Research Article

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# Multi-point forming springback compensation control of two-dimensional hull plate

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### **Abstract**

Springback is always a technical problem in sheet metal forming. In this article, the rapid springback compensation control of two-dimensional hull plate is realized by theoretical calculation and numerical simulation. For the cylindrical shell, according to the bending forming theory of medium and thick plates, the total elastic-plastic bending moment is established, and the curvature change before and after springback is deduced. The curvature correction coefficient is determined by the precise numerical simulation technology. At the same time, the validity of the method is verified by cold bending experiment. For the shell with variable curvature, it is divided into several cylindrical surfaces according to the curvature gradient of its geometric section line. The compensation curvature array is obtained by the correction compensation algorithm of springback curvature of cylindrical plate, and the algorithm is verified by numerical simulation. The results show that the method is very close to the expected results. Thus, the efficiency and precision of forming will be improved, and the foundation of digitization of sheet metal forming is established.

### **Keywords**

Springback compensation, surface reconstruction, numerical simulation, two-dimensional shell, forming theory

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## Introduction

The formation of hull shells, especially of large curved shells, has always been a most difficult problem in the field of plastic hull processing. The simple and effective shaping of ship plates with different thicknesses and curvatures to meet design requirements has always been the development direction of hull plate-forming technology. For a long time, two-dimensional parts, such as cylindrical and conical parts, have been formed by rolling or bending shown in Figure 1.1,2 When the bending resistance of the roller is relatively large, the roller will be distorted and deformed so that the processed workpiece is in the shape of a barrel. Therefore, when roller process a relatively thick material with a relatively high material strength, it is necessary to increase the support or make the shape of the roller barrel offset the effect of its deformation.<sup>3</sup> At the same time, the traditional symmetrical three-roller coiling machine has a straight edge problem at both ends of the sheet, so the sheet material is pre-bent.<sup>4</sup> Threedimensional curved parts are mostly formed by waterfired curved plates, but there are many problems with this approach, such as high labor intensity, high requirements for workers, and low production efficiency. With the development of the computer

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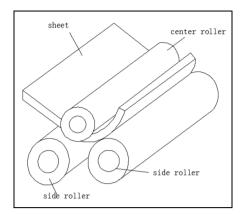


Figure I. Roll forming.

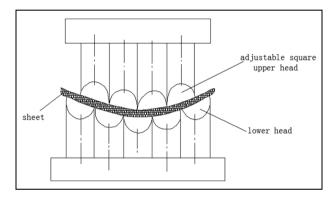


Figure 2. Asymmetric stamping square head adjustable die.

numerical control (CNC) cold bending machine, multipoint CNC cold press forming has gradually become the main process in sheet metal forming.<sup>5</sup> The asymmetric stamping square head adjustable die shown in Figure 2, developed by Wuhan University of Technology in China in 2012, can solve the indentation and wrinkling problems that easily occur in the cold forming process; at the same time, compared with the overall die, the height of each basic body punch can be arbitrarily adjusted, and the envelope surface of different shapes can be quickly formed, and the springback of the sheet forming can be compensated by correcting the shape of the envelope surface, that is, the dierepairing process of the whole die is omitted. Multipoint forming has the advantage of reducing costs, increasing production efficiency, and automating the forming process.<sup>6</sup> However, calculating springback has always been a technical problem in sheet metal forming.

Hill<sup>7</sup> first established the exact mathematical theory of plate bending by proposing the basic theory of elastoplastic bending under plane-strain conditions and deducing the complete solution without considering the hardening of materials under pure bending conditions.

Wenner<sup>8</sup> studied the work of hardening and the springback of sheet metal forming under plane strain. Chu<sup>s</sup> pointed out that the sidewall part experienced a complex bending-draw process during the forming process when he analyzed the springback problem of U-shaped parts. Based on the uniform deformation and planestrain conditions, the isotropic follow-up hardening model was adopted. The effects of in-plane restraint, materials, and process parameters on springback were analyzed. Wang et al. 10 studied the bending springback problem of V-shaped parts and established a bending moment with material strength coefficient, a hardening index, a thickness anisotropy coefficient, plate thickness, plate width, and the convex die fillet as independent variables. The model is used to predict springback, formability, stress and strain distribution, and maximum die load during sheet metal forming. Wagoner et al.11 established some mechanical models for describing and calculating springback predictions. Braga et al.<sup>12</sup> established a theoretical model of springback prediction based on an elastic pad, and its feasibility was verified through experimentations. Oliveira et al. 13 used the questions in NUMISHEET2005 as an example to study the effects of different work hardening models on the accuracy of springback prediction, indicating that the selection of material strengthening models can affect springback results. Marko et al. 14 proposed a thin-plate pure bending mechanics model that considered the thickness anisotropy coefficient r and the hardening exponent n to evaluate springback, bending, and the maximum bending moment. Chen<sup>15</sup> carried out theoretical calculation to calculate single curvature springback and the surface reconstruction is carried out by non-uniform rational basis spline (NURBS) method.

From the above research, the traditional analytical method has made numerous assumptions and simplifications in solving the springback problem, but it rarely considered the influence of process parameters on springback during the actual forming process, which brings a large error to the calculation results. In this article, based on the theory of medium-thickness plate bending, the theory of cylindrical and conical hull shell forming springback is discussed in detail. Based on this, an asymmetric stamping square head adjustable die forming simulation model is established. The accurate numerical simulation method is used to correct the springback curvature to effectively control springback and improve forming precision.

### Theoretical calculation model

The calculations of springback are based on the theorem that the distortion caused by the unloading of a force or moment is equal to the deformation caused by the loading of a force or moment of the same

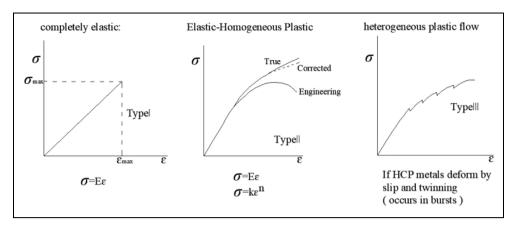


Figure 3. Stress strain curve.

magnitude but opposite direction. Therefore, the accuracy of the springback calculation depends on the accuracy of the force's or moment's calculations, as well as the accuracy of the moment of inertia calculations.<sup>16</sup>

## Basic assumption

According to the bending forming theory of medium and thick plates, the inner material elements stress and strain are determined by the width (W)-thickness (t) ratio W/t and the relative bending curvature  $\overline{k}$ . <sup>17</sup>

When  $\overline{k} < 0.1$  and W/t > 8, horizontal warpage of the sheet is ignored, and the plate is in elastoplastic deformation. The material in the inner elastic zone is subjected to unidirectional stress, the material in the outer plastic zone is in plane strain, and the neutral layer is not moved. In line with the bending engineering theory, the thickness reduction can be neglected.

Meet the following basic assumptions:

- 1. Plate bending process is pure bending;<sup>16</sup>
- 2. It only undergoes elastic deformation and elastoplastic deformation, it does not undergo plastic deformation, and it satisfies the plane-strain condition, so that the strain in the width direction  $\varepsilon_b = 0$  and the variation of the thickness can be neglected;<sup>17</sup>
- 3. There is no extrusion between the longitudinal fibers of the plate during the bending process, and there is no transverse stress between the fiber layers, and each fiber in the slat is in a uniaxially stretched or uniaxially compressed stress state;
- 4. The intermediate layer and the neutral layer are coincident. According to the mathematical theory of plastic bending, under the condition of pure bending, when the bending radius of the center layer of the plate is greater than 10 times thickness, the difference between the

intermediate layer and neutral layer can be neglected, and the forming process of the bilge of the hull meets these conditions.

## Mathematical model

Using the power function stress–strain mathematical model to approximate the true stress–strain relationship of the material, <sup>18</sup> the stress–strain relationship of the material is shown in equation (1) and Figure 3

$$\sigma = \begin{cases} E \cdot \varepsilon & 0 < |\varepsilon| \le \varepsilon_e \\ \sigma_0 + K \varepsilon^n & |\varepsilon| > \varepsilon_e \end{cases}$$
 (1)

where  $\sigma$  is the true stress of the steel,  $\sigma_0$  is the yield strength of steel, K is the hardening coefficient, n is the hardening index, E is elastic modulus, and  $\varepsilon_e$  is the strain value at which the material yields initially.

As shown in Figure 4, the material in the inner and outer shadow portions of the ship plate is plastically deformed, and the material in the middle portion is elastically deformed.<sup>16</sup>

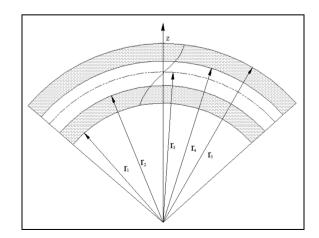


Figure 4. Three zones of plate during bending.

$$\begin{cases} r_4 = (1 + \varepsilon_e)r_3 \\ r_2 = (1 - \varepsilon_e)r_3 \end{cases}$$
 (2)

Moment of elastically deformation zone. According to the theory of metal bending forming, <sup>19</sup> the internal strain of the plate is shown in equation (3) and the bending moment is shown in equation (4)

$$\varepsilon = \frac{r - r_3}{r_3} = \frac{r}{r_3} - 1 \tag{3}$$

$$M = b \int_{r_2}^{r_4} \sigma(r - r_3) dr \tag{4}$$

where M is the bending moment,  $r_3 = 1/\overline{k}$ , b is the width of the plate, and r is the radius of any lamellar parallel to the intermediate layer.

Submit equations (1)–(3) into equation (4) and obtain the bending moment of the elastic deformation region

$$M_{2,4} = b \int_{r_2}^{r_4} \sigma(r - r_3) dr = b \int_{r_2}^{r_4} E\varepsilon(r - r_3) dr$$

$$= b \int_{r_2}^{r_4} E\left(\frac{r}{r_m} - 1\right) (r - r_3) dr = \frac{2}{3} E\varepsilon_e^2 r_3^2 b$$
(5)

Moment of elastoplastic deformation zone. Moment of plastic deformation zone on the outer side of ship plate

$$M_{4,5} = b \int_{r_4}^{r_5} \sigma_{\theta}(r - r_3) dr$$
 (6)

According to the plate strain assumption and the material incompressibility condition,  $\overline{\epsilon} = 2\epsilon_{\theta}/\sqrt{3}$ . And according to the plane strain and the Mises yield strain criterion,  $\sigma_{\theta} - \sigma_r = 2\overline{\sigma}/\sqrt{3}$ , where  $\epsilon_{\theta}$  is the longitudinal strain,  $\sigma_{\theta}$  is the longitudinal stress,  $\sigma_r$  is the radial stress, and  $\overline{\sigma}$  is the equivalent stress. When the plate is in elastoplastic bending deformation, the degree of bending deformation is small, the longitudinal stress of the material unit is much larger than the radial stress, and the longitudinal stress of the material unit is

$$\sigma_{\theta} \approx 2\overline{\sigma}/\sqrt{3}$$
 (7)

Submit equation (1) into equation (7) and obtain the plastic deformation zone to obtain the stress

$$\sigma_{\theta} \approx 2\overline{\sigma}/\sqrt{3} = 2/\sqrt{3}(\sigma_0 + k\varepsilon^n)$$
 (8)

Submit equations (1), (2), and (8) into equation (6)

$$M_{4,5} = b \int_{r_4}^{r_5} \sigma_{\theta}(r - r_3) dr = b \int_{r_4}^{r_5} 2/\sqrt{3} (\sigma_0 + k\varepsilon^n) (r - r_3) dr$$

$$= \frac{b\sigma_0}{\sqrt{3}} \left[ \left(\frac{t}{2}\right)^2 - (\varepsilon_e r_3)^2 \right] + \frac{2bk}{\sqrt{3}(n+2)r_3^2} \left[ \left(\frac{t}{2}\right)^{n+2} - (\varepsilon_e r_3)^{n+2} \right]$$
(9)

Before springback curvature. The total bending moment of the elastoplastic bending is equation (10).

$$M = M_{2,4} + 2M_{4,5} \tag{10}$$

The change in curvature due to springback is 19

$$\Delta k = \frac{M}{E \times I} \tag{11}$$

$$k_1 = k_2 + \Delta k \tag{12}$$

where  $k_1$  is the curvature before springback,  $k_2$  is the curvature after springback, and  $\Delta k$  is the curvature change.

It can be known from the equation (12) that for a plate having a single curvature shape, if the material parameters, the thickness, and the target curvature are known, the springback curvature value after unloaded can be predicted. Similarly, the curvature of the target surface is taken as the curvature after springback. The curvature before springback of the upper and lower basic groups can be obtained which will improve the springbact effect by changing the target curvature.

Curvature correction coefficient. The theoretical model is based on some assumptions, which the actual spring-back process is affected by material parameters and process parameters (such as punching pressure, friction state, stamping speed, stamping displacement, etc.). Therefore, there will be a deviation between the actual stamping and the predicted curvature. Here, accurate numerical simulation technology is used instead of experiment. Calculate the curvature correction coefficient  $\eta$ 

$$\eta = \frac{k^m}{k^t} \Rightarrow k^m = \eta \cdot k^t = \eta \cdot (k + \Delta k) \tag{13}$$

where  $k^m$  is springback curvature by numerical simulation;  $k^t$  is the springback curvature by theoretical prediction; k is the target curvature; and  $\Delta k$  is the curvature deviation.

# Surface reconstruction technology based on NURBS

### Discrete curvature

According to the curves differential geometry theory,<sup>22</sup> discrete the geometric section curve of any shape plate

into a segment arc and use the curvature ratio before and after springback of each arc to characterize the springback. Each discrete unit uses the bending theory to predict the curvature after springback and is corrected by the numerical simulation result. The discrete curvature of the compensation die surface required for the next pressing is obtained. Reconstruct the compensation surface and get the ideal part by trimming die and re-punching.

The key of this method is to transform the spring-back problem of complex irregular section stamping into the springback problem of regular equal curvature by discrete method, and obtain the compensation curvature array by curvature correction compensation algorithm. Finally, the compensation surface is reconstructed from the differential geometry algorithm of the curved curve based on the curvature data.

The bilge shells at the non-parallel middle body usually have conical surface. It can be discretized into *n* cylindrical surface according to the gradient of the curvature of its geometric section line. When *n* is relatively large, it can be considered that each small segmented curve is an equal curvature arc, and the plate is discrete into *n* cylindrical surfaces. Set the curvature of the center point of each arc to the average curvature of the arc. Finally, the geometric model of the compensation surface of the whole curvature part is obtained by using the discrete arc length curvature data of each small segments, and the initial die surface is corrected according to the geometric model.

### Discrete data

Assume that the curvature of two adjacent discrete points has been calculated, and the arc length between the two points is  $\Delta s$  ( $\Delta s = s_{i+1} - s_i$ ). Suppose the curvature between two adjacent discrete points is linear with the arc length shown in equation (14).

$$k = m \times s + n \tag{14}$$

where m and n are two coefficients. In order to determine the value of m and n, the curvature of two adjacent discrete points  $k_i$ ,  $k_{i+1}$  can be substituted into equation (15).

$$\begin{cases} k_{i+1} = m \times s_{i+1} + n \\ k_i = m \times s_i + n \end{cases}$$
 (15)

The equation (16) can be solved by equation (15)

$$\begin{cases}
 m = (k_{i+1} - k_i)/(s_{i+1} - s_i) \\
 n = (k_i \times s_{i+1} - k_{i+1} \times s_i)/(s_{i+1} - s_i)
\end{cases}$$
(16)

According to the differential geometry of the curve, equation (17) is known

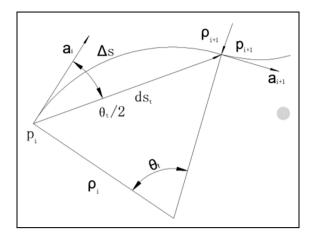


Figure 5. Recursive curve position by curvature integral.

$$\varphi = \int k(s)ds = \int (m \times s + n)ds = \frac{1}{2}m \times s^2 + n \times s + c$$
(17)

For the curve between two adjacent discrete points, the segmented curve can be further divided into a plurality of small arc segments. For the *i*-th segments, the expression of the curvature is obtained from equation (15), and then according to the integral equation (17) to obtain the expression of the equation (18)

$$\begin{cases} k_i = m \times s_i + n \\ \varphi(s_i) = \frac{1}{2} \times m \times s_i^2 + n \times s_i + c \end{cases}$$
 (18)

Then in the *i*-th segment, the equation (19) is established

$$\begin{cases} \theta_{i} = \Delta s/\rho_{i} = \Delta s \times k_{i} \\ ds_{i} = 2 \times \sin(\theta_{i}/2)/k_{i} & (k_{i} \neq 0) \\ ds_{i} = \Delta s & (k_{i} = 0) \\ \Delta x_{i} = ds_{i} \times \cos(\varphi_{i} + \theta_{i}/2) \\ \Delta y_{i} = ds_{i} \times \sin(\varphi_{i} + \theta_{i}/2) \end{cases}$$
(19)

See Figure 5, where  $\rho_i$ ,  $\rho_{i+1}$  is the radius of the curvature at two adjacent points  $P_i$ ,  $P_{i+1}$  on the curve, the corresponding curvature is  $k_i$  and  $k_{i+1}$ .  $\Delta s$  and  $ds_i$  are the arc length and chord length.  $\theta_i$  is the corresponding central angle, and  $\Delta x_i$ ,  $\Delta y_i$  is the coordinate increment of  $P_{i+1}$  relative  $P_i$ .

Then get the coordinate recursion equation (20) from  $P_i$  to  $P_{i+1}$ 

$$\begin{cases} x_{i+1} = x_i + \Delta x_i \\ y_{i+1} = y_i + \Delta y_i \end{cases}$$
 (20)

According to equation (20), the position of the curve can be obtained point by point.<sup>23</sup>

# Discrete point surface reconstruction based on NURBS

The NURBS surface equation obtained by the parameters in  $\mu$ ,  $\nu$  directions are shown in equation (21)

$$S(u,v) = \frac{\sum_{i=0}^{n} \sum_{j=0}^{m} \varpi_{i,j} d_{i,j} N_{i,k}(u) N_{j,l}(v)}{\sum_{i=0}^{n} \sum_{j=0}^{m} \varpi_{i,j} N_{i,k}(u) N_{j,l}(v)}$$
(21)

where  $d_{i,j}$  is the control grid point of the matrix composed of discrete points  $S^{point}$ ;  $\varpi_{i,j}$  is a set of weight factors greater than 0 associated with  $d_{i,j}$ ;  $N_{i,k}(u)$  is the k-direction spline function in the  $\mu$  direction; and  $N_{j,l}(v)$  is the L-order spline function in the  $\nu$  direction, where the basic functions of the surface and the curve are the same.<sup>24</sup>

In this article, the NURBS method is used and the surface is controlled by the control points obtained in "Discrete data" section, so that the shape of the surface can be flexibly changed by editing the control points, and the discrete nodes are fitted into curves by NURBS method, which can improve the smoothness of the curve and get a smoother surface. CATIA software<sup>25</sup> is selected to surface reconstruction, which facilitates the processing of the generated surface in the subsequent steps, ensuring the consistency of the surface in the design and manufacturing process.

Springback compensation control flow chart based on NURBS is shown in Figure 6.

# Example of equal curvature springback compensation control

## Numerical simulation

In order to obtain the springback curvature correction coefficient, the numerical simulation of the cylindrical sheet with a curvature radius of 2000 mm is carried out. The key techniques of numerical simulation are as follows.

Establishment of 3D model. Solidworks is used to establish a 3D model of finite element multi-point forming. There is a rotating spherical crown under the square head die, which allows the head to swing freely. In the contact stage, the remaining parts are not in contact with any part, except the upper surface of the square head is in contact with the sheet. In order to save the modeling workload and calculation time, the spherical crown behind the square head is simplified shown in Figure 7. The upper dies and the lower dies adopt a misalignment distribution. The model is established shown in Figure 8.

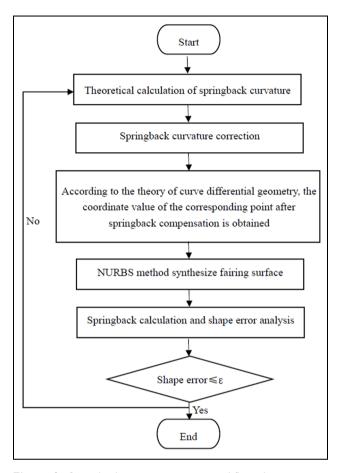


Figure 6. Springback compensation control flow chart.

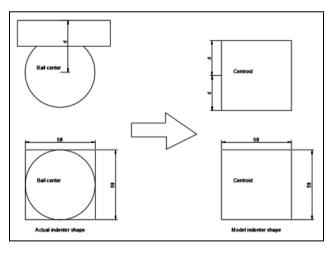


Figure 7. Simplify the head die process.

Material properties. In the multi-point forming analysis, during the forming process the deformation of the die is small, so it is treated as a rigid body. The elements of the rigid body material are not involved in the calculation of stress and strain, and the elastic modulus and Poisson's ratio are mainly used to calculate the contact

Table 1. Properties of material.

R (mm)	B (mm)	t (mm)	L (mm)	$\sigma_{\rm s}({\sf MPa})$	E (GPa)	λ	K	n
2000	2000	16	2000	345	206	0.3	630.3	0.236

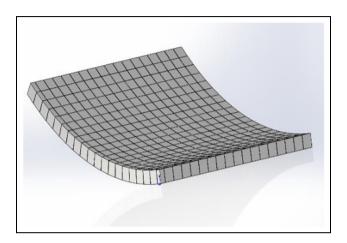


Figure 8. The die model.

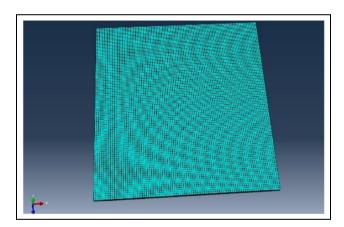


Figure 9. Meshing of plate.

force on the surface when the sheet and the die come into contact. The plate material parameter settings are shown in Table 1.

In Table 1, R is cylindrical plate radius of curvature; B is shell width; L is shell length; t is shell thickness;  $\sigma_s$  is yield strength; E is elastic modulus;  $\lambda$  is Poisson's ratio; K is hardening coefficient, and n is hardening index.

Contact conditions. The contact mode between the die and the sheet is universal contact. The friction model directly affects the calculation accuracy. The modified Coulomb friction model is used in the article.<sup>25</sup> This is a friction model that has been proved to be more

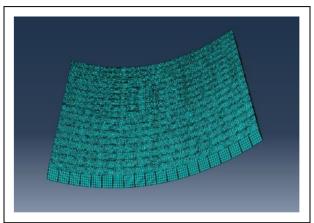


Figure 10. Meshing of upper die.

realistic in practice. The friction coefficient is set to 0.1 in the article.

In the springback phase, in order to visually observe the springback displacement of the plate, the center point of the plate is rigidly fixed to avoid unnecessary displacement.

Unit and meshing. The plate is made of a structural hexahedron mesh, and the number of meshes is divided. The number of meshes with 1000 mm lengths on both sides is 100 and the thickness direction is 4 (see Figure 9). Since the upper and lower dies are rigid bodies, in order to reduce calculation, the mesh size is 11.6 mm and the maximum deviation factor is 0.1. The upper die mesh is shown in Figure 10. The forming stage unit type is selected as explicit, linear geometric order, C3D8R eight-node linear hexahedral element, reduced integral and hourglass control. In the springback phase unit is selected as implicit, linear geometric order, C3D8R eight-node linear hexahedral element, reduced integral and hourglass control.

Calculation results analysis. The displacement distribution cloud after springback is shown in Figure 11. The maximum springback displacement is 45.47 mm. The radius of the curvature after springback is 2412.4 mm by geometric calculation.

# The springback curvature correction coefficient

It can be seen from the Figure 11 that the springback gradually increases from the center to the edge of the

Table 2. Springback curvature correction coefficient.

Radius of curvature before springback	2500	2200	2000	1800	1500
Correction coefficient	1.026	1.023	0.971	0.941	0.949

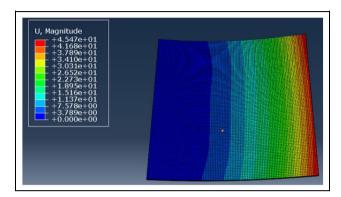


Figure 11. Displacement distribution cloud after springback.

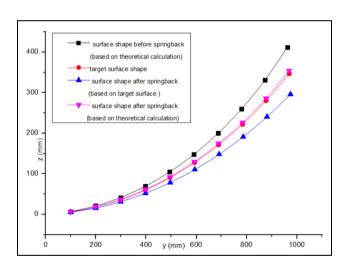


Figure 12. Forming result shape comparison.

shell and the maximum springback is 45.47 mm. The spline curve is obtained with a radius of curvature of about 2412 mm after springback. Using the above theoretical study, the target radius of curvature is 2412 mm and the radius of the curvature before springback should be 1923 mm. Curvature correction coefficient is 1.040. In order to improve the calculation accuracy, five sets of data are selected to average shown as Table 2, finally select the springback curvature correction coefficient is 0.982.

# Numerical simulation verification

According to equations (5) and (9)–(13), the radius of the curvature of the Q345 plate with a target curvature



Figure 13. The lower dies curved surface.

of 2000 mm is calculated to be 1677 mm. In order to verify the accuracy of the theoretical calculation method, numerical simulation method is use to verify the validation. The calculation results are shown in Figure 12. It can be seen from the figure that the curvature radius before springback is calculated by the method in this article, and the curvature after springback is very close to the target curvature.

## Experimental verification

The springback experiment of Q345 plate with a curvature radius of 1677 mm was carried out. Import the prepared experimental data into the instrument, and the instrument will automatically adjust the position of the lower dies according to the data. After the completion of the profile adjustment, the lower dies will form a certain curved surface, as shown in Figure 13.

In the process of shape adjustment, the indenter moves vertically, the indenter plane is horizontal, and the surface formed is not completely smooth. In the process of pressing, the indenter will rotate under the force of the upper indenter and finally fit into a better surface. Then fix the plate on the formed die, as shown in Figure 14, and operate the upper indenter to lower and compact, as shown in Figure 15. After pressing, keep the pressure for a certain period of time to prevent the plate from springback immediately. The effect after pressing is shown in Figure 16. It can be seen that the non-symmetric pressure head can still be successfully fitted into a suitable surface. In addition, compared with other types, there is no need to add special blank holder around the plate.

The measurement is carried out after pressing. Figure 17 shows the comparison between the

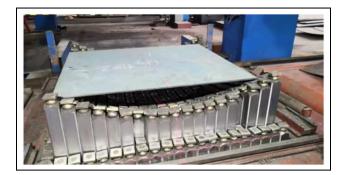


Figure 14. The plate on the lower dies.



Figure 15. Plate is compacted by dies.

measurement results of the laser measuring instrument and the target surface. Red is the measurement value and black is the theoretical value. The other is to measure on the surface with the already made template. Figure 18 is to measure with the template.

After the experiment, the radius of curvature measured after springback is 2025mm, while the radius of curvature calculated by theory is 2000mm, the relative accuracy is 98.77%, which meets the requirements of process accuracy.

# **Example of variable curvature springback** compensation control

The bilge of a ship in a non-parallel middle body is a conical surface. Its properties of material are same as Table 1. The partially discrete curvature along the length direction is shown in Table 3. The target shape is shown in Figure 19.

The linear interpolation method is used to discretize the curvature. Based on the two-dimensional spring-back calculation theory, the radius of the curvature before springback of each discrete section is obtained, and the above-mentioned curvature-based surface reconstruction method is used to obtain the discrete points. Some data are shown in Table 4. CATIA<sup>26</sup> method is used for surface reconstruction shown in

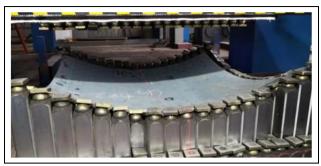


Figure 16. The effect after pressing.



Figure 17. Measurement results of the laser.



Figure 18. Measure with the template.

Figure 20. The surface shape before and after spring-back is shown in Figure 21.

### Conclusion

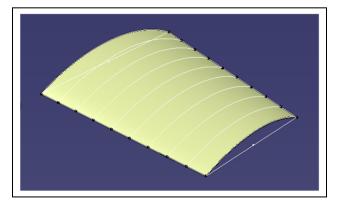
Although the theory of two-dimensional sheet metal forming is a traditional topic, with the continuous

**Table 3.** Partially discrete curvature along the length direction.

x	0	200	400	600	800	1000	1200	1400	1600	1800	2000
Radius of curvature	2000	1950	1900	1850	1800	1750	1700	1650	1600	1550	1500

 Table 4. Discrete data points before springback.

X =	0										
у	0	99.87	199.46	298.67	397.40	495.51	592.88	689.36	784.77	878.92	971.57
z	0	5.90	17.73	35.55	59.47	89.63	126.23	169.55	219.92	277.76	342.45
X =	200										
у	0	97.87	197.47	294.68	391.42	487.57	582.99	677.55	771.08	863.38	954.23
z	0	5.78	17.73	35.30	58.85	88.52	124.52	167.10	216.61	273.47	337.69
X =	400										
у	0	95.87	195.47	290.69	385.44	479.62	573.09	665.72	757.36	847.81	936.85
z	0	5.66	17.74	35.07	58.26	87.46	122.85	164.71	213.38	269.27	332.90
X =	600										
у	0	93.88	193.47	286.70	379.46	471.66	563.18	653.88	743.62	832.20	919.42
z	0	5.55	17.76	34.86	57.70	86.43	121.24	162.39	210.23	265.18	328.11
X =	800										
у	0	91.88	191.48	282.70	373.47	463.70	553.26	642.02	729.85	816.57	901.95
z	0	5.43	17.79	34.67	57.17	85.43	119.67	160.14	207.17	261.19	323.30
X =	1000										
у	0	89.88	189.48	278.70	367.49	455.73	543.32	630.15	716.06	800.89	884.43
z	0	5.32	17.84	34.49	56.66	84.48	118.16	157.95	204.19	257.31	318.48
X =	1200										
у	0	87.89	187.49	274.71	361.49	447.75	533.38	618.26	702.25	785.19	866.87
z	0	5.21	17.90	34.34	56.18	83.57	116.69	155.82	201.29	253.53	313.65
X =	1400										
у	0	85.89	185.49	270.71	355.50	439.77	523.43	606.35	688.42	769.45	849.27
z	0	5.10	17.97	34.21	55.74	82.70	115.29	153.77	198.49	249.87	308.81
X =	1600										
у	0	83.89	183.50	266.71	349.50	431.78	513.46	594.43	674.56	753.68	831.61
z	0	4.99	18.07	34.11	55.33	81.87	113.94	151.79	195.77	246.31	303.95
X =	1800										
у	0	81.89	181.50	262.70	343.49	423.79	503.49	582.49	660.67	737.88	813.92
z	0	4.89	18.18	34.02	54.95	81.09	112.65	149.89	193.15	242.87	299.09
X =	2000										
у	0	79.89	179.50	258.70	337.49	415.78	493.50	570.53	646.76	722.04	796.17
z	0	4.78	18.31	33.97	54.60	80.35	111.41	148.06	190.62	239.54	294.20



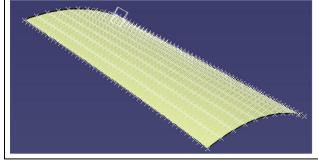


Figure 20. Curved surface shape before springback.

Figure 19. Target surface shape.

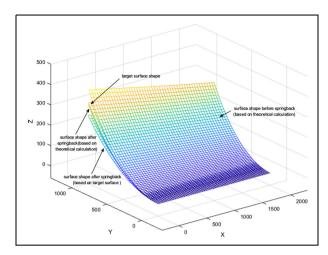


Figure 21. Surface shape before and after springback.

development of mechanization, intelligence and simulation technology, the forming equipment is constantly updated. The theoretical calculation model of sheet metal forming is corrected based on the advanced equipment of asymmetric pressure square head CNC cold bending machine. The following conclusions can be obtained by calculations.

We can find the theoretical springback formula has a certain range of application, from equation (9). When the radius of the curvature is greater than 4770 mm, the plastic deformation is small and the moment of plastic deformation zone on the outer side of ship plate is negative value. This formula will go beyond the scope of use and no longer valid.

Forming theory is unlikely to produce very accurate springback results, but it is very close to the desired result, so that the times of stamping can be reduced as much as possible and improve the processing efficiency.

In addition, the results of theoretical analysis and simulation need to be verified by a large number of experiments. The theoretical model and the finite element model should be corrected by the results of the experiments, and finally a more accurate model will be obtained. Due to time and cost, only some experimental data have been added in the article. In the next work, we will use more experimental data and accurate numerical simulation algorithm, and try to apply the theory and method of two-dimensional sheet metal forming to three-dimensional sheet metal forming.

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