times F_B) - (d_g \times F_R)} \right]. \quad (7)$$

223 Stability is achieved when the vehicle can make contact with the seafloor after tilting at this  
 224 angle. The maximum possible height of the protrusion  $h_i$  for safe landing after making contact  
 225 at protrusion  $P_i$  can be calculated as,

$$226 \quad h_i = (0.5 \times b_u + d_i) \times \sin \theta_i. \quad (8)$$

227 In the extreme condition (Fig. 11c), the vehicle lands along its long edge  $l_u$  on point  $P_c$  of the  
 228 protrusion. In this scenario, the condition is identical to the slope condition, with  $\theta_i = \theta_c$  forming  
 229 the limiting condition. The maximum height of the protrusion  $h_c$  in this limiting condition can  
 230 be calculated as:

$$231 \quad h_c = b_u \times \sin \theta_c \quad (9)$$

232 To achieve stable landing, any part of the vehicle should be prohibited from landing on  
 233 protrusions above the height of  $h_c$ . For smaller protrusions of height  $h_i$ , the  $C_G$  of the vehicle

should not be allowed to land within a distance  $d_i$  from the protrusion depending on the height of the protrusion.

2) *Determining landing exclusion zone*: The algorithm identifies landing area in the mapped bathymetry by generating an exclusion zone where the  $C_G$  of the vehicle cannot enter for landing along any landing heading  $\alpha$  using conditions identified in III-B1 as:

- Conditions for landing on sloping surfaces from Equations 2 and 6 to generate an exclusion zone for areas of the seafloor with slope more than  $\theta_c$ .
- Conditions for landing on protrusions from Equations 7, 8 and 9 to generate an exclusion zone around protrusions on the seafloor based on their height.

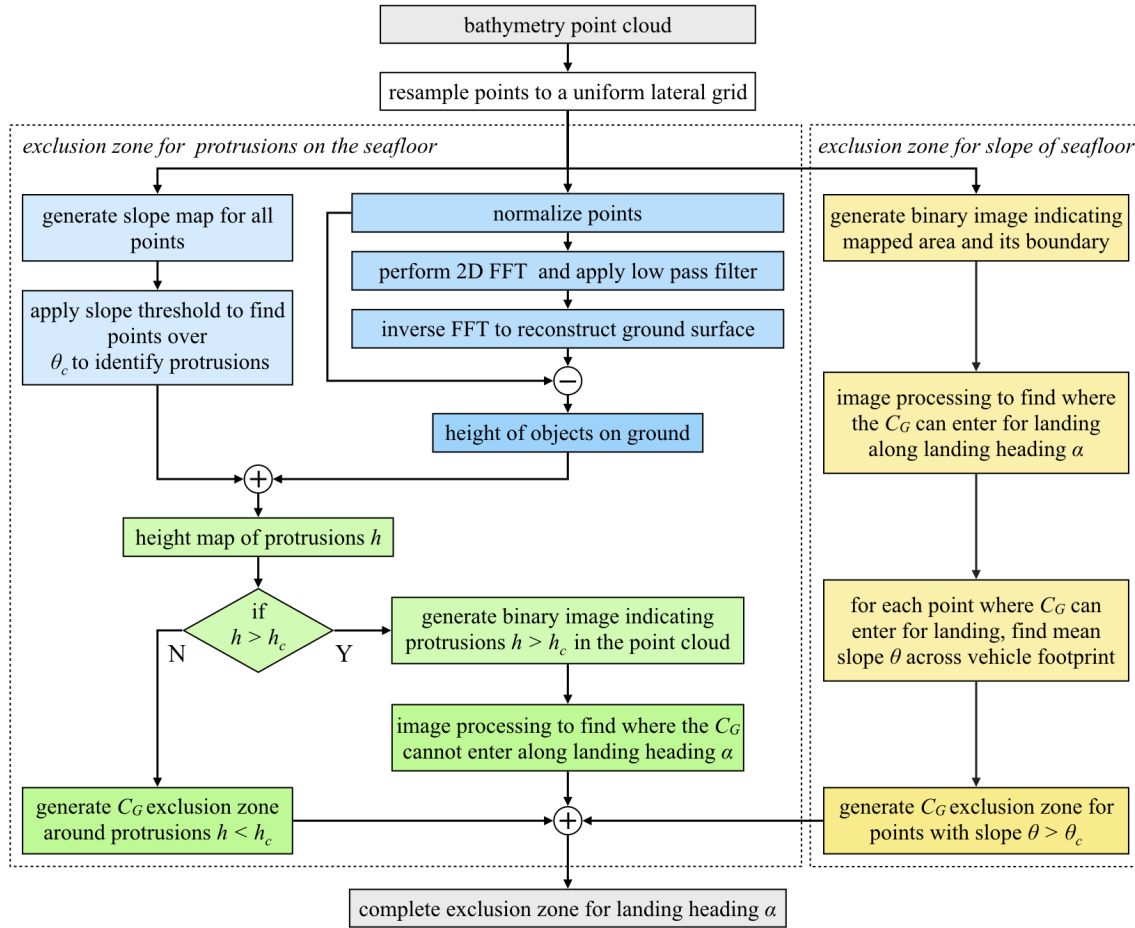


Fig. 12. Flowchart for detection exclusion zone in mapped bathymetry

The approach is summarized in Fig. 12 for the vehicle parameters given in Table II, with the various components described in the following sections. Detailed analysis has been shown for a

245 landing heading  $\alpha = 55^\circ$  as an example, which is approximately opposite to the heading of the  
 246 vehicle during mapping.

247 *Exclusion zone for slope of the seafloor:* The point cloud is analysed to generate an exclusion  
 248 zone for areas with slope more than  $\theta_c$  where the  $C_G$  of the vehicle cannot enter for landing.  
 249 For this, at first a binary image representing the mapped area and its unmapped boundary is  
 250 generated as in Fig. 13a. Binary image morphological operations are performed on this image  
 251 using a structural element with dimensions of the vehicle for a landing heading of  $\alpha = 55^\circ$  to  
 252 generate a map where the  $C_G$  of the vehicle can enter for landing as shown in Fig. 13b. The  
 253 mean slope of the ground over the footprint of the vehicle for each point where the  $C_G$  of the  
 254 vehicle can enter for landing is calculated as in Fig. 13c. A threshold of  $\theta_c = 17.7^\circ$  is applied  
 255 to identify the  $C_G$  exclusion zone for slope of the seafloor as seen in Fig. 13d.

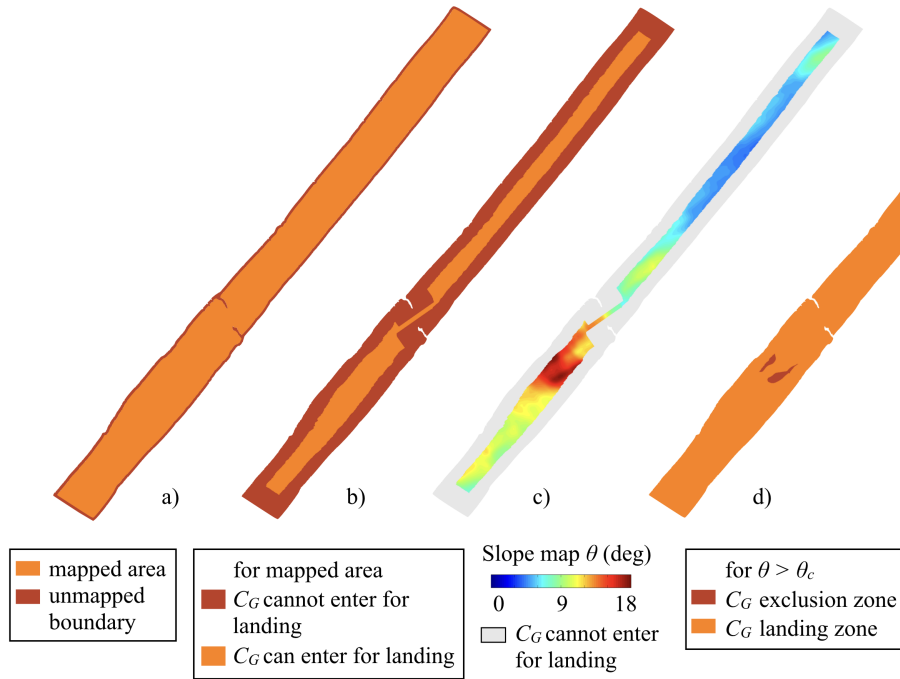


Fig. 13. a) Mapped area and its unmapped boundary b) Binary image showing area where the  $C_G$  of vehicle can enter for landing heading of  $55^\circ$  c) Slope map  $\theta$  for vehicle footprint at landing heading of  $55^\circ$  d)  $C_G$  Exclusion zone for slope  $\theta > \theta_c$

256 *Exclusion zone for protrusions on the seafloor:* An exclusion zone  $e_i$  is defined around each  
 257 protrusion  $P_i$  where the  $C_G$  of the vehicle cannot enter for landing, as shown in Fig. 14. For  
 258 the vehicle in Table II, the maximum protrusion height is determined using Equation 9 to be  
 259  $h_c = 0.15$ , where for protrusions higher than  $h_c$ , the  $C_G$  exclusion zone is set to half the length

260 of the vehicle along a given landing heading  $\alpha$ . This prevents the vehicle in this heading from  
 261 making contact with the protrusion. For protrusions with height  $h_i \leq h_c$ , the exclusion zone is  
 262 the distance  $d_i$  around the protrusion where the  $C_G$  of the vehicle is not allowed to land based  
 263 on Equations 7 and 8. The height of protrusions is calculated for distances  $d_i$  in steps of 5  
 264 mm from the  $C_G$ - $C_B$  centre line, similar to the mapping resolution, to produce Fig. 15 used for  
 265 applying the exclusion zone. Protrusions  $< 5$  mm are not considered as protrusions since they  
 266 cannot be distinguished from noise.

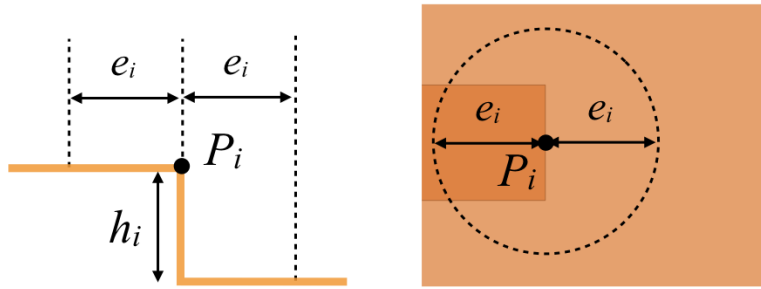


Fig. 14.  $C_G$  exclusion where the centre of gravity  $C_G$  of the vehicle is prohibited from landing

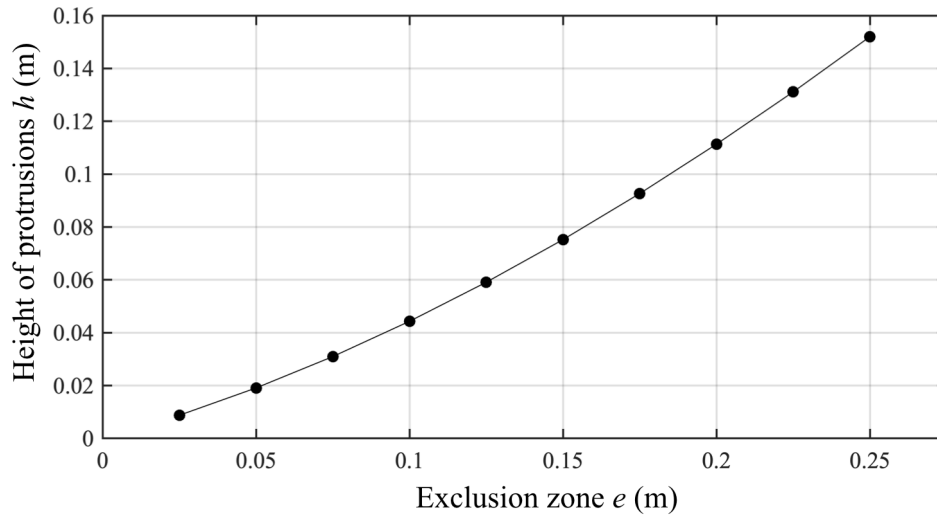


Fig. 15. Exclusion zone for height of protrusions

267 To identify protrusions in the mapped point cloud, a slope map is generated as shown in  
 268 Fig. 16a for the three areas A, B and C. Since the vehicle can land on areas with slope less the  
 269  $\theta_c$ , neighbouring points with slope less than this are not considered as protrusions. The slope

threshold  $\theta_c = 17.7^\circ$  is applied to generate the binary map Fig. 16b to identify protrusions whose height is further analysed.

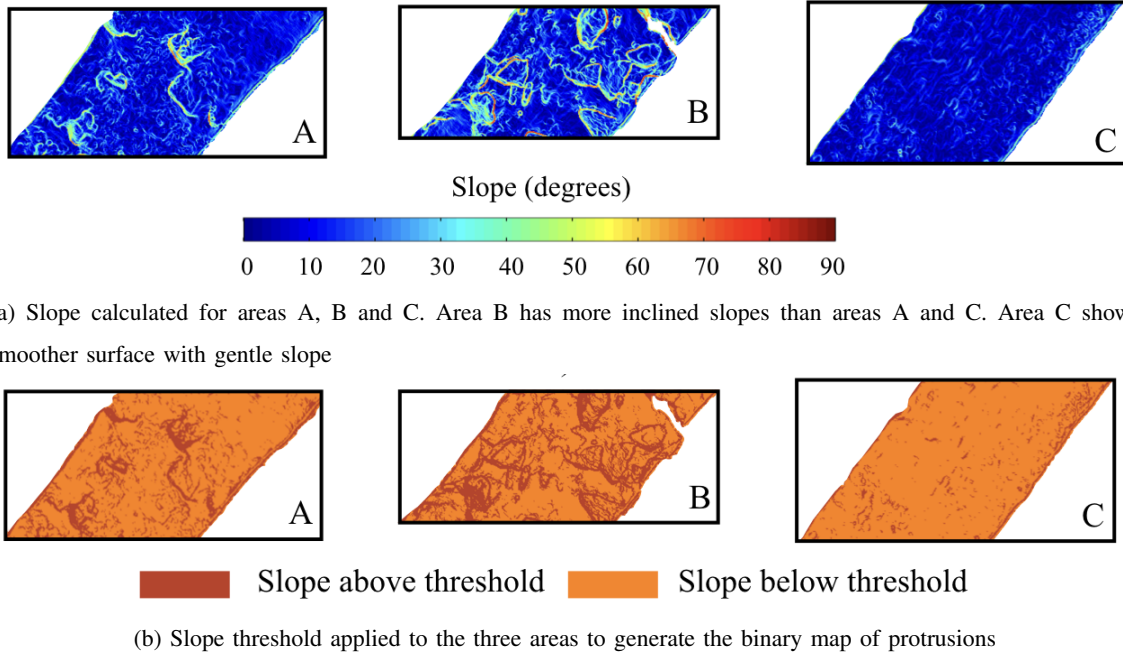


Fig. 16. Identifying protrusions in the mapped point cloud

To detect height of protrusions, two dimensional Fourier analysis is performed on the uniformly resampled bathymetry. The points to be analysed are zero filled on all sides to form an  $N \times N$  matrix, where  $N$  is the next power of 2 more than the largest dimension of the area. The depth values of the points are normalized to remove the zero frequency component by subtracting the mean depth and aligning the points with their Eigenvectors. A two dimensional  $N$  point Fast Fourier Transform (FFT) is performed on the normalized values to convert them to the frequency domain. The frequency bins are  $n \times f_s/N$ , where  $n = 1, 2, \dots, N/2$  and sampling frequency is  $f_s = 1/g_{res}$ , for grid resolution  $g_{res}$ . For a cut-off frequency  $f_c$ , filter order  $n$  and frequency bins  $f$ , a low pass filter is applied as,

$$h_l(f) = \frac{1}{\sqrt{1 + (\frac{f}{f_c})^{2n}}}. \quad (10)$$

A 3rd order filter is used to provide suitable sharpness of damping with 3 dB attenuation at the cut-off frequency. This was determined using the method described in [37] [38] as  $f_c = 2/\sigma$ , where  $\sigma$  is taken as the diagonal length of the vehicle's geometry as a multiple of the sampling resolution. The filter function is rotated around the zero frequency to form an  $N \times N$  point filter



286 and multiplied to the frequency domain values element by element. An inverse two dimensional  
 287 FFT produces a smooth surface representing the low frequency component. This is subtracted  
 288 from the unfiltered surface to generate a map of protrusions, as seen in Fig. 17.

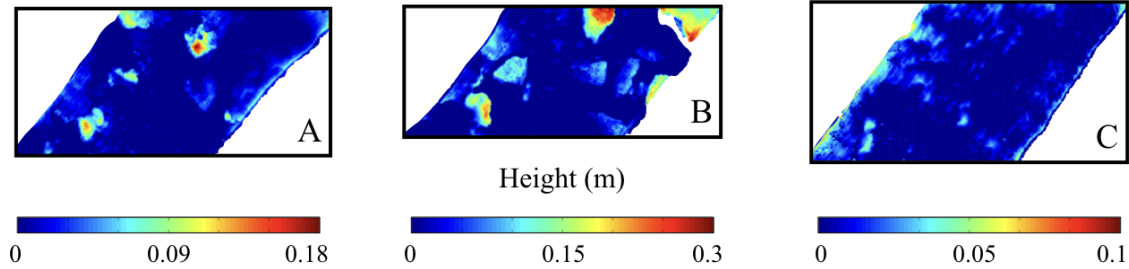


Fig. 17. Height map for areas A, B and C. Areas A and B show large objects on the seafloor compared to area C

289 The height of protrusions  $h$  is extracted in areas above the slope threshold  $\theta_c$  from the height  
 290 map of objects as in Fig. 18b. For  $h < h_c$ , the  $C_G$  exclusion zone is generated using the  
 291 relationship in Fig. 15 giving the  $C_G$  exclusion regions shown in Fig. 18c.

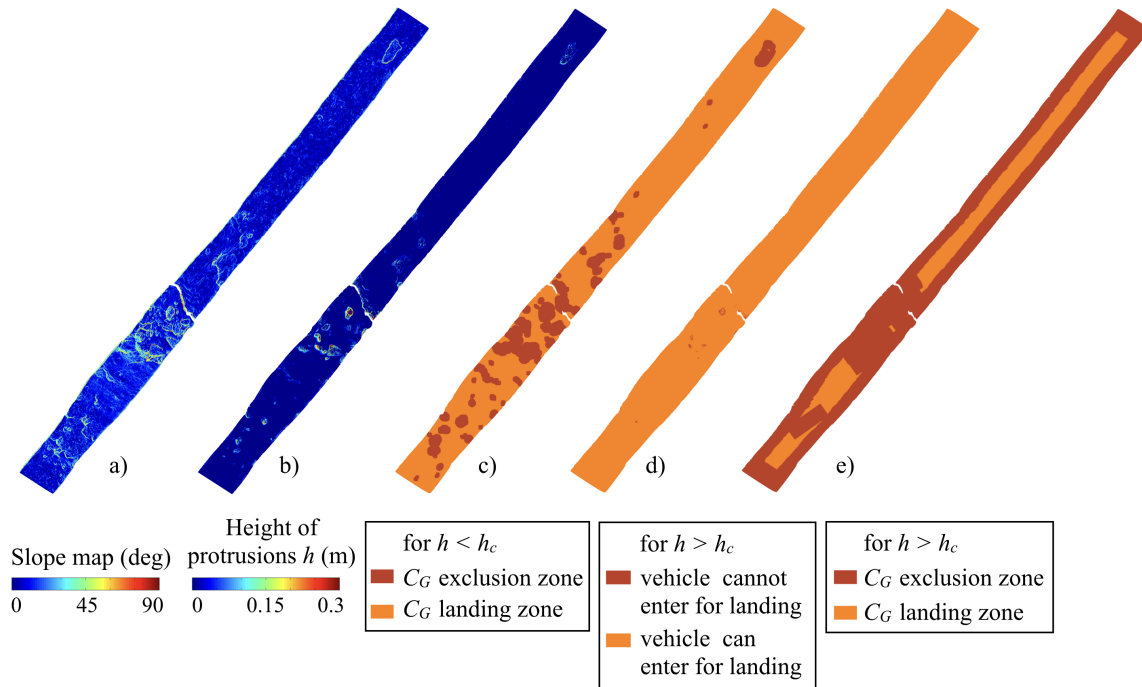


Fig. 18. a) Slope map of the mapped bathymetry b) Height of protrusions  $h$  c)  $C_G$  exclusion zone for protrusions  $h < h_c$  d) Binary image showing protrusions with height  $h > h_c$  where the vehicle cannot enter for landing e)  $C_G$  Exclusion zone for protrusions  $h > h_c$  at  $55^\circ$  landing heading

For protrusions above the threshold  $h > h_c$ , the  $C_G$  exclusion zone takes into account the landing heading  $\alpha$  and the geometry of the vehicle such that the vehicle avoids making contact with the protrusion. To achieve this, a binary image representing protrusions where the vehicle cannot enter for landing is generated as in Fig. 18d. Binary image morphological operations are performed on this image using a structural element with the dimensions of the vehicle for the landing heading  $\alpha = 55^\circ$  as shown in Fig. 18e.

### C. Site identification

The exclusion zones for slope of the seafloor in Fig. 13d and for protrusions in Fig. 18c and Fig. 18e are combined to form the full exclusion zone for that landing heading. The exclusion zone for landing heading of  $55^\circ$  is seen in Fig. 19a. Within this, landing sites are identified by verifying the eight neighbor connectivity between pixels where the vehicle  $C_G$  can land. A landing coordinate is calculated for each group of pixels where the vehicle can land furthest away from the group's boundary.

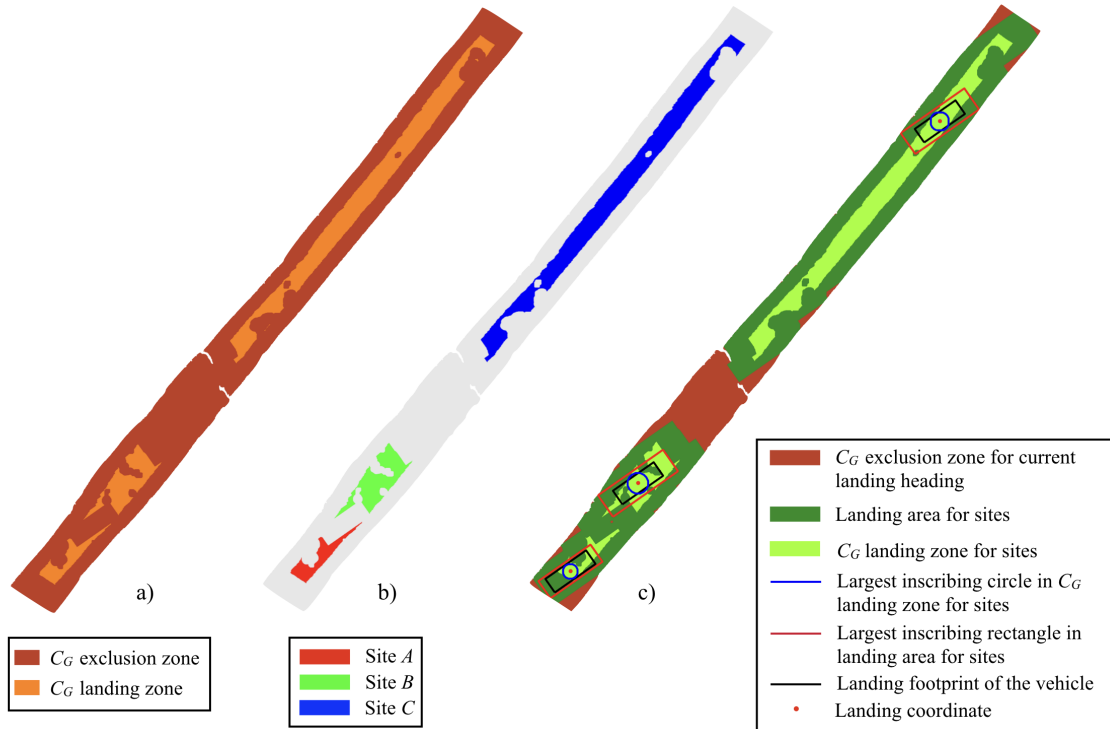


Fig. 19. a) Full  $C_G$  exclusion zone for a  $55^\circ$  landing heading, b) Landing sites identified for  $55^\circ$  landing heading, and c) Landing point calculation

305 This is achieved by determining the largest circle that can be inscribed within the group  
 306 of pixels, where the centre of this circle is the landing coordinate. Groups where the radius  
 307 of the largest inscribed circle is  $< 0.3$  m are rejected considering AUV positioning errors when  
 308 navigating back to each site. The remaining groups are labelled as candidate landing sites, where  
 309 Fig. 19 shows the candidate sites identified for a vehicle heading of  $55^\circ$ . Each candidate site  
 310 is characterised by extracting the following seafloor properties from within the footprint of the  
 311 vehicle for each landing coordinate and heading, as illustrated in Fig. 19c.

- 312 • *Slope  $P_s$* : The mean landing slope  $\bar{\theta}$  is calculated for the points in the landing footprint  
 313 of the vehicle. The slope is normalized using the threshold slope  $\theta_c$  to determine the slope  
 314 cost as follows,

$$315 \quad P_s = \frac{\bar{\theta}}{\theta_c}. \quad (11)$$

- 316 • *Safety  $P_f$* : Landing safety is determined by calculating the area of the largest rectangle  
 317 with the same aspect ratio as the vehicle that can be inscribed in the landing site along  
 318 the landing heading. The ratio of this area  $A_r$  to the footprint of the vehicle  $A_f$  is used to  
 319 calculate the safety cost. Since beyond a certain ratio, having a larger area to land does not  
 320 increase safety, the value of the cost  $P_f$  is limited to between 0 and 1 as follows,

$$321 \quad P_f = \frac{1}{4} \left( 5 - \frac{A_r}{A_f} \right) \text{ where } 0 < P_f < 1. \quad (12)$$

- 322 • *Roughness  $P_r$* : The mean deviation of height of protrusions  $h$  (Fig. 18b) from its mean  
 323 value  $\bar{h}$  is used to determine the roughness  $R$  for the landing footprint. This is normalized  
 324 by the maximum value of roughness,  $R_{max}$ , which is calculated for the terrain with max-  
 325 imum possible height of protrusions in the landing footprint. To determine this terrain, the  
 326 maximum height of protrusions around the centre of footprint are determined as in Fig. 20a  
 327 which will result in the centre being out of the  $C_G$  exclusion zone for landing as in Fig. 20b.

$$328 \quad R = \frac{1}{n} \sum_{i=1}^n |h_i - \bar{h}| \quad (13)$$

$$329 \quad P_r = \frac{R}{R_{max}} \quad (14)$$

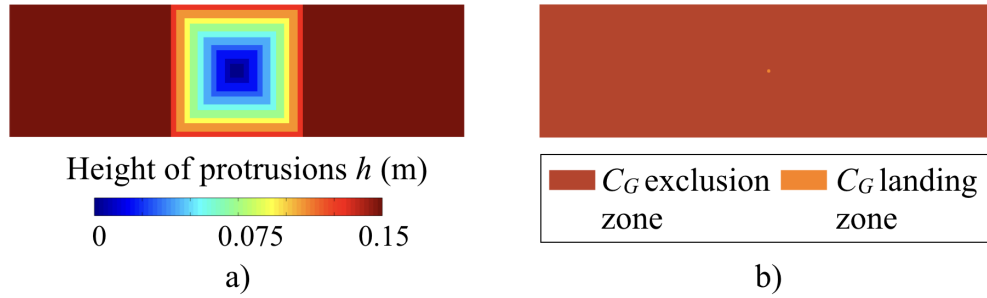


Fig. 20. a) Terrain in the landing footprint with maximum possible height of protrusions resulting in the largest roughness,  $R_{max}$  b)  $C_G$  exclusion zone calculated for this terrain with centre of the footprint out of the exclusion zone for landing

#### 331 D. Site selection

332 In order to select the most suitable landing site, the following cost function is defined based  
 333 on the features of each candidate site,

$$334 \quad C_s = \frac{1}{3} [P_s + P_f + P_r]. \quad (15)$$

335 The candidate site with the lowest cost is selected as the most suitable landing site. The landing  
 336 costs calculated for the three sites in Fig.19c are shown in Table III, where in this case site C  
 337 is the most suitable of the three candidates.

TABLE III  
LANDING SITE PROPERTIES

Site	$P_s$	$P_f$	$P_r$	$C_s$
A	0.54	0.85	0.29	0.56
B	0.67	0.62	0.32	0.54
C	0.34	0.65	0.15	0.38

338 For each section of mapped bathymetry, the most suitable landing site is selected from  
 339 candidate sites determined for multiple landing headings. These are calculated between  $-90^\circ$   
 340 and  $90^\circ$ , where the costs for landing sites in the opposite direction are identical for a rectangular  
 341 vehicle geometry. Costs are calculated in steps of  $5^\circ$  to achieve a balance between resolution  
 342 and computational cost.

343 The landing costs calculated for all the landing candidates in the bathymetry shown in Fig. 4  
 344 are plotted in Fig. 21. The analysis shows that landing site C for a landing heading of  $\alpha = 40^\circ$

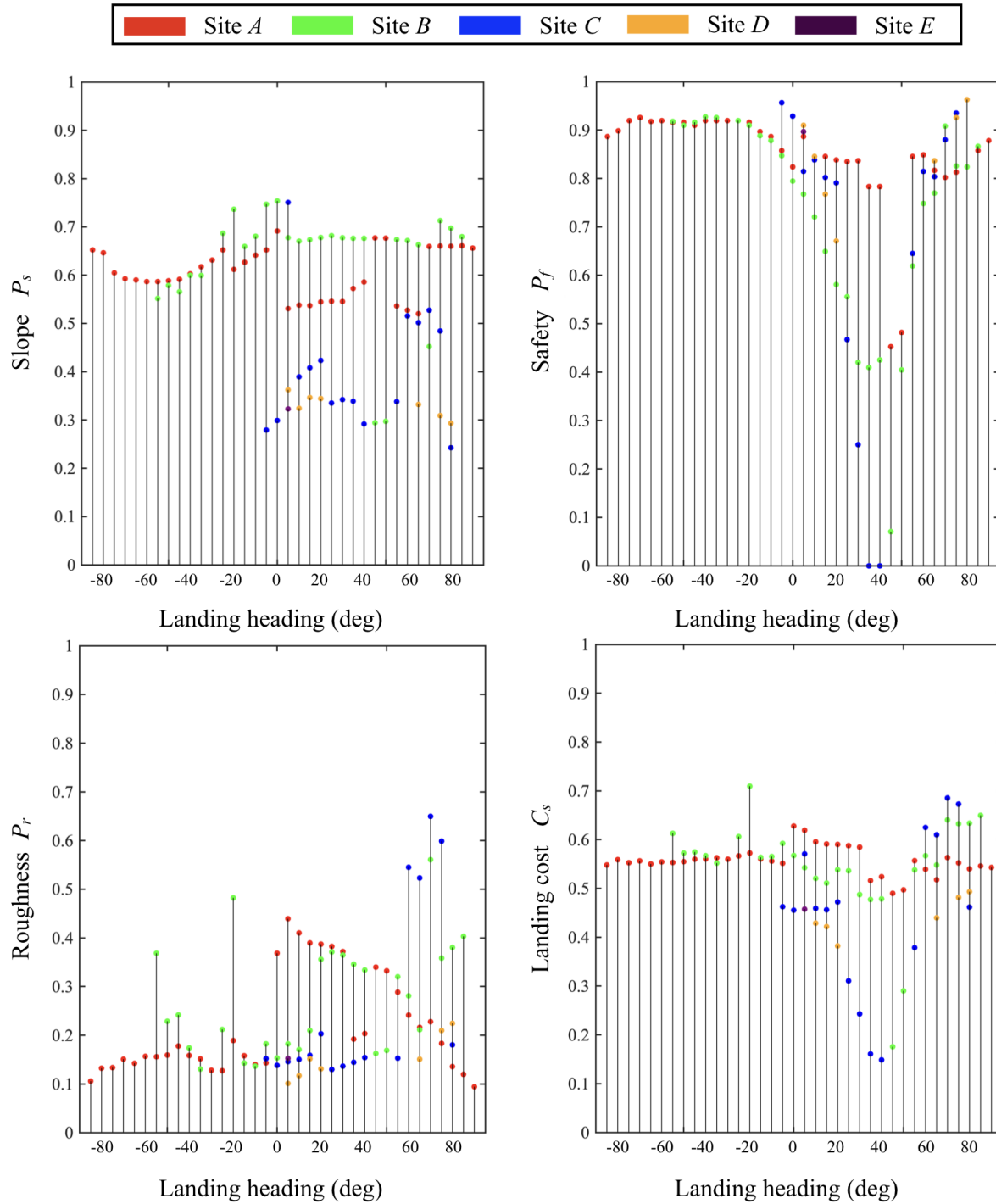


Fig. 21. Properties of landing sites between landing headings  $-90^\circ$  and  $90^\circ$  in steps of  $5^\circ$

has the lowest cost and is selected as the final landing site in this region of seafloor, with the properties of the site shown in Table IV.

TABLE IV  
PROPERTIES FOR FINAL LANDING SITE

Property	Value
Landing heading	40°
Mean depth	1379.7 m
$P_s$	0.29
$P_f$	0.00
$P_r$	0.16
$C_s$	0.15

#### IV. IMPLEMENTATION ON SEAFLOOR DATA

The landing algorithm is applied to a 500 m section of high resolution seafloor laser bathymetry data collected on the southern slopes of the Takuyo Daigo seamount between depths of 1379 and 1429 m. The data was obtained by the AUV BOSS-A using the same mapping system described in Section III. In order to perform the analysis, the data is split into twenty 25 m long submaps as illustrated in Fig. 22.

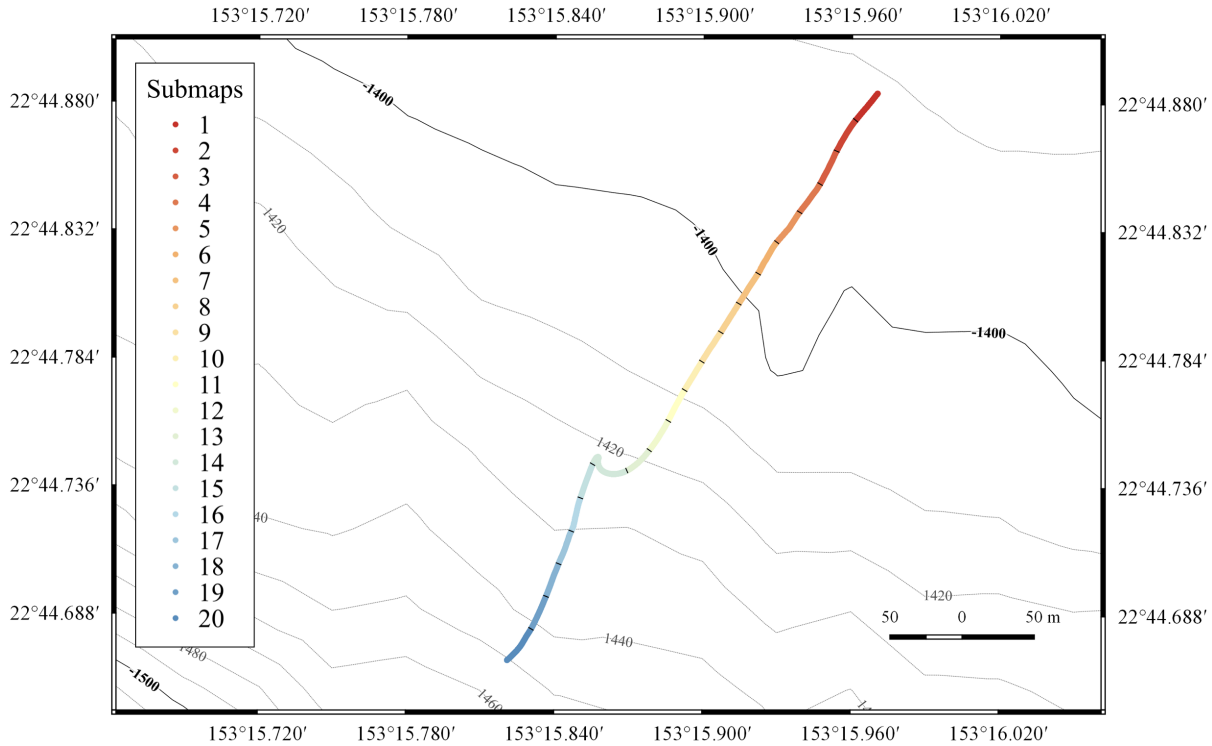


Fig. 22. Vehicle trajectory at Takuyo Daigo seamount where the algorithm is implemented on twenty 25 m long submaps

353 The northern part of the transect (submaps 1 to 9) is gently sloped with relatively flat, exposed  
 354 manganese crusts. The southern parts of the transect (submaps 10 to 20) is more steeply sloped  
 355 with a higher degree of roughness [18], [21]. The algorithm was applied to determine the most  
 356 appropriate landing coordinate and heading within each submap for the vehicle parameters given  
 357 in Table II. The algorithm found at least one landing sites in each submap with a total of 754  
 358 landing sites, the landing cost details of which are shown in Table V.

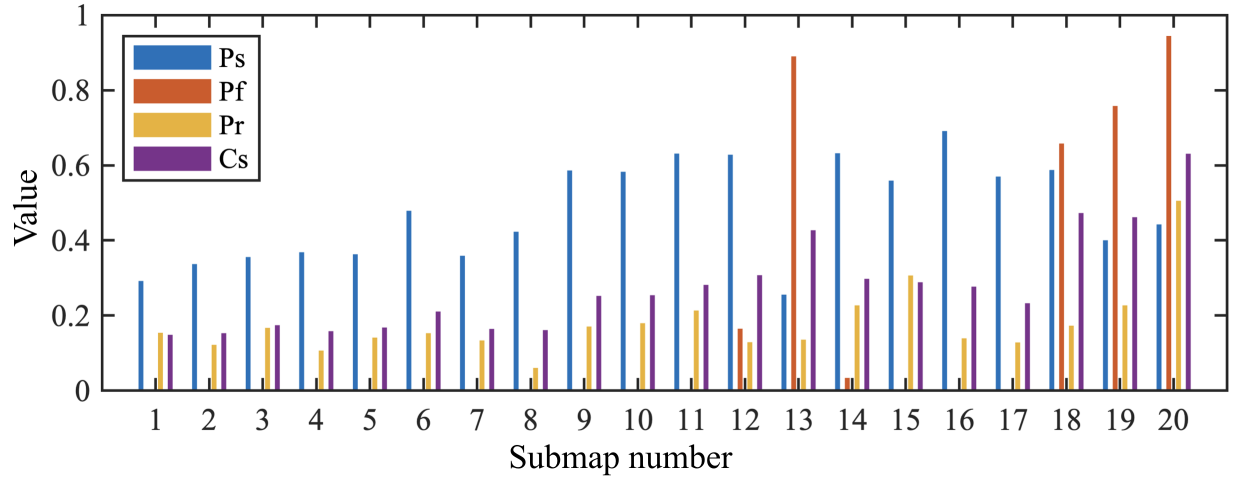
TABLE V  
RESULTS OF ANALYSIS ON MAPPED BATHYMETRY

Property	Value
Transect length	521 m
Area surveyed	794.4 sq.m.
Number of sites	754
Mean landing cost	0.28
Landing cost range	0.15 – 0.63
Landing cost Std. Dev.	0.13

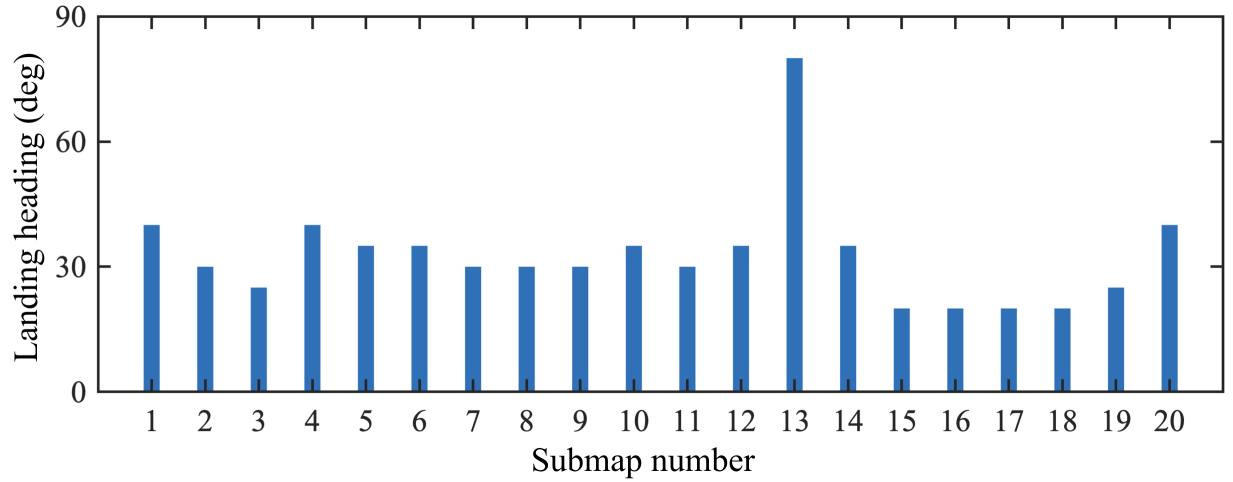
359 The properties of the final landing site with the least landing cost in each submap can be seen  
 360 in Fig. 23a. The landing heading for each submap is close to the mapping heading due to the  
 361 narrow swath of the bathymetry as shown by Fig. 23b. The narrow swath of the bathymetry is  
 362 clearly visible for the submaps shown in Fig. 24. The sites towards the south of the transect  
 363 (submaps 10 to 20) have higher landing costs than the sites towards the north of the transect  
 364 (submaps 1 to 9) with larger values of  $P_s$  and  $P_r$  due to the increased steepness and roughness  
 365 of the seafloor. Fig. 24a shows submap 1, which has a low landing cost due to the wide swath  
 366 and gently sloping smooth surface. Submap 13 (Fig. 24b) shows a narrow landing area due to  
 367 reduced mapping swath while turning on a smooth, gently sloping area of the seafloor. Submap  
 368 20 (Fig. 24c) towards the lower end of the transect has a narrow landing area on an area with  
 369 high roughness and a steep slope, resulting in a high landing cost. Fig. 25 shows the locations  
 370 of each landing site identified in the submaps with their  $C_s$  values.

371 The algorithm was implemented in MATLAB R2018 and executed on a computer, the specifi-  
 372 cations of which are given in Table VI. Parallel processing was not used for this implementation  
 373 and the algorithm was run using a single CPU thread. Each 25 m submap takes the vehicle  
 374 about 250 seconds to record raw images and position data at a forward velocity of 0.1 m/s. The





(a) Properties and landing costs determined for the sites selected within each submap



(b) Landing headings for the sites selected within each submap

Fig. 23. Characteristics of the landing sites determined by the algorithm for the submaps analysed in this work.

375 program to convert the raw data into a bathymetry point cloud is implemented in C++ [21] and  
 376 takes an average of 33.5 sec to execute for a submap.

377 Each submap is split into an average of 14 blocks of 2.5 m mapping transect length to generate  
 378 slope maps and height of protrusion maps for the blocks. The individual maps generated for all  
 379 the blocks are then combined to make full maps for the submap. An exclusion zone is calculated  
 380 using these full maps for landing headings between  $-90^\circ$  and  $90^\circ$  in steps of  $5^\circ$  followed by  
 381 landing site identification and final landing site selection. The average computation time for  
 382 different steps of the algorithm for all submaps is also seen in Table VI. The processes for  
 383 generation of individual maps for blocks and identification of landing sites for different landing

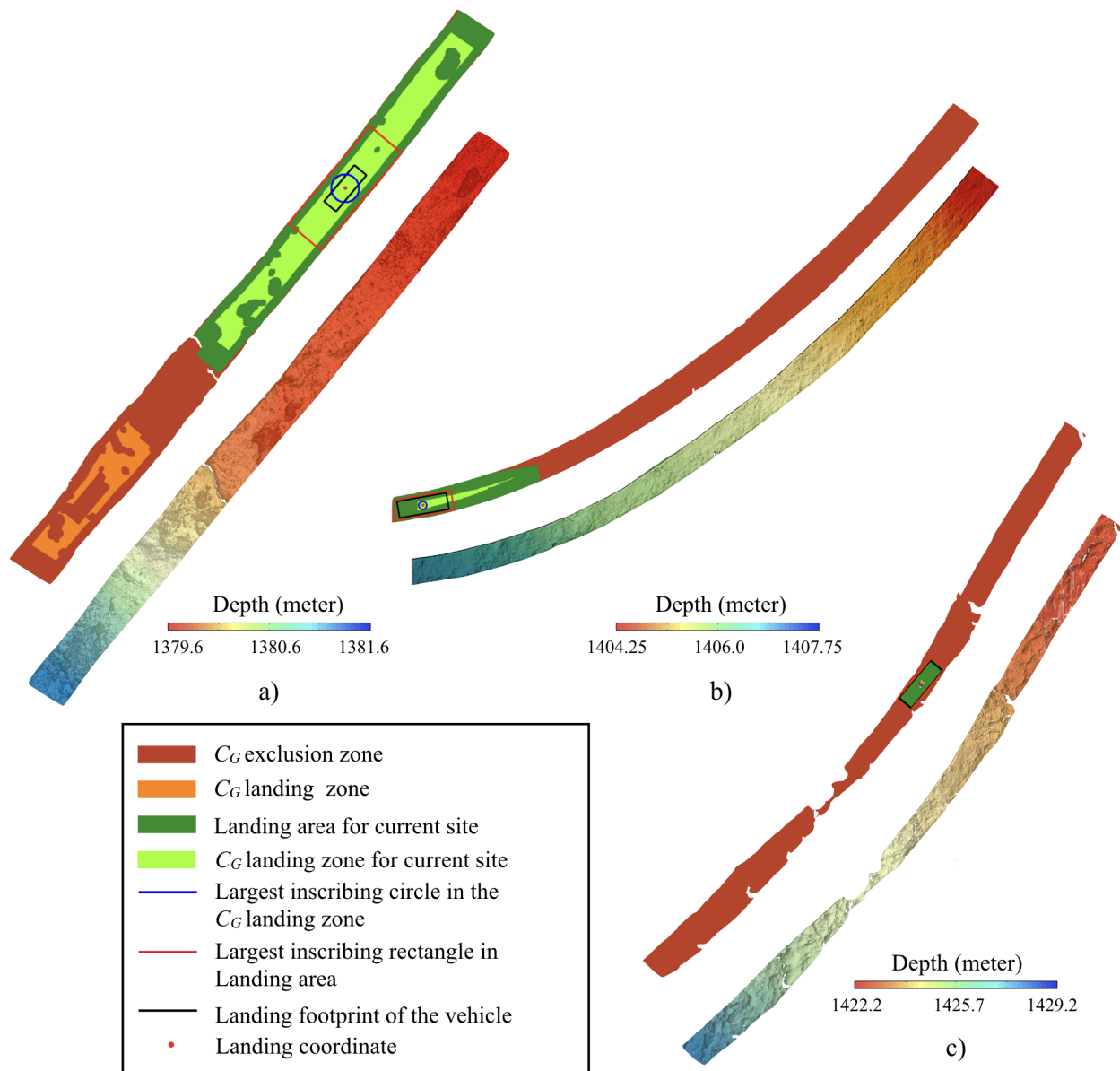


Fig. 24. Landing coordinate detected for a) Submap 1 at start of the transect on a gently sloping smooth surface b) Submap 13 towards the middle of the transect with narrow mapping swath c) Submap 20 at the end of the transect on a steep sloping rough surface

384 headings can be executed in parallel since they are independent and not memory intensive. This  
 385 can reduce the overall computation time of the algorithm for execution during seafloor surveys.

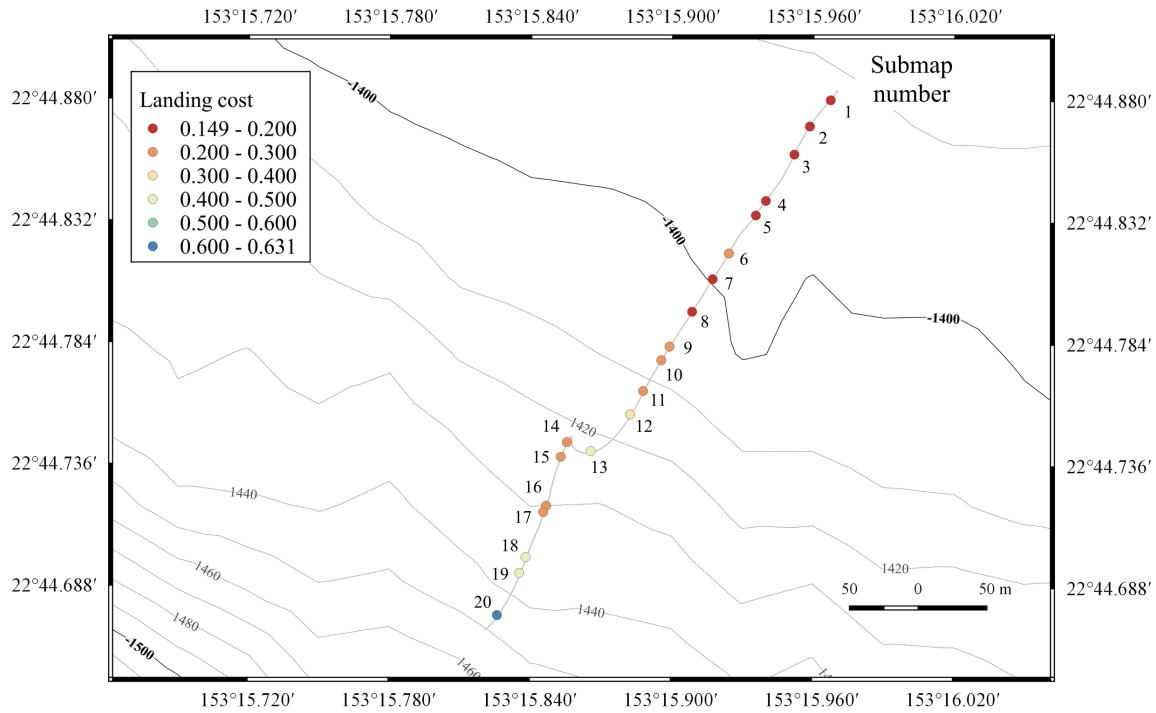


Fig. 25. Landing costs for identified final landing sites for submaps in the transect

## V. CONCLUSIONS

- This paper has described vehicle hardware and seafloor mapping system considerations specific to autonomous landing, together with the concept of a negatively buoyant, fast transiting landing vehicle using wing sections.
- A novel, fully automated method to identify safe landing sites for an AUV based on mm-resolution seafloor bathymetry has been developed and demonstrated based on data collected during field operations.
- The conditions for safe landing have been derived, taking into consideration the seafloor slope, protrusions and vehicle geometry.
- An algorithm to identify and rank potential landing sites (location and heading) has been developed and its effectiveness demonstrated on a 500 m transect of seafloor data acquired using high resolution laser mapping system.
- The described method can allow autonomous platforms to land safely, where the computations required for the algorithm can be completed in dive relevant time frames, as demonstrated using a standard mid-range CPU.

TABLE VI  
COMPUTER SPECIFICATIONS AND COMPUTATION TIMES

Specification	Value
Processor	Intel Core i7
Processor speed	2.2 GHz
RAM	16 GB
Graphics card	Intel Iris Pro 1536 MB
Number of threads	1
Software	MATLAB R2018

Process	Subprocess	Time (sec)
<b>Processing raw points</b>		
	Resampling to lateral uniform grid	<b>14.1</b>
<b>Processing one block</b>		
	Generate slope map	0.2
	Normalize points	0.1
	FFT and smoothing	1.7
	Generate protrusion height map	0.2
<b>Processing for submap</b>		
	Generating maps for 14 blocks	<b>30.8</b>
	Combining blocks to full maps	<b>6.7</b>
	Generate exclusion zone for protrusions $h < h_c$	<b>4.3</b>
<b>Landing sites for one landing heading</b>		
	Generate exclusion zone for slope	1.8
	Generate exclusion zone for protrusions $h > h_c$	1.1
	Identify landing sites	0.1
	Generate landing site properties	2.5
<b>Final landing site for submap</b>		
	Landing sites for headings between $-90^\circ$ and $90^\circ$ in steps of $5^\circ$	<b>198</b>
	Final landing site selection	<b>1.0</b>
<b>Total time for submap</b>		<b>254.9</b>

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 406 Fundamental Tools for the Utilization of Marine Resources of the Japanese Ministry of Education.

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**Sangekar Mehul Naresh** (M10) received the M.Sc. degree in underwater robotics and Ph.D. degree in ocean technology from The University of Tokyo, Japan, in 2010 and 2014, respectively. He is working at present as a project researcher at the Thornton Lab, Institute of Industrial Science, The University of Tokyo. Currently, he works on techniques for intelligent, multi-layer resolution mapping of the seafloor using autonomous underwater vehicles. He is also working on developing algorithms for analysis of high resolution seafloor bathymetry and correlating it with other measured seafloor parameters.

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**Thornton Blair** (M07) received the B.Eng. degree in ship science and the Ph.D. degree in underwater robotics from The University of Southampton, Southampton, U.K., in 2002 and 2006, respectively. He is Associate Professor at the University of Southampton with a cross appointment with the Institute of Industrial Science, The University of Tokyo. His research interests focus on developing scalable methods for visual and in-situ chemical seafloor observation through improved sensing and autonomy. He is dedicated to fielding real systems in real environments and overcoming bottlenecks in the flow of information from data collection to human insight.



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**Adrian Bodenmann** (M09) received the M.Sc. degree in microengineering from the Ecole Polytechnique Fdrale de Lausanne (EPFL), Lausanne, Switzerland, in 2009. At present he is working at the University of Southampton as a senior research assistant. His research interests are the development of camera systems for high altitude seafloor mapping and algorithms for generating 3D seafloor reconstructions based on photos of the seafloor. He also works on quantifying the certainty in these reconstructions and identifying correlations between identified objects in reconstructions and data from acoustic and chemical sensors.

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**Ura Tamaki** (M91SM02F07) graduated from the Faculty of Engineering, The University of Tokyo, Japan, in 1972 and received the degree of Doctor of Engineering from the same university in 1977. He is at present the Professor emeritus of The University of Tokyo and Director, Distinguished Professor of Center for Socio-Robotic Synthesis, Kyushu Institute of Technology, and Director of Underwater Technology Center of National Maritime Research Institute. He has developed various types of Autonomous Underwater Vehicles (AUVs) and related application technologies including navigation methods, a new sensing method using a chemical sensor, precise seafloor mapping methods, a precise seabed positioning system with a resolution of a few centimeters, a new sensing system of the thickness of cobalt-rich crust, etc. Finally, he exemplified using these technologies that AUVs are practicable and valuable tools for deep-sea exploration.