

1 **Determination of strain rate dependence at intermediate strain rates using acceleration**
2 **information**

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17 **ABSTRACT**

18 Precise stress-strain characteristics of materials for intermediate strain rates need to be utilized for
19 analyzing various events in industry such as metal forging, sheet metal forming in manufacturing
20 processes and automotive crash tests. However, the accurate evaluation of the load is not easy at
21 intermediate or high strain rates, owing to the inertial effect. The present study aims at characterizing
22 the hardening behavior using acceleration data without utilizing the load information with an
23 application to a dual-phase DP980 steel sheet sample. Virtual measurements were obtained from a
24 finite element model to check for the minimum acceleration magnitude necessary for stable
25 identification. The same identification procedure as that used for the experiments was adopted. Also,
26 a high-speed tensile testing equipment for steel sheet specimens was modified to increase the
27 acceleration magnitude to implement the proposed methodology experimentally. The virtual fields

28 method was chosen as an inverse tool to determine the strain-rate dependence of the sheet metal
29 specimens. The stress–strain curve of an advanced high-strength steel at intermediate strain rates
30 obtained from the acceleration was compared with the curve from the load data, and promising results
31 were obtained.

32

33 Keywords: Dynamic hardening behavior; Inverse identification; Digital image correlation;
34 Intermediate strain rates; Sheet metal; Acceleration;

35

36 1. Introduction

37 Material behaviors at intermediate strain rates ($5 \text{ s}^{-1} \sim 500 \text{ s}^{-1}$) [1] have gained great interest to
38 understand the mechanisms and predict the behaviors in various fields of industry such as metal
39 forging, sheet metal forming in manufacturing processes [2,3] and automotive crash tests [4-
40 6]. Stress-strain characteristic of materials at intermediate strain rates is the fundamental information
41 to analyze such events.

42

43 Numerous experimental techniques have been developed to identify the strain-rate dependence at
44 high rates of deformation since commonly, material properties change depending on the strain rate.
45 Based on the method of experimentally generating dynamic conditions, high-speed test systems can
46 be classified into several groups [7,8]: methods utilizing potential energy, such as the drop weight
47 test, methods employing kinetic energy, such as the Taylor impact test, split Hopkinson pressure bars
48 using kinetic energy and stress wave propagation, and servo-hydraulic machines. Among those, high-
49 speed servo-hydraulic test machines have been widely used to obtain stress–strain curve information
50 at various intermediate strain rates [9].

51

52 Nonetheless, the dynamic strain-hardening behavior of metallic materials at intermediate or high
53 strain rates is not characterized easily since the precise measurement of the applied load is difficult

54 owing to the effect of inertia [10]. Stress waves are produced at high strain rates, inducing high
55 acceleration magnitudes. This results in the ringing problem in load cells of standard test machines
56 [11].

57

58 Consequently, numerous studies have been conducted to mitigate the load-ringing phenomenon.
59 Researchers have primarily focused on investigating three approaches. The first involves directly
60 acquiring the load from a dogbone specimen by bonding strain gauges to its gripped section which
61 deforms elastically [9,12]. The second is to increase the natural frequency of a jig part between the
62 load cell and the specimen [11]. The final approach is to develop a load sensor by applying strain
63 gauges to lightweight grips [9,13,14] because the ringing can be significant if the mass of the part
64 between the specimen and the load cell is large [15].

65

66 Alternatively, several attempts have been made to determine material parameters at high rates using
67 acceleration information instead of load data. The idea is to use the acceleration field as an embedded
68 load cell through the dynamic equilibrium equation, provided that full-field displacements are
69 available experimentally with sufficient temporal resolution to accurately derive accelerations.
70 Moulart et al. [16], Pierron and Forquin [17], and Pierron et al. [18] used this concept to identify the
71 elastic moduli of concrete [17] and composite [16,18] specimens. They used the grid method [19] as
72 a full-field measurement technique to measure the strain and acceleration fields, in combination with
73 the virtual fields method (VFM) [20] as an inverse method to identify material parameters at high
74 rates. The VFM is an inverse identification method for acquiring the constitutive parameters
75 following the principle of virtual work.

76

77 In addition, the VFM was used for elasto-plastic cases to characterize dynamic behaviors using only
78 the acceleration information without acquiring the load [21-24].

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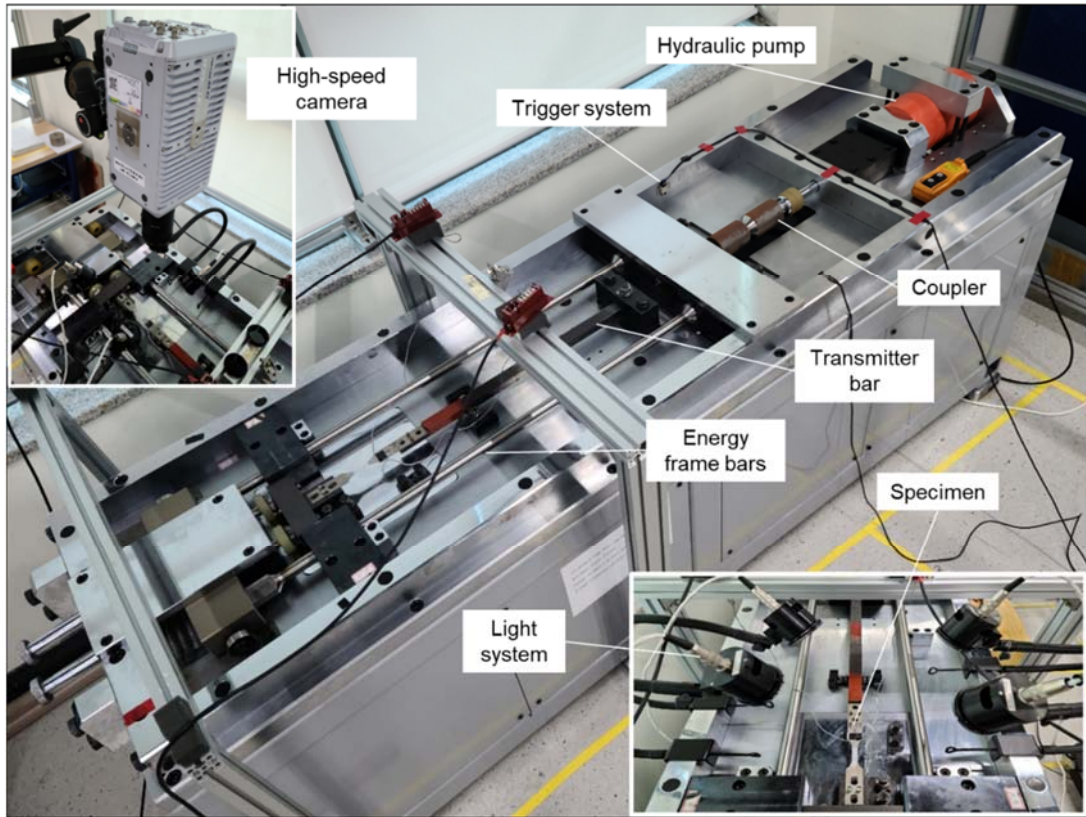
80 Recently, several studies tried to calibrate the rate-dependent hardening properties from strain and
81 acceleration fields adopting the image-based inertial impact (IBII) test [18]. For elasto-plasticity, the
82 test is akin to a Taylor impact test but with a flat plane stress specimen. Bouda et al. [23] optimized
83 the IBII test to characterize the dynamic behavior of titanium at high rates by optimizing the specimen
84 geometry through a computational approach. Fourest et al. [24] retrieved the parameters of a
85 Johnson–Cook model for a titanium alloy. Linear and Voce hardening parameters at different strain
86 rates were identified for stainless steel from nonhomogeneous strain-rate data in Fletcher et al. [25].
87
88 However, the target strain rates in the above studies were very high (up to $3,000\text{ s}^{-1}$) and not in the
89 intermediate range. The objective of this study is to identify the dynamic hardening parameters at
90 intermediate strain rates for thin steel sheet material by using acceleration rather than load cell data.
91 The VFM was adopted in this study as an inverse tool to retrieve the strain-rate dependence of the
92 sheet metal specimens. The idea of determining the strain rate dependence for various strain rates
93 comes from the observation that the strain rate distribution is heterogeneous over the area of interest
94 (AOI) [27]. This means that sufficient strain rate information can be fed into the VFM to retrieve the
95 strain rate dependence with a proper rate dependent model. FE simulations were performed using the
96 Abaqus software to check the minimum acceleration required because Leem et al. [26] revealed that
97 a critical magnitude of acceleration is necessary for identification using the VFM. Subsequently, a
98 high-speed tensile equipment for sheet metal specimens [27] was modified to increase the
99 acceleration based on the observations from the FE simulation. Several aspects of the modification
100 were closely investigated to optimize the configuration of the tester and specimen. The digital image
101 correlation (DIC) technique was applied to calculate the displacement, strain, strain rate, and
102 acceleration during the experiments using a high-speed camera. Then, identification using the VFM
103 with the Johnson–Cook rate-dependent model was performed from the measured quantities. Finally,
104 some features of the test results are discussed.

105

106 2. Methodology

107 2.1 Impact frame high-speed (IFHS) tensile equipment

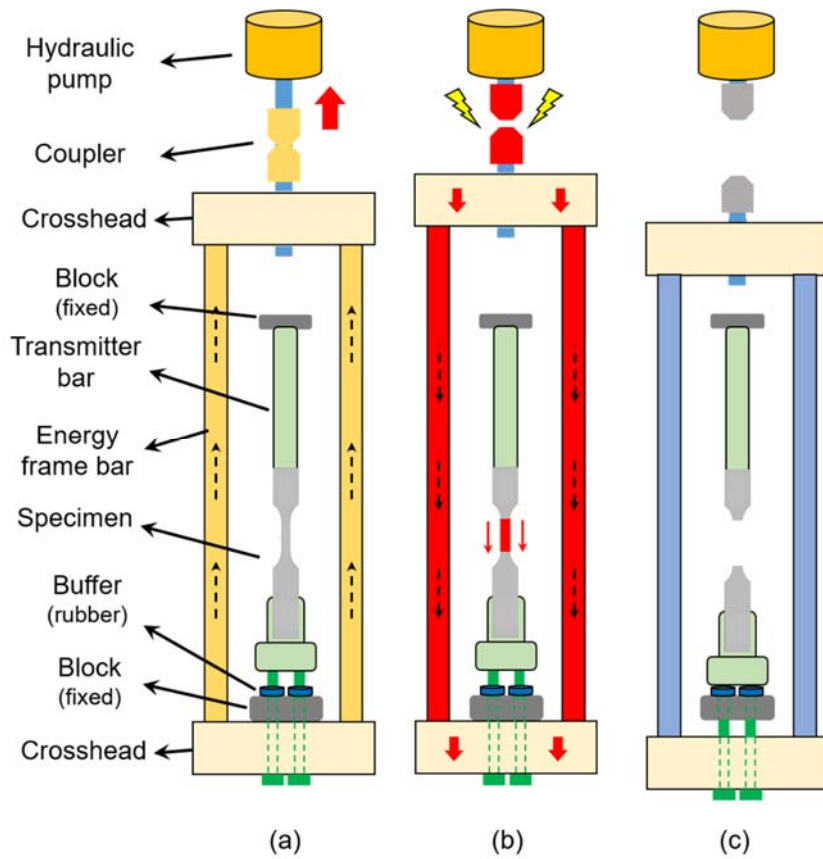
108 A new type of high-speed tensile test equipment for steel sheet specimens on the basis of the work of
109 Tran and Kim [28] was manufactured [27]. Views of the experimental equipment are presented in
110 Fig. 1 and the testing principle is shown in Fig. 2. Unlike typical impact testing systems that utilize
111 potential or kinetic energy, this one uses the elastic strain energy stored in frame bars to generate
112 high-speed impact pulses. Two frame bars are linked to a hydraulic pump using a coupler. The coupler
113 is in the form of a cylinder with a notch at the center; therefore, the coupler breaks when a critical
114 amount of load is applied. Once the pump applies a tensile force to the coupler, it sustains the load up
115 to a critical value of load as shown in Fig. 2(a). In the meanwhile, the frame bars accumulate elastic
116 strain energy simultaneously. If the load is high enough to fracture the coupler as shown in Fig. 2(b),
117 the elastic strain energy stored in the frame bars is converted to kinetic energy to draw the steel sheet
118 specimen attached to the frame parts in tension at a high strain rate, as shown in Fig. 2(c). In addition,
119 photoelectric sensors (transmitter and receiver) are located at both sides of the coupler as a trigger
120 system in Fig. 1. This trigger system sends a signal to the high-speed camera when the coupler breaks
121 so that the high-speed camera can capture deformation images. Adjusting the designs of the frame
122 bars and the coupler can change the strain rate.



123

124 **Fig. 1.** Views of the IFHS tester.

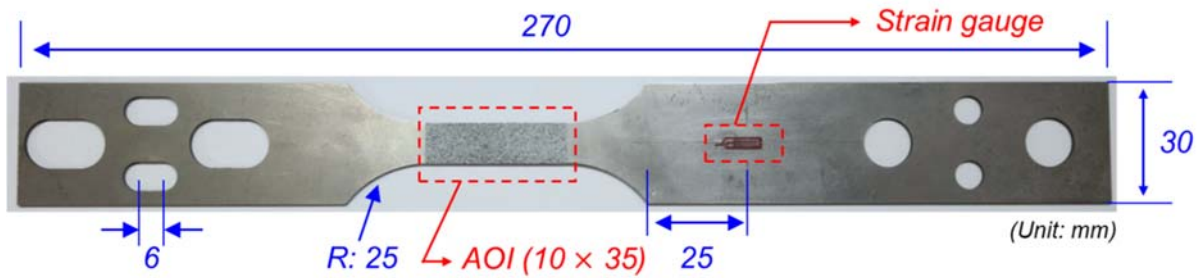
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126

127 **Fig. 2.** The principle of the IFHS tester.

128 A dogbone test specimen was designed for the experiments, as depicted in Fig. 3. The specimen has
129 circular holes on one grip end, and rounded rectangles on the other end. The left-hand side with the
130 rounded rectangles acts as a slack adaptor, allowing the movable part to the left to gather enough
131 speed before the load is transmitted to the specimen. In this study, a dual-phase DP980 steel sheet
132 specimen with 1.2 mm thickness was chosen.



133
134 **Fig. 3.** Geometry of specimen for the IFHS tests (unit: mm).

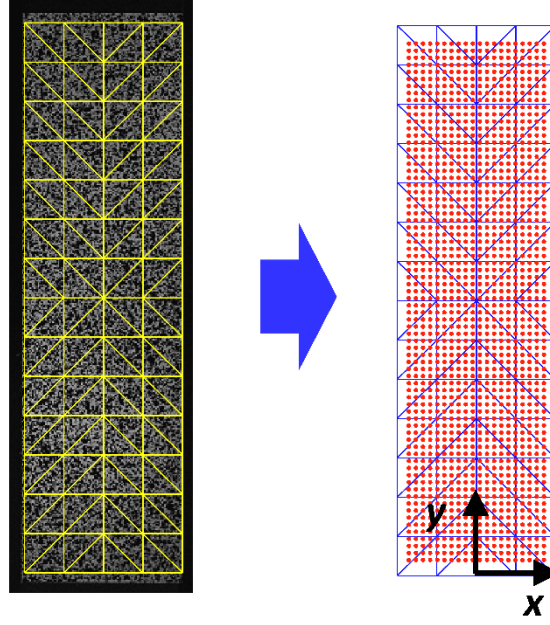
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136 2.2 Full-field measurements with DIC technique

137 A speckle pattern was sprayed with black paint on the area of interest (AOI) of steel sheet specimens
138 initially covered with white paint for the DIC analysis [29]. The speckle pattern images were captured
139 using a high-speed camera at 120,000 frames per second (fps) during the high-speed tensile tests. The
140 size of the AOI was 35 mm (initial gauge length) \times 10 mm (width), sampled by 640×128 pixels.
141 Vic-2D software from Correlated Solutions was utilized for the DIC calculation. Information on the
142 imaging and DIC parameters is provided in Table 1, according to the recommendations of Jones and
143 Iadicola [30]. The red dots in the AOI (Fig. 4) represent the measured DIC points.

144

145 In order to calculate the logarithmic strain and acceleration fields to be used for determination of
146 strain-rate dependence using the VFM, the exact AOI dimension was obtained from a FE software
147 and the whole AOI was meshed using triangular elements as shown in Fig. 4.



148

149 **Fig. 4.** Speckle patterns in the AOI and measured DIC points (y and x indicate the rolling and
 150 transverse directions, respectively).

151

152 Based on the known coordinates of each triangle nodes and of the measured DIC points inside each
 153 triangle in the undeformed configuration, the mapping which projects the measured DIC points onto
 154 the nodal points can be calculated using a basis of piecewise linear functions [31] as adopted in the
 155 FE method [32]. This procedure is conducted by linear interpolation (least-squares fitting). Next, the
 156 nodal coordinates of each triangle in the deformed configuration at each loading stage are
 157 approximated from the coordinates of measured DIC points in the deformed configuration by
 158 applying the piecewise functions. The relationship between the deformed DIC points and the
 159 deformed nodal points can be expressed as:

$$\{\bar{x}\} = [\Phi]\{\bar{X}\} \quad (1)$$

160 where $\{\bar{x}\}$ is the location vector of the measured DIC points, $[\Phi]$ the matrix of the piecewise linear
 161 functions and $\{\bar{X}\}$ the location vector of nodal points. Then, the location vector of nodal points can
 162 be obtained using the explicit direct solution to the least squares minimization problem [32]:

$$\{\bar{X}\} = ([\Phi]^T[\Phi])^{-1}[\Phi]^T\{\bar{x}\} \quad (2)$$

163 This fitting process has the advantage of allowing for the reconstruction of data up to the specimen

164 edges, which is critical for the VFM. In addition, the fitting process has an effect of spatial smoothing
 165 to reduce the experimental noise.

166

167 Next, the linear relationship between the nodal coordinates of each triangle in the undeformed and
 168 deformed configuration can be calculated using a 2D affine transformation:

$$x_d = a_1 + a_2 X_u + a_3 Y_u, \quad y_d = a_4 + a_5 X_u + a_6 Y_u \quad (3)$$

169 where x_d, y_d are nodal coordinates in the deformed configuration and X_u, Y_u nodal coordinates
 170 in the undeformed configuration. The six coefficients ($a_1 \sim a_6$) for each triangle can be obtained from
 171 six equations (two equations per node and total three nodes). Finally, the deformation gradient F in
 172 each triangle is calculated by Eq. (4) assuming a plane stress and the incompressibility in plasticity
 173 ($\det(F) = 1$).

$$F = \begin{bmatrix} \frac{\partial x}{\partial X} & \frac{\partial x}{\partial Y} & 0 \\ \frac{\partial y}{\partial X} & \frac{\partial y}{\partial Y} & 0 \\ 0 & 0 & \frac{1}{\frac{\partial x \partial y}{\partial X \partial Y} - \frac{\partial x \partial y}{\partial Y \partial X}} \end{bmatrix} = \begin{bmatrix} a_2 & a_3 & 0 \\ a_5 & a_6 & 0 \\ 0 & 0 & \frac{1}{a_2 a_6 - a_3 a_5} \end{bmatrix} \quad (4)$$

174 Now, the logarithmic strain tensor ε_{ln} is calculated from the deformation gradient F through the
 175 right stretch tensor U ($U^2 = F^T F$) as in Eq. (5).

$$\varepsilon_{ln} = \sum_{i=1}^3 \ln(\lambda_i) r_i \otimes r_i \quad (5)$$

176 where λ_i and r_i are the eigenvalues and eigenvectors of the right stretch tensor U respectively.

177

178 The strain rate $\dot{\varepsilon}$ in each triangular element at each time step is calculated from the obtained
 179 logarithmic strain. In the rest of manuscript, ‘‘AOI-averaged’’ indicates the average of values in all
 180 triangular elements.

$$\dot{\varepsilon}_i \left(t + \frac{\Delta t}{2} \right) = \frac{\varepsilon_i(t + \Delta t) - \varepsilon_i(t)}{\Delta t} \quad (6)$$

181 where ε indicates strain, i corresponds to x , y , or xy and t represents time. Also, the
182 displacement u can be derived at the centroid of each triangle because the nodal coordinates of each
183 triangle in the undeformed and deformed configuration are known. Then, the velocity v and the
184 acceleration a in each triangular element are obtained at each time step from the measured
185 displacement u by simple finite difference and second order finite differences, respectively.

$$v_j \left(t + \frac{\Delta t}{2} \right) = \frac{u_j(t + \Delta t) - u_j(t)}{\Delta t} \quad (7)$$

186 where j can be either x or y .

187

$$a_j(t) = \frac{u_j(t + \Delta t) + u_j(t - \Delta t) - 2u_j(t)}{\Delta t^2} \quad (8)$$

188

189 In order to calculate the strain and acceleration resolution (the smallest value which can be detected
190 above the noise), 21 successive speckle pattern images were obtained at rest using a high-speed
191 camera at 120,000 fps. Then, the logarithmic strain and acceleration were derived as described above
192 and the standard deviation values of the obtained logarithmic strain and acceleration in the triangle
193 elements were calculated. The information of strain and acceleration resolution is presented in Table 1.

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208 **Table 1**

209 Information on the DIC technique (DP980).

Camera	Photron FASTCAM SA-X2
Sensor and digitization	1024 × 1024 pixels, 12-bit
Lens, imaging distance	Sigma 105 mm Macro Lens, 390 mm
Total image number (recording time)	54 (0.45 ms)
Frame rate	120,000 fps
Image resolution (in pixel size)	640 × 128
Conversion of pixel-to-mm	1 pixel = 0.089 mm
Subset and its offset (step)	21, 5 pixels
Interpolation, shape functions, correlation criterion	Cubic spline, affine, Normalized squared differences
Strain smoothing technique	Linear shape functions (triangular finite element)
Strain resolution	1.21×10^{-5} (ε_{xx}), 1.67×10^{-5} (ε_{yy}), 7.29×10^{-5} (ε_{xy})
Acceleration resolution	3.54 km/s^2 (a_{xx}), 3.62 km/s^2 (a_{yy})

210

211 2.3 Constitutive model

212 An elasto-plastic constitutive model that can properly capture the dynamic strain-hardening behavior
 213 is required. In this research, the plane stress von Mises yield criterion and the Johnson–Cook model
 214 for a rate-dependent hardening law were taken because the flow stress of steels depends on the strain
 215 rate. In particular, advanced high-strength steels (AHSSs), prominent in automotive applications, tend
 216 to exhibit a larger flow stress at higher strain rates [33]. The classical associated flow rule was
 217 assumed.

218

219 The yield condition can be expressed as

$$f(\sigma, \varepsilon_p) = \sigma_{eq}(\sigma) - \sigma_s(\varepsilon_p) = 0 \quad (9)$$

220 where σ_{eq} denotes the von Mises equivalent stress, σ_s denotes the current flow stress, and ε_p
 221 denotes the equivalent plastic strain.

222

223 The Johnson–Cook model has quasi-static and dynamic terms,

$$\sigma_s = \sigma_r \left\{ 1 + C \ln \left(\frac{\dot{\varepsilon}_{p,d}}{\dot{\varepsilon}_{p,r}} \right) \right\} \quad (10)$$

224 where σ_r and $\dot{\varepsilon}_{p,r}$ indicate the stress and reference equivalent plastic strain rate in quasi-static
 225 condition, respectively, and $\dot{\varepsilon}_{p,d}$ is the equivalent plastic strain rate in dynamic condition. $\sigma_r(\dot{\varepsilon}_{p,r})$
 226 can be any hardening law. The strain-rate sensitivity C is a constant in the Johnson–Cook model.

227

228 In this study, the hardening parameters of σ_r are obtained from quasi-static uniaxial tensile tests, and
 229 the dynamic parameter C is calibrated with the VFM using acceleration data from intermediate-
 230 strain-rate experiments to lower the parameter numbers to be determined using the VFM.

231

232 2.4 The virtual fields method (VFM)

233 In this research, the virtual fields method (VFM) was chosen as an inverse identification scheme to
 234 determine the Johnson–Cook material parameter C . The VFM uses the principle of virtual work,
 235 which indicates the global equilibrium condition. The equilibrium equation in dynamic elasto-
 236 plasticity can be presented as follows (in the absence of body forces):

$$-\int_V \left[\int_0^t \frac{d\sigma_{ij}}{dt} dt \right] \varepsilon_{ij}^* dV + \int_{S_f} T_i u_i^* dS = \int_V \rho a_i u_i^* dV \quad (11)$$

237 where $d\sigma/dt$ is the stress rate tensor, V the measured volume, T the force distribution applied on
 238 the specimen boundary S_f , ε^* the virtual strain field obtained from the virtual displacement field
 239 u^* , ρ the density, and a is the acceleration. The summation convention is applied to repeated
 240 indices.

241

242 The equilibrium equation describes that the external virtual work (EVW) from the surface tractions
243 T is equal to the sum of the acceleration virtual work (AVW) due to the acceleration a and the
244 internal virtual work (IVW) due to the stress. The AVW is negligible during quasi-static conditions
245 but should be included in the VFM identification at high rates, since the acceleration effects cannot
246 be neglected.

247

248 The entire loading history needs to be considered for the identification of the elasto-plastic behavior.
249 An iterative procedure is needed to minimize the difference between the (EVW – AVW) and IVW
250 at each deformation step [34]. The stress increments were calculated repeatedly at each time step until
251 the virtual work gap was minimized. The radial return algorithm suggested by Sutton et al. [35] was
252 used for stress calculation. The nonlinear least squares algorithm in MATLAB was utilized for the
253 optimization.

254

255 In order to conduct the optimization to extract the dynamic parameter, suitable virtual fields must be
256 selected. The virtual displacement fields need to be continuous and piecewise differentiable across
257 the AOI. Two approaches were adopted to identify the parameter C of the Johnson–Cook model in
258 this study. The first was to use the acceleration data without load, and the second was to utilize the
259 load data only for the validation of the results obtained using the acceleration. Therefore, two sets of
260 virtual fields were chosen for each case. The first set of virtual fields for acceleration is given by
261 Eq. (12). In this case, u^* is chosen to eliminate the effect of the EVW term during the iteration.

$$u_{1x}^* = 0, u_{1y}^* = (y - y_{min})(y - y_{max}) \quad (12)$$

262 where x and y denote the coordinates in the horizontal and vertical directions, respectively, as shown
263 in Fig. 4. The second set of virtual fields for the load is given by Eq. (13).

$$u_{2x}^* = 0, u_{2y}^* = y \quad (13)$$

264

265 It should be noted here that regarding the VFM identification using the load data and the second set

266 of virtual fields for the validation, the same identification scheme adopted in Park et al. [27] will be
267 used for the rest of this study.

268

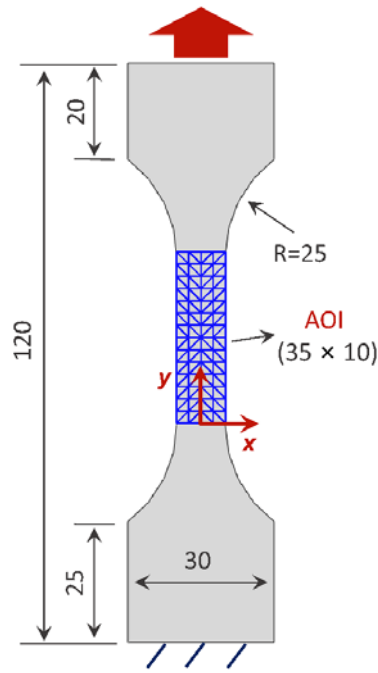
269 3. FE simulation

270 The specific goal of the FE simulation was to verify the VFM identification routines of the Johnson–
271 Cook parameter using acceleration, and to find the minimum acceleration magnitude required for the
272 identification. Virtual measurement data was obtained using the FE program Abaqus/Explicit. The
273 same identification procedure used for the experiments was applied as well in this case.

274

275 3.1 FE model

276 An FE model according to the real geometry with the 1.0 mm thick specimen was built, as presented
277 in Fig. 5. To imitate measured DIC points, four-node shell elements (S4R) and a fine mesh size (0.3
278 mm) were chosen. The AOI was selected to be the central area with a 35 mm gauge length and meshed
279 as described in Section 2.2. The virtual measurement data was obtained at uniformly spaced time
280 intervals (100 time steps). From each triangular element, the strain and acceleration were calculated
281 at each time step. The load was measured using the reaction force acting on the bottom edge, as shown
282 in Fig. 5. Table 2 lists the simulation conditions used in this study.



283

284 **Fig. 5.** Model for the dynamic FE simulations (unit: mm, upper edge: velocity boundary condition

285 & $U_x=0$, bottom edge: $U_x=U_y=0$).

286

287 **Table 2**

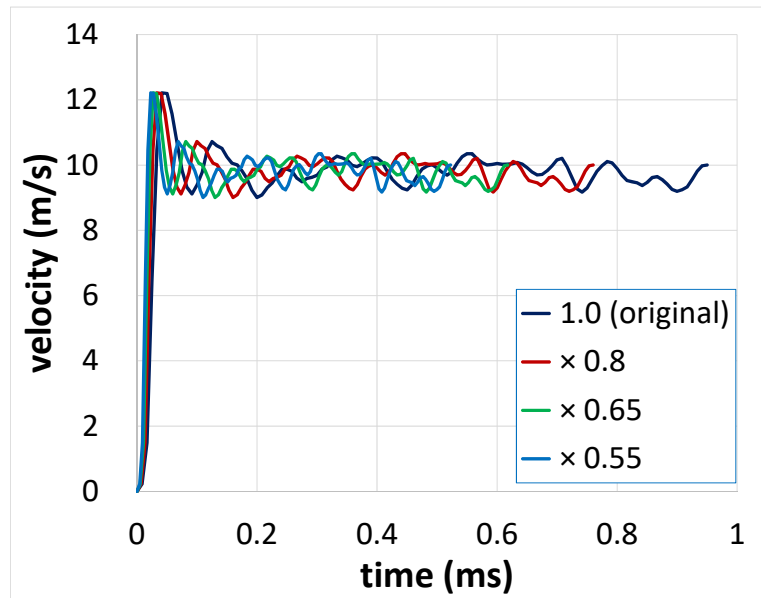
288 Detailed information for the FE simulation.

Analysis mode	Abaqus / Dynamic Explicit		
Element type	Four-node shell quadrilateral element (S4R)		
Mesh size	0.3 mm		
Boundary condition	Velocity boundary condition		
Simulated time	0.47 ms		
Step number	100 steps		
Rate-dependent hardening law	Swift + Johnson–Cook		
	$X_1(X_2 + \varepsilon_p)^{X_3} \left(1 + C \ln \left(\frac{\dot{\varepsilon}_{p,d}}{\dot{\varepsilon}_{p,r}} \right) \right)$		
Material properties			
X_1, X_2, X_3 (MPa)	1430	0.0021	0.157
C	0.03		
$\dot{\varepsilon}_{p,r}$ (s^{-1})	0.001		
Elastic modulus	200 GPa		
Poisson's ratio	0.3		
Density	7,800 kg/m ³		

289

290

291 To mimic the real loading conditions in the experiments, the velocity of the specimen pulling side
292 during the IFHS tensile test was applied as boundary condition to the FE model [27], as shown in
293 Fig. 6 as ‘original’. $\times 0.8$, $\times 0.65$ and $\times 0.55$ correspond to faster loading ramps, this will be explained
294 later.



295

296 **Fig. 6.** Velocity boundary conditions.

297

298 3.2 VFM validation with the Johnson–Cook model

299 The built-in Johnson–Cook model in Abaqus was used. The input coefficients chosen for the FE
300 simulations are listed in Table 2. For the quasi-static flow stress σ_r in Eq. (10), the Swift model in
301 Eq. (14) was selected

$$\sigma_r = X_1(X_2 + \varepsilon_p)^{X_3} \quad (14)$$

302 in which the coefficients were set to known values for the identification.

303

304 The dynamic parameter C only was determined using acceleration as listed in Table 3. The relative
305 error of the identified Johnson–Cook model parameter C was 49.3 % (original elapsed time

306 condition), which led to a large difference with the reference target flow curve determined at 200 s^{-1}
 307 strain rate as presented in Fig. 7. However, the identified result was similar to the input value when
 308 the parameter was evaluated using load information [27].

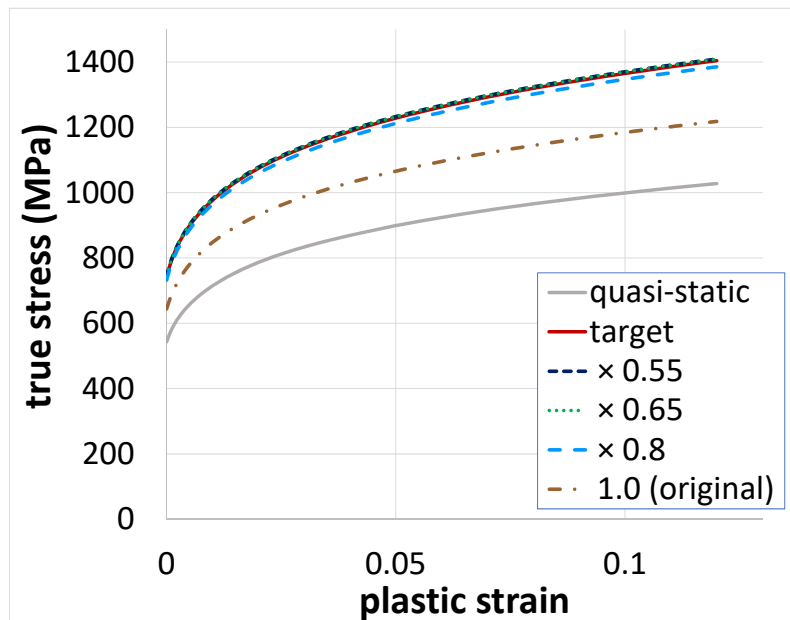
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313 **Table 3**

314 Maximum acceleration magnitude and determined Johnson–Cook parameter using acceleration for
 315 each case (FE simulation, R.E.: relative error).

Elapsed time condition	Maximum acceleration magnitude (km/s^2)	Input	Identified	R.E. (%)
1.0 (original)	463.3		0.0152	49.3
$\times 0.8$	749.8	0.03	0.0285	5.00
$\times 0.65$	981.3		0.0304	-1.33
$\times 0.55$	1147		0.0303	-1.00

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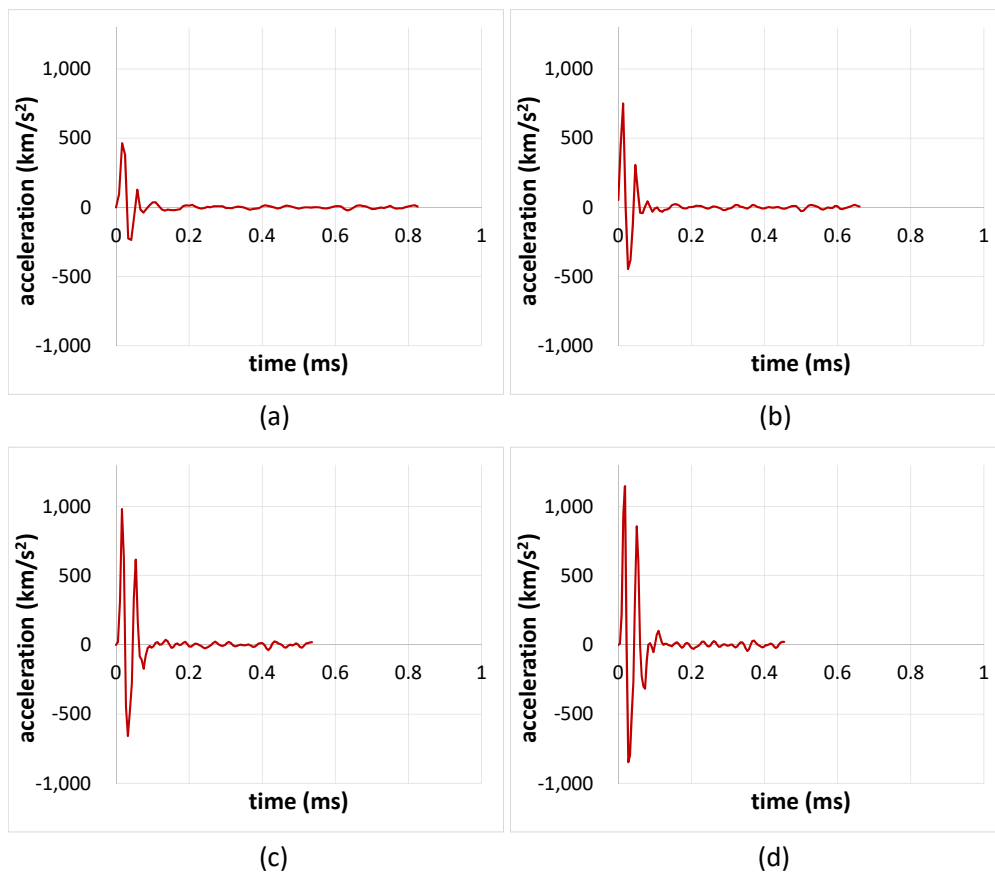
318 **Fig. 7.** Stress–strain curve comparison between the identified results using acceleration and the input
 319 at 200 s^{-1} strain rate.

320

321 A previous study [26] conducted an FE analysis and observed that a minimum threshold of
322 acceleration magnitude was required for proper identification using the VFM. One of the purposes of
323 the simulation in this study was to check the minimum acceleration magnitude required for the VFM
324 identification. Therefore, the total testing time was decreased by 0.8, 0.65, and 0.55 times to increase
325 the magnitude of the acceleration while maintaining the magnitude of the velocity, as shown in Fig. 6.
326 Subsequently, the new velocity boundary conditions were fed to the FE simulations, and the same
327 identification processes were repeated.

328

329 The maximum acceleration magnitude and the evolution of the AOI-averaged acceleration for each
330 case are presented in Table 3 and Fig. 8.

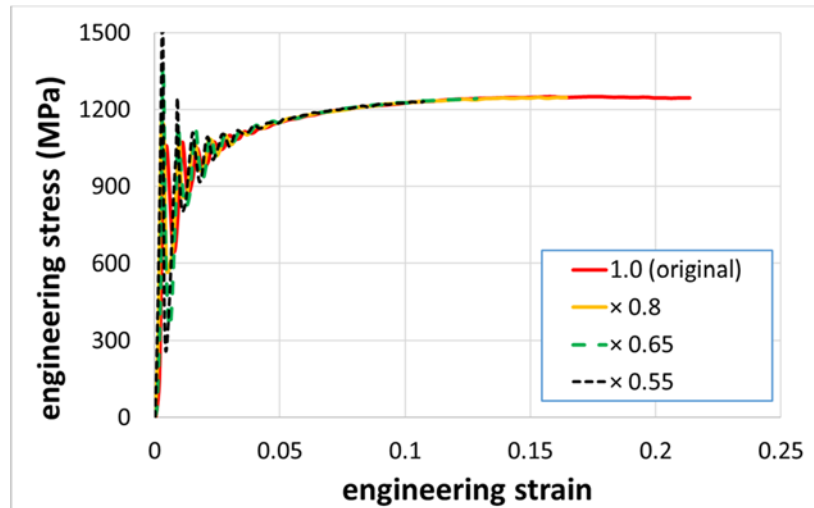


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332 **Fig. 8.** AOI-averaged acceleration evolution: (a) 1.0 (original), (b) $\times 0.8$, (c) $\times 0.65$, and (d) $\times 0.55$.

333 The different velocity boundary conditions resulted in different elongations as shown in Fig. 9. The

334 obtained maximum engineering strains are 0.211, 0.164, 0.128 and 0.108 for 1.0 (original), $\times 0.8$,
335 $\times 0.65$ and $\times 0.55$, respectively where the engineering strain at maximum strength is 0.125. Therefore,
336 it is considered that strain level is appropriate for the identification of Johnson–Cook parameter for
337 all the cases.



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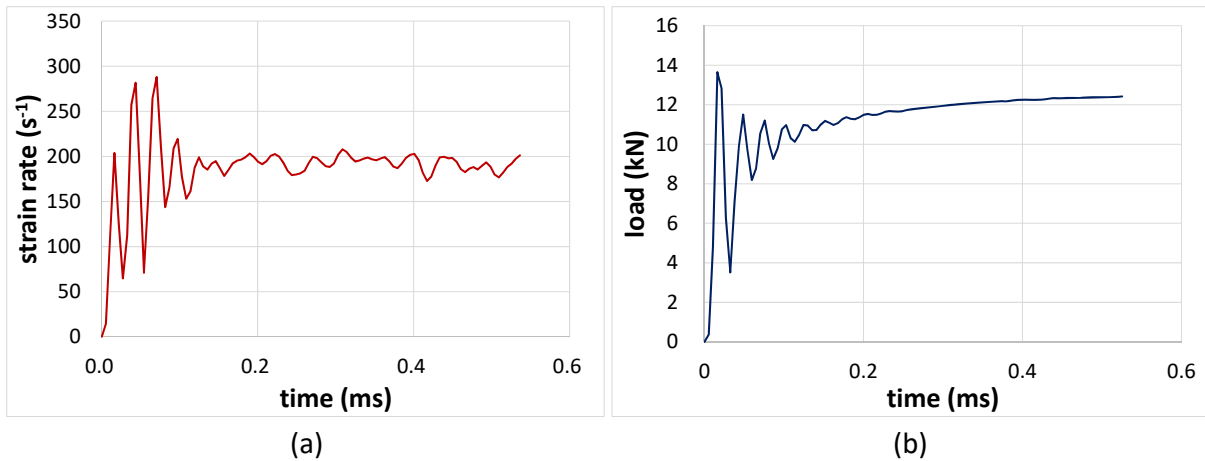
339 **Fig. 9.** Engineering stress-strain curves obtained from different boundary conditions.

340

341 The relative errors of the determined parameters in Johnson–Cook model and the corresponding
342 calculated flow stresses at a strain rate of 200 s^{-1} are presented in Table 3 and Fig. 7, respectively. The
343 Johnson–Cook model parameter was correctly identified using the acceleration only when the
344 maximum acceleration magnitude was approximately $1.0 \times 10^3 \text{ km/s}^2$ ($\times 0.65$ times). Although the
345 parameter obtained with this acceleration ($\times 0.65$ times) was slightly under-predicted (-1.33 %), the
346 difference in flow stress was less than 1 %.

347

348 In addition, the evolution of the AOI-averaged strain rate and measured load in the loading direction
349 (reaction force from the bottom edge in Fig. 5) for the $\times 0.65$ elapsed time condition are shown in
350 Fig. 10.



351

352 **Fig. 10.** Evolution of (a) AOI-averaged strain rate (b) load in the loading direction (FE simulation,
 353 $\times 0.65$ elapsed time condition).

354

355 4. Experiment

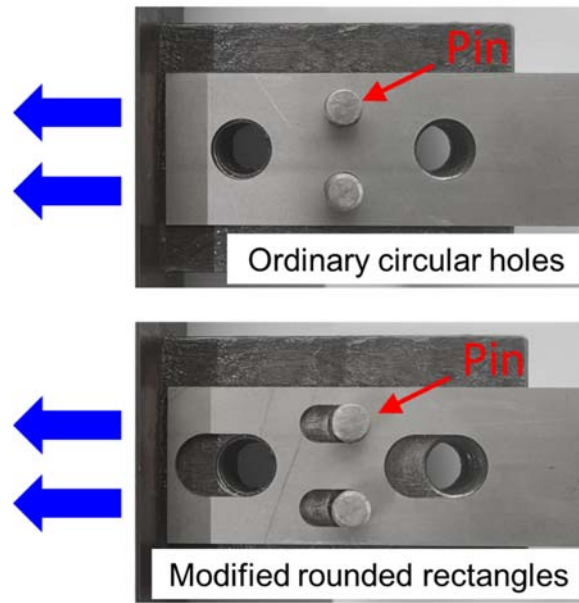
356 4.1 Modification of IFHS tester to increase acceleration

357 As the current study aimed to determine the dynamic properties from acceleration, the magnitude of
 358 acceleration in the IFHS experiments should be sufficiently high to provide information for
 359 identification. Thus, several methods have been considered and applied to increase the acceleration
 360 magnitude.

361

362 4.1.1 Modification of grip condition

363 As shown in Fig. 11, the grip condition was revised in the current test setup [27]. For the ordinary
 364 circular holes, the specimen is pulled in tension from the start with the strain rate increasing gradually
 365 during deformation but, this configuration diminishes the initial acceleration. Since, the original
 366 circular holes are unsuitable for providing a sudden deformation jump, special attention was paid to
 367 modifying the grip conditions. The specimen must initiate its deformation after the pulling
 368 mechanisms reach a certain speed, which was achieved by replacing the circular holes with rounded
 369 rectangles, as shown in Fig. 11. The sliding distance for the modified specimen with rounded
 370 rectangles was set to 6 mm.



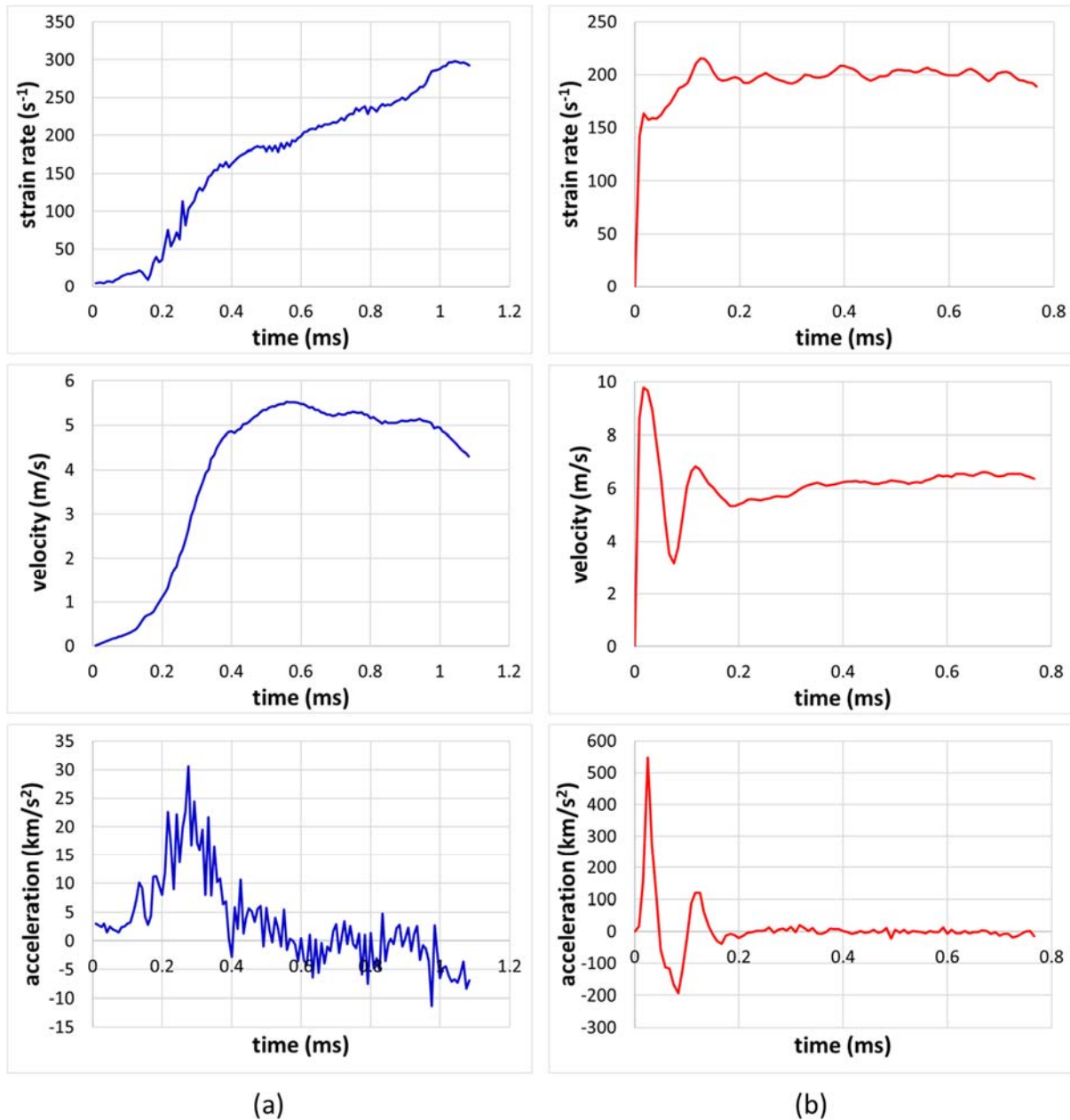
371

372 **Fig. 11.** Modification of the grip condition.

373

374 The AOI-averaged velocity, strain rate and acceleration in the loading direction of the original and
375 modified grips were compared, as shown in Fig. 12. Interestingly, the strain rate after modification
376 increased abruptly and remained comparatively steady and the acceleration magnitude increased
377 remarkably. Notably, the velocity and acceleration were not raised more when the sliding distance
378 was elongated from 6 to 9 mm for the modified grips.

379

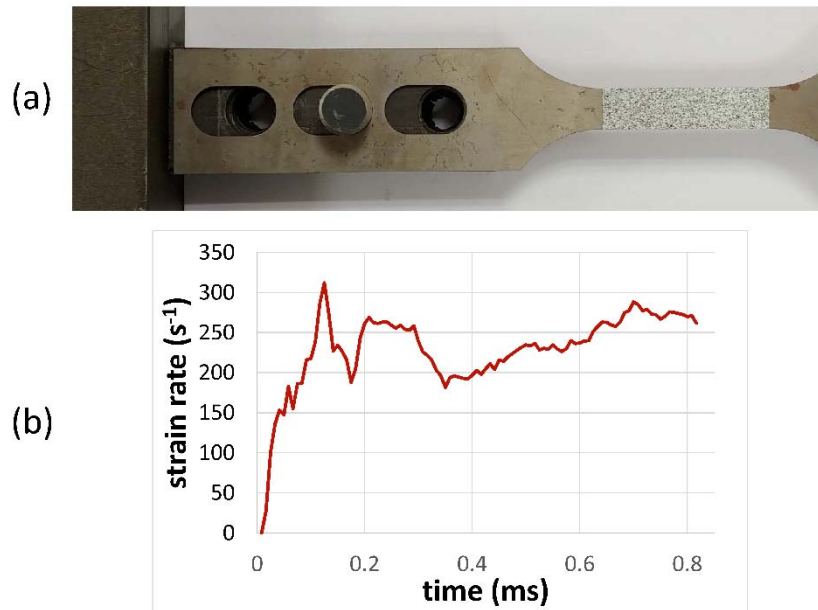


380

381 **Fig. 12.** Comparison of AOI-averaged strain rates, velocities and accelerations (a) circular holes (b)
 382 rounded rectangles.

383

384 In addition, one rounded rectangle was used for another geometrical configuration to raise the
 385 acceleration magnitude, as depicted in Fig. 13(a). The pin diameter was extended from 7 to 12 mm
 386 as it was thought that a stronger strike would raise the acceleration magnitude. However, the resulting
 387 strain rate behavior, Fig. 13(b), exhibited a sudden decrease early on because the stronger strike due
 388 to the high velocity of the pin crumpled the edge of the rounded rectangle.

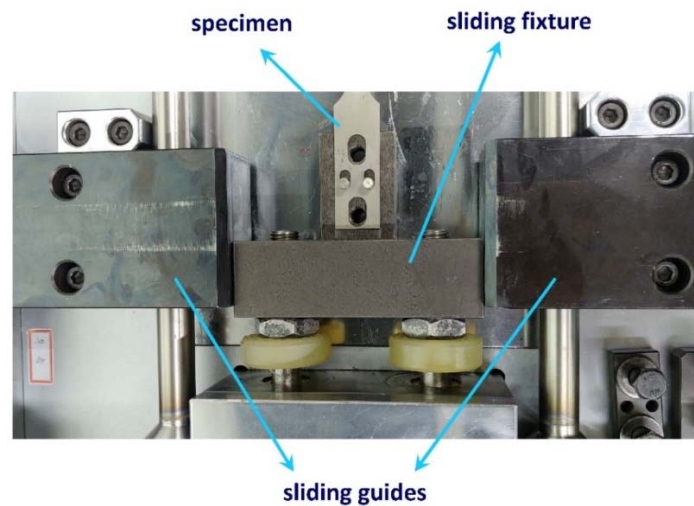


389

390 **Fig. 13.** (a) Another geometrical configuration for the grip condition and (b) the measured strain rate.

391

392 Therefore, a grip configuration with two pins was selected although it may provide additional
 393 vibrations during the dynamic test if the alignment between the specimen and the sliding fixture is
 394 not accurate. Thus, special attention was paid to the alignment using additional sliding guides, as
 395 shown in Fig. 14.



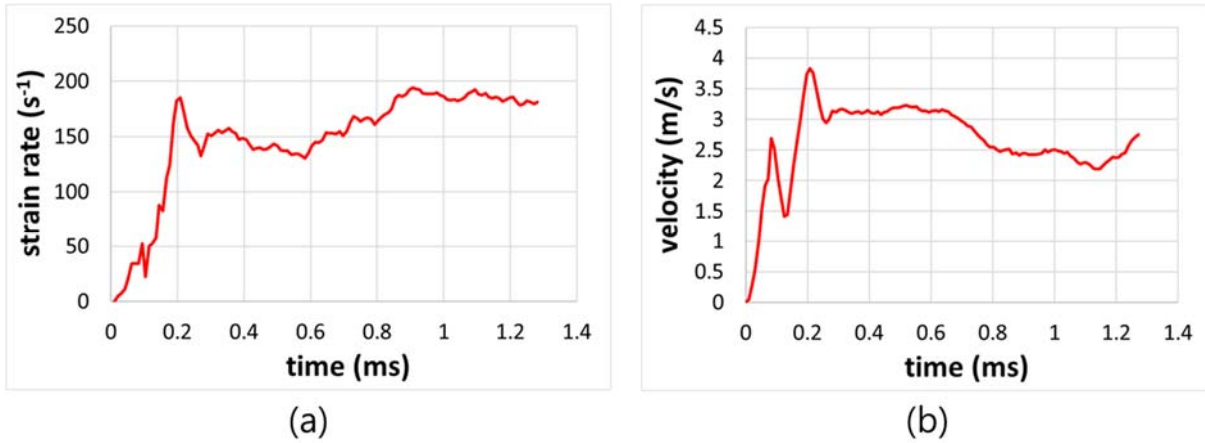
396

397 **Fig. 14.** Sliding guides for the alignment.

398

399 Initially, the moving crosshead in Fig. 2 was made of steel. It was found that the strain rate and
 400 velocity increased rather slowly due to a significant mass of the moving crosshead even after

401 replacing the circular holes with rounded rectangles as shown in Fig. 15. Therefore, the material of
 402 the moving crosshead was changed from steel to duralumin and the structure of the crosshead was
 403 modified to make an empty space inside. Finally, total weight of the moving crosshead was reduced
 404 to 25 % of the original weight, resulting in a sudden deformation jump as shown in Fig. 12.



405
 406 **Fig. 15.** Evolution of (a) AOI-averaged strain rate (b) velocity with an initial crosshead.

407
 408 4.1.2 Increase of elastic strain energy

409 Another approach for increasing the acceleration was to increase the total elastic strain energy. As
 410 explained in Section 2.1, the mechanism of the IFHS tester is based on the energy transformation
 411 from elastic strain energy to kinetic energy. If the amount of elastic strain energy increases in the two
 412 frame bars, the kinetic energy of the specimen also increases.

413
 414 The elastic strain energy (strain energy density \times frame bar volume) stored in each frame bar is

$$\text{elastic strain energy} = \frac{1}{2} \sigma_{bar} \varepsilon_{bar} \times A_{bar} L_{bar} = \frac{1}{2} E_{bar} \varepsilon_{bar}^2 A_{bar} L_{bar} \quad (15)$$

415 where σ_{bar} , ε_{bar} , E_{bar} , A_{bar} , and L_{bar} are the stress, strain, Young's modulus, cross-sectional
 416 area, and length of the frame bar, respectively. The accumulated elastic strain energy in the frame
 417 bars is affected by their length, diameter, and material.

418
 419 When the coupler is broken, the energy in the frame bar is released and converted to kinetic energy

420 (no energy loss and constant velocity of the frame bar are assumed).

$$\text{kinetic energy} = \frac{1}{2} m_{bar} v_{bar}^2 = \frac{1}{2} (A_{bar} L_{bar} \rho_{bar}) v_{bar}^2 \quad (16)$$

421 where m_{bar} , ρ_{bar} , and v_{bar} denote the mass, density, and velocity of the frame bar, respectively.

422 For an accurate calculation, the mass of the additional frame module parts was also considered in
423 m_{bar} .

424
425 v_{bar} is the impact velocity, which can be derived from Eqs. (15) and (16).

$$\text{impact velocity } v_{bar} = \sqrt{\frac{E_{bar}}{\rho_{bar}}} \varepsilon_{bar} \quad (17)$$

426 The initial frame bar was made of steel, and its length and diameter were 1270 mm and 25 mm,
427 respectively. The IFHS tester was initially designed after considering several factors, such as
428 sufficient elastic strain energy stored in the frame bars to break the AHSS specimens and target strain
429 rates [36,37]; however, these factors are not explained here in detail because they are beyond the
430 scope of this study.

431
432 Eq. (17) indicates that the impact velocity is affected by the material type and the strain magnitude
433 of the energy frame bar. The strategy was to increase the impact velocity to obtain a larger magnitude
434 of acceleration. Accordingly, several methods were proposed to increase the impact velocity.

435
436 The first method consisted in changing the material from steel to titanium. For an applied force of
437 400 kN, corresponding to the coupler capacity, the theoretically calculated impact velocities given in
438 Table 4 show that the value is higher for titanium compared to steel. In the calculation of impact
439 velocity, the total mass of the frame bars and additional frame module parts was used. In addition, the
440 expected strain rate in the AOI was calculated based on the following simple approximation:

$$\text{strain rate } \dot{\varepsilon} = \frac{\varepsilon}{t_e} = \frac{\frac{l}{l_0}}{t_e} = \frac{v_{bar}}{l_0} \quad (18)$$

441 where t_e is the elapsed time, l_0 the undeformed gauge length, and l the deformed gauge length.

442

443 **Table 4**

444 Change in impact velocity and strain rate depending on the material (25 mm diameter).

Material	Young's modulus (GPa)	Density (kg/m ³)	Stress (MPa) / strain	Impact velocity (m/s)	Strain rate (s ⁻¹)
Steel	200	7870	408 / 0.0020	6.74	193
Titanium	110	4500	408 / 0.0037	10.3	294

445

446 The second method was to increase the coupler capacity allowing to raise the stress and elastic strains
 447 in the frame bars, thus the amount of stored elastic strain energy in Eq. (15). However, when the
 448 capacity of the coupler was increased from 400 to 500 kN, unexpected damage occurred in other
 449 parts of the multi-components IFHS tester and this method was discarded.

450

451 The third method was to decrease the diameter of the two energy frame bars to induce a higher impact
 452 velocity. When the applied force was the same, the stress and strain levels of the frame bars increased
 453 with decreasing diameter, resulting in an increase of the impact velocity, as shown in Eq. (17).
 454 Therefore, the diameter of the frame bars, initially 25 mm, was reduced to 20 mm. Table 5 compares
 455 the theoretically calculated values when the diameter of the frame bar was changed from 25 to 20 mm.
 456 With a diameter reduction of 5 mm, the elastic strain energy increased significantly enough to change
 457 the impact velocity from 10.3 m/s to 13.8 m/s.

458

459 Although the impact velocity theoretically increases with the amount of kinetic energy, it may not
 460 translate into acceleration, which is the slope of the velocity–time graph. This aspect should be
 461 verified experimentally.

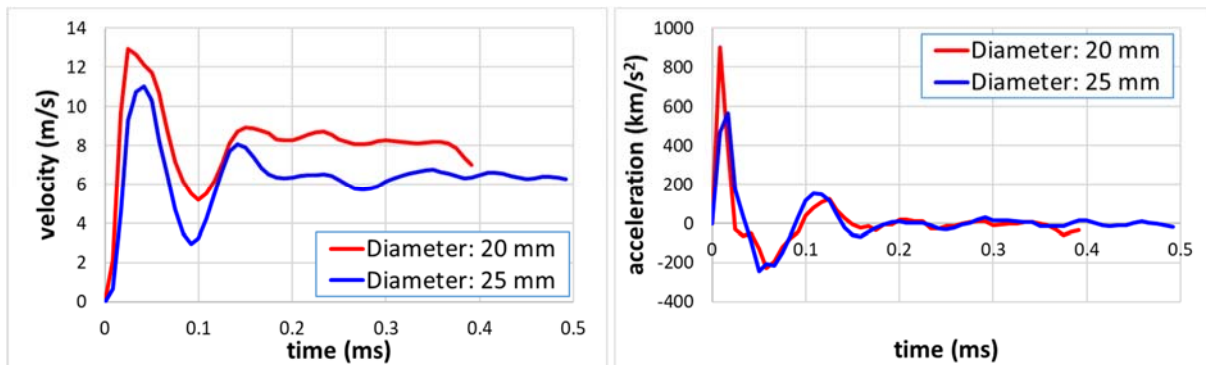
462 **Table 5**

463 Change in impact velocity depending on the diameter (Titanium bar).

Diameter of frame bar	25 mm	20 mm
Coupler capacity	400 kN	
Stress / strain	408 MPa / 0.0037	637 MPa / 0.0058
Strain energy density	755 kJ/m ³	1,850 kJ/m ³
Total strain energy	941 J	1473 J
Impact velocity	10.3 m/s	13.8 m/s

464

465 For this purpose, a 1.2 mm thick dual-phase DP980 steel sheet specimen was investigated. Dynamic
466 tensile tests were carried out with the IFHS tester using the frame bars of diameters 25 and 20 mm.
467 The AOI-averaged velocity and acceleration were calculated from the displacements and are shown
468 in Fig. 16. The average velocity increased from approximately 6 to 9 m/s, and the maximum
469 acceleration increased by approximately 300 km/s², from 568 to 899 km/s², after the change in
470 diameter.



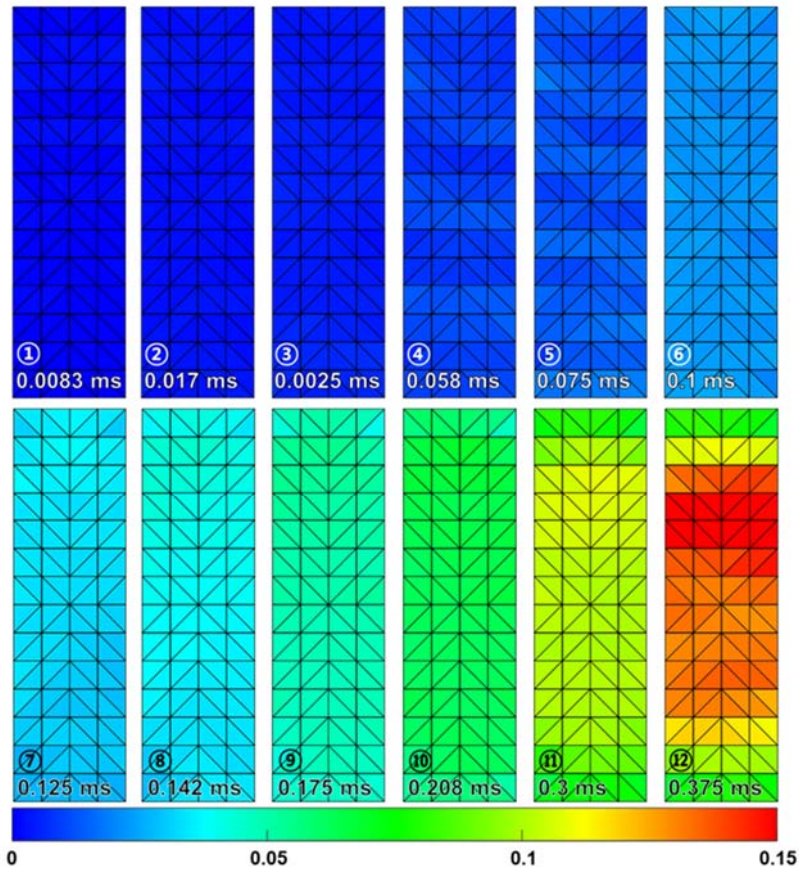
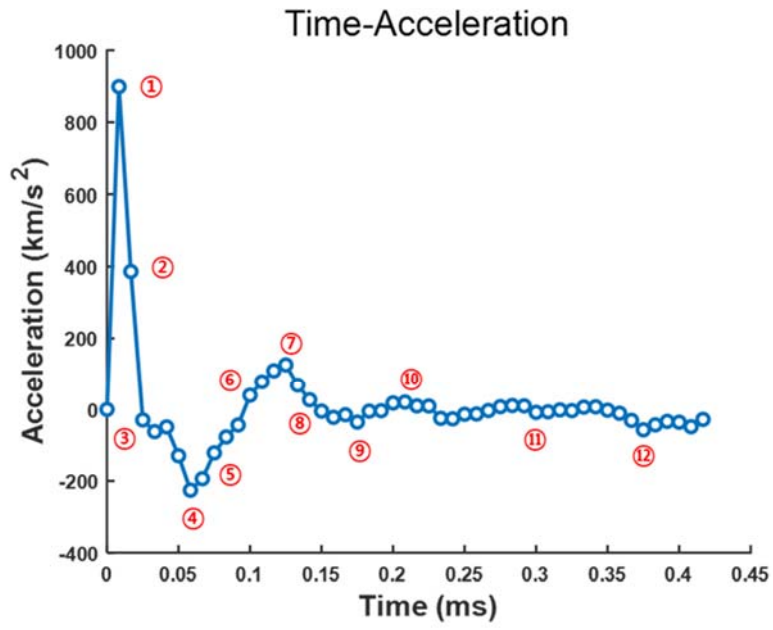
471

472 **Fig. 16.** Comparison of measured velocity and acceleration between the diameters of 25 mm and
473 20 mm.

474

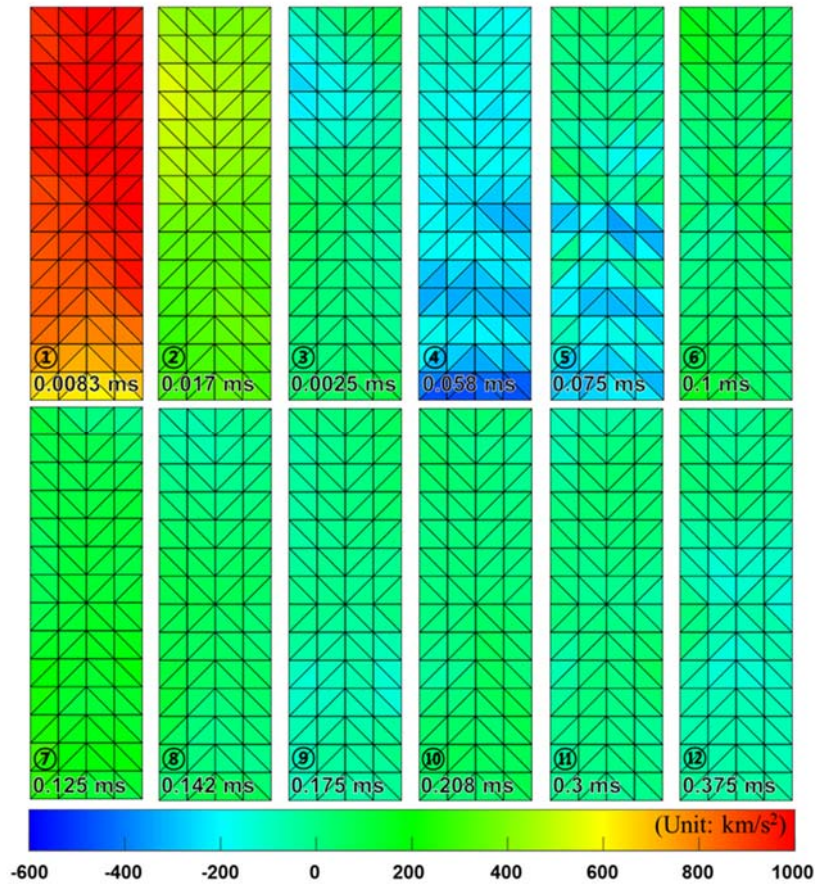
475 As observed from the simulation results, the Johnson–Cook model parameter was identified correctly
476 using only acceleration when the maximum acceleration magnitude was approximately
477 1.0×10^3 km/s². Interestingly, the experimental maximum acceleration magnitude for the diameter of
478 20 mm was close to the target maximum acceleration magnitude from the simulation.

479 In addition, the evolution of strain and acceleration distribution over the area of interest of the
480 specimen for the diameter of 20 mm is shown in Figs. 17, 18, and 19. The distribution is shown in
481 undeformed configuration for the sake of comparison. Each contour map corresponds to the step with
482 a number shown in the acceleration curve in Fig. 17. It was found that the acceleration distribution is
483 very heterogeneous over the AOI and local acceleration magnitude is higher than $1,000 \text{ km/s}^2$ in some
484 triangular elements. The acceleration information starts to disappear at around 0.15 ms. It is worth
485 noting here that the average strain reaches about 4.5 % at around 0.15 ms.



486

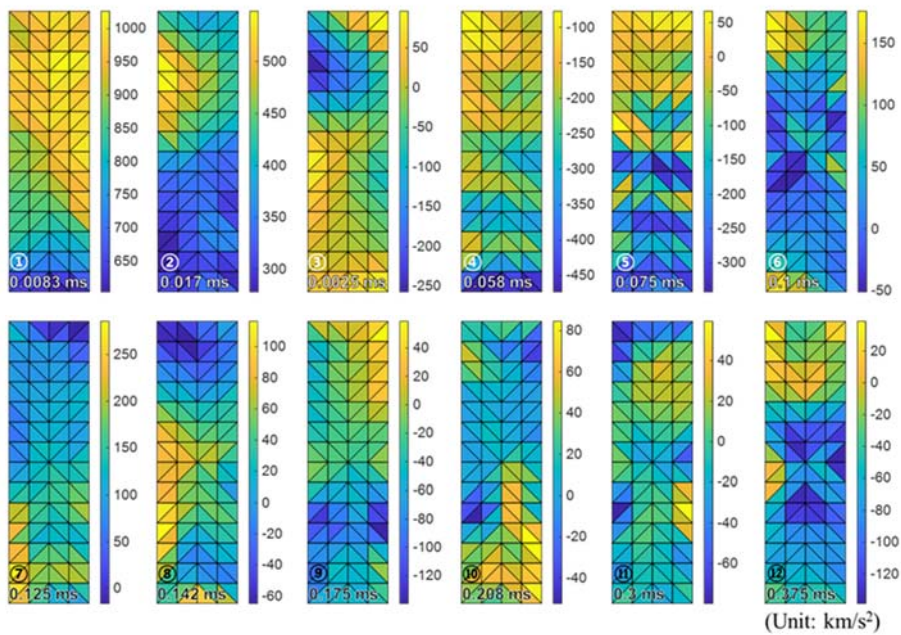
487 **Fig. 17.** Evolution of strain distribution over the area of interest for the diameter of 20 mm.



488

489 **Fig. 18.** Evolution of acceleration distribution over the area of interest for the diameter of 20 mm

490 (with the same scale bar).

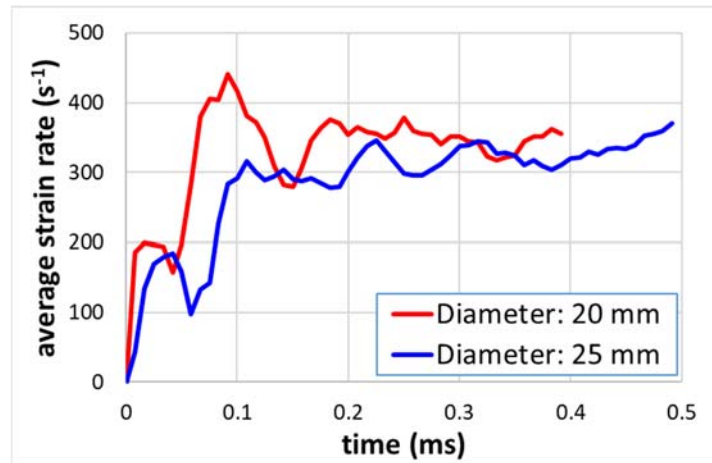


491

492 **Fig. 19.** Evolution of acceleration distribution over the area of interest for the diameter of 20 mm

493 (with different scale bars).

494 The average strain rate for the bar diameter of 20 mm also increased (to approximately 350 s^{-1})
495 compared with that for the diameter of 25 mm (approximately 300 s^{-1}), as shown in Fig. 20. The
496 decrease in strain rate at around 50 microseconds is probably due to the dynamics of the contact
497 between the pin and rounded rectangle (loss of contact for a small time, like a rebound).



498
499 **Fig. 20.** Comparison of average strain rate between the diameters of 25 mm and 20 mm.

500
501 It is worth noting here that the acquisition frame rate is important for capturing the deformation in a
502 high-speed dynamic test. In this study, a rate of 120,000 frames per second (fps) was selected with a
503 spatial sampling of 640×128 pixels. However, based on Fig. 16, this rate appeared insufficient to
504 define the peak velocity accurately. Therefore, the actual maximum acceleration might have been
505 higher than the apparent measured value. Since the important acceleration data is provided in the
506 initial deformation stage of very short duration in the IFHS test, the high-speed camera fps should be
507 increased as much as possible to prevent information loss [38]. However, an increase of the fps value
508 is accompanied by a decrease of the spatial sampling owing to the characteristics of the high-speed
509 camera which, in turn, results in a higher noise level affecting the identification. Therefore, a
510 compromise between the spatial and temporal resolutions is required.

511
512 **4.2 Standard quasi-static uniaxial tensile tests**

513 Quasi-static uniaxial tension tests were performed with ASTM E8 tensile specimens. Engineering

514 stress–strain curves were obtained at a strain rate of 0.002 s⁻¹. The 0.2 % offset method was applied
 515 to calculation of the true stress–plastic strain curves σ_r in Eq. (10). The curves were fitted using
 516 various hardening laws and the best turned out to be a combination of the Swift and modified Voce
 517 hardening [27].

$$\sigma_r = X_1(X_2 + \varepsilon_p)^{X_3} + X_4 + X_5\varepsilon_p + X_6(1 - \exp(-X_7\varepsilon_p)) \quad (19)$$

518
 519 The hardening parameters, determined through a standard optimization procedure, are listed in
 520 Table 6. A maximum uniform plastic strain of 4.1 % was achieved for the DP980 steel sheet.

521
 522 **Table 6**

523 Hardening parameters of DP980 at quasi-static (units for X₁, X₄, X₅, X₆: MPa).

X ₁	X ₂	X ₃	X ₄	X ₅	X ₆	X ₇
640.4	0.0001022	0.1328	530.9	101.4	143.5	149.9

524
 525 4.3 Acquisition of load data in the dynamic tests

526 For load acquisition, strain gauges were bonded to the grip area of the specimen, where the
 527 deformation was linearly elastic. As shown in Fig. 3, the grip end at the fixed side is longer than the
 528 other grip end to provide space for strain gauges and this area is also wider to ensure that only elastic
 529 deformation occurs during the test. In order to measure the pure material response at the strain gauges
 530 attached on the specimen, the plastic stress wave front generated in the specimen gauge area should
 531 reach the strain gauges before the reflected elastic stress wave front at the end of transmitter bar
 532 arrives at the strain gauges [28,36]. The length of the transmitter bar was determined based on a
 533 calculation method in [28] to avoid the effect of the reflected stress wave. The calculated minimum
 534 length of transmitter bar was 275 mm in this study and the length of 600 mm was chosen finally for
 535 the sake of safety.

536
 537 Vibration of the specimen was observed in the normal direction to the sheet plane. Therefore, strain

538 gauges were bonded on each side of the specimen (front and back) to eliminate the bending effects
539 [27]. The elastic strain was determined by averaging the voltage signals of the two gauges. The load
540 F was calculated on the basis of Hooke's law and the cross-sectional area

$$F = AE\varepsilon \quad (20)$$

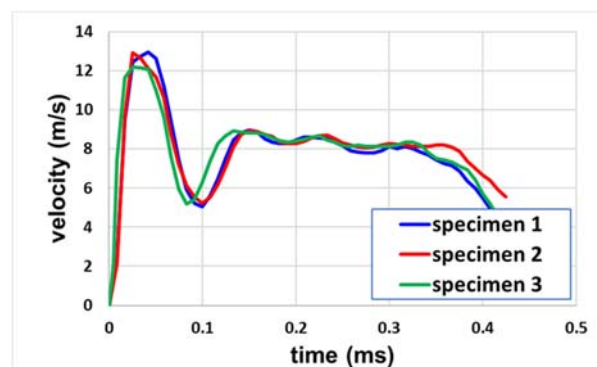
541 with a Young's modulus E of 209.7 GPa for DP980, as determined from the quasi-static condition.
542 Synchronized load data was acquired from the strain gauges using a data acquisition (DAQ) system
543 when speckle pattern images were taken by a high-speed camera.

544

545 4.4 Experimental results with the VFM

546 Dynamic tensile tests were conducted with three DP980 specimens for each case (bar diameters of
547 25 and 20 mm). To check the reproducibility, the measured velocity profiles as a function of time of
548 the three specimens for 20 mm bar diameter are shown in Fig. 21. It can be seen that the
549 reproducibility is reasonable, so the specimen data with the highest maximum acceleration was used
550 for the VFM identification for each case. All the experimental data presented previously was fed into
551 the VFM routines to identify the Johnson–Cook dynamic parameter C . The value of the initial
552 estimate to initiate the identification process was found to be irrelevant to determine the global
553 optimum within the interval [0.0001, 0.3]. The dynamic coefficient identification took less than 5
554 minutes.

555



556

557 **Fig. 21.** Velocity profiles as a function of time of three specimens for 20 mm bar diameter.

558

559 In the case of the 25 mm bar diameter, the identification using the acceleration data only was not
 560 successful owing to the insufficient acceleration magnitude. The identified flow curve with
 561 acceleration at the strain rate of 300 s^{-1} led to flow stresses close to the quasi-static curve. For 20 mm
 562 bar diameter, the load and acceleration data were obtained simultaneously from the same experiment
 563 but the Johnson–Cook coefficient was identified using the acceleration data only or the load data only.
 564 The parameters identified with the two methods are listed in Table 7. The curve obtained with the
 565 acceleration data at the strain rate of 300 s^{-1} was compared with the curve from the load data in Fig.
 566 22. It can be observed that the two curves are almost identical.

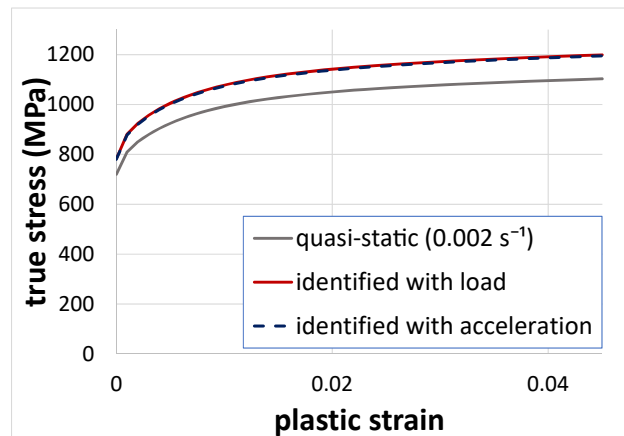
567

568 **Table 7**

569 Identified Johnson–Cook parameters (for the diameter of 20 mm).

From load only	From acceleration only
0.0069	0.0066

570



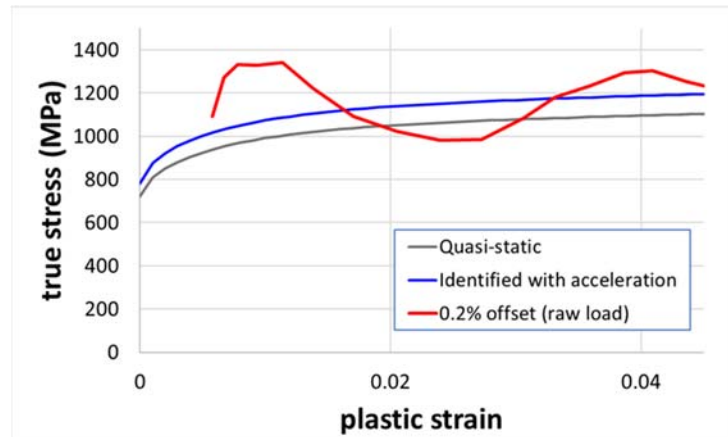
571

572 **Fig. 22.** Identified stress–strain curves at the strain rate of 300 s^{-1} (for the diameter of 20 mm).

573

574 The curve obtained with the acceleration data at the strain rate of 300 s^{-1} was compared with the curve
 575 from the load data using the VFM in Fig. 22. It can be observed that the two curves are almost
 576 identical. Also, comparison with the curve obtained from the raw load data using 0.2 % offset method
 577 (similar in a quasi-static test) is shown in Fig. 23. Though the stress-strain curve from the raw load

578 data is a bit fluctuating, two curves are in reasonable agreement.

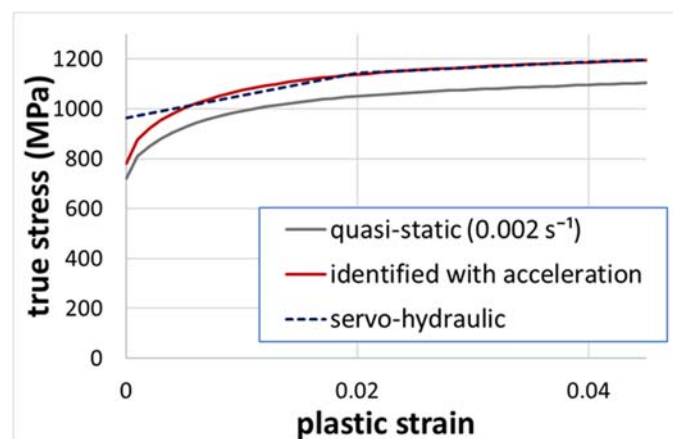


579

580 **Fig. 23.** Identified stress–strain curves at the strain rate of 300 s^{-1} (for the diameter of 20 mm) and
581 comparison with the stress-strain curve from the raw load data.

582

583 In addition, for further validation of the identified Johnson–Cook parameter using acceleration
584 information, the determined curve with Johnson–Cook model at the strain rate of 200 s^{-1} is compared
585 with the true stress-plastic strain curve obtained from an Instron servo-hydraulic high-speed tensile
586 tester at the strain rate of 200 s^{-1} in Fig. 24 (POSCO provided the curve data). Though the initial
587 region including the initial yield stress shows some discrepancy because a load curve smoothing
588 technique was applied to the servo-hydraulic tester data, it can be seen that the general trend of flow
589 stress is very similar to each other which indicates that the identification result obtained using the
590 VFM with the acceleration data is reliable.



591

592 **Fig. 24.** Comparison of stress–strain curves at the strain rate of 200 s^{-1} between the current method
593 and a servo-hydraulic high-speed tensile tester.

594

595 5. Limitations

596 In this section, some limitations of the current approach will be discussed.

597

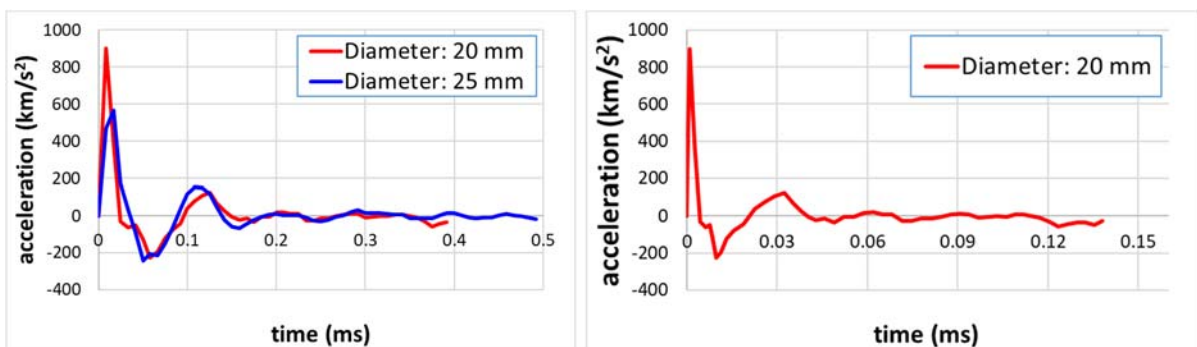
598 5.1. Duration time of acceleration information

599 In this study, various techniques have been applied to increase the acceleration during the experiments
600 because it was found from FE simulations that a critical magnitude of acceleration is required for the
601 identification using the VFM. The maximum acceleration increased significantly up to 899 km/s²
602 after applying various techniques in the experiments, leading to successful identification of Johnson–
603 Cook parameter. However, it should be noted that not only sufficient acceleration magnitude but also
604 sufficient duration time of acceleration is important for the identification.

605

606 The acceleration information starts to disappear at around 0.15 ms for the diameter of 20 mm in
607 Fig. 25(a), indicating no information is present for the identification after 0.15 ms, which may result
608 in poor identifiability. In this study, an advanced high-strength steel, DP980 was chosen and the
609 maximum strain (at maximum strength) of DP980 at quasi-static is around 4~5 %. As can be seen in
610 Fig. 25(b), the strain level is around 4.5 % at around 0.15 ms. Therefore, it is considered that the
611 duration time of acceleration is sufficient for the identification of strain rate dependence of DP980 in
612 this study.

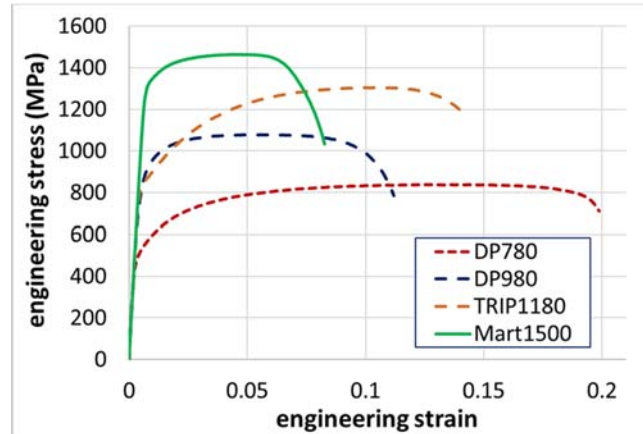
613



614 **Fig. 25.** Evolution of the acceleration (a) as a function of time (b) as a function of strain for the
615 diameter of 20 mm.

616

617 Nonetheless, there are various types of advanced high-strength steel which have longer elongation as
618 shown in Fig. 26. Therefore, in order to provide good identifiability, methods of increasing the
619 duration time of acceleration information will have to be investigated in more depth.



620

621 **Fig. 26.** Engineering stress-strain curves of various AHSSs at quasi-static.

622

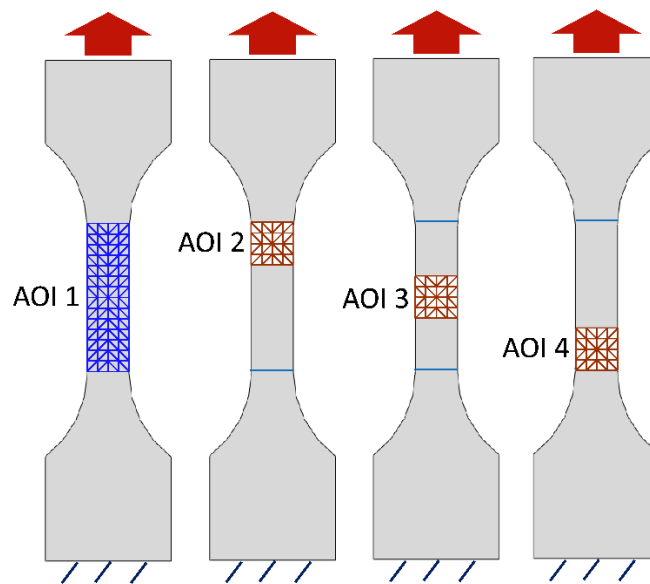
623 5.2. Spatial and temporal resolution

624 As described in section 4.1.1, temporal and spatial resolutions are important parameters affecting
625 precision of the current approach. In this study, a rate of 120,000 fps was selected with a spatial
626 sampling of 640×128 pixels. It was observed that the current fps appeared insufficient to measure
627 the maximum acceleration accurately. This issue can be overcome with the usage of an ultra-high-
628 speed camera, which is very costly.

629

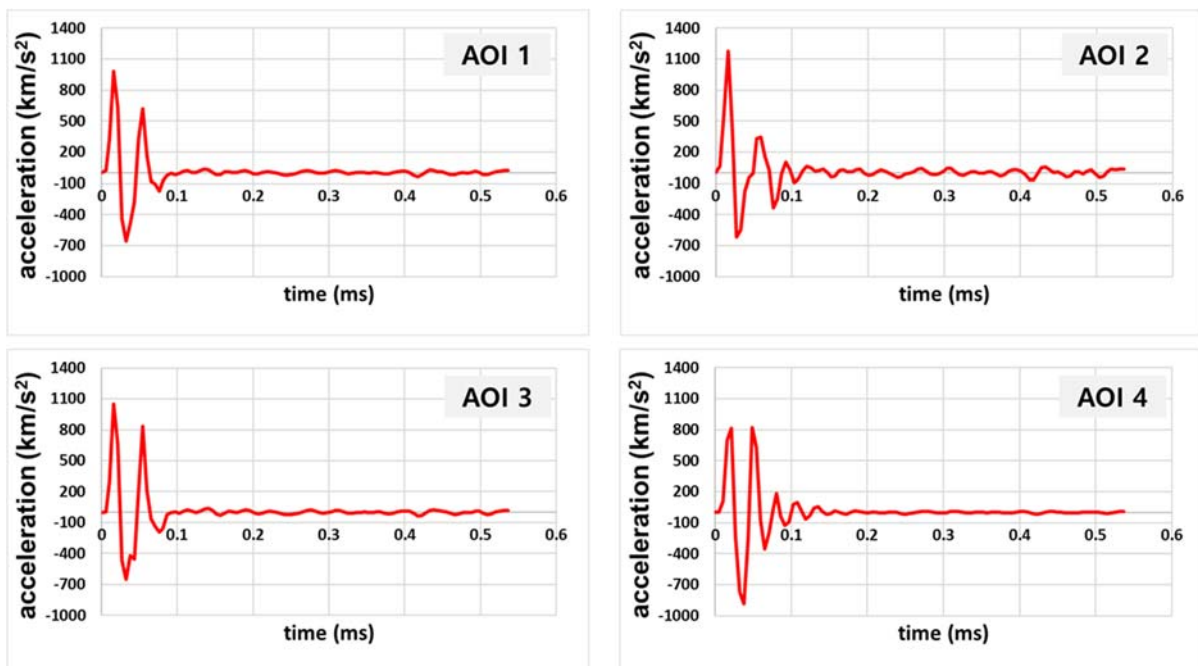
630 Therefore, a compromise between the spatial and temporal resolutions was obtained with the current
631 experimental set-up including the high-speed camera. However, an optimal setting for the spatial and
632 temporal resolutions is not straightforward because several factors should be considered such as
633 maximum acceleration related to temporal resolution, strain rate information and experimental noise
634 level related to spatial resolution. The goal of this study was to determine the strain rate dependence
635 at intermediate strain rates using acceleration information. If the purpose is just to increase the
636 acceleration magnitude, the AOI size can be reduced with a higher frame rate and smaller spatial

637 sampling size as shown in Fig. 27. For the case of $\times 0.65$ times in section 3.1, indeed, the maximum
638 acceleration increased in the cases of AOI 2 and AOI 3 as shown in Fig. 28. However, none of the
639 three cases (AOI 2, AOI 3, and AOI 4) resulted in correct identification. As discussed in [27], rich
640 strain rate information should be provided to identify strain rate dependence with a proper rate
641 dependent model using the VFM. It is considered that strain rate information is not sufficient for
642 determination of strain rate dependence in the three cases.



643
644 **Fig. 27.** Different AOI size and location.

645



646
647 **Fig. 28.** AOI-averaged acceleration evolution for the case of $\times 0.65$ in section 3.1.

648

649 Current spatial and temporal resolutions were chosen based on trial and error. It should be investigated
650 further to make optimal settings for spatial and temporal resolutions in the current measurement
651 technique more systematically. This issue can be handled using a simulator as presented in [39].

652

653 5.3 Control of coupler capacity

654 As explained in section 2.1, the current IFHS tensile equipment utilizes the elastic strain energy stored
655 in frame bars to generate high-speed impact pulses. The accumulated elastic strain energy is
656 controlled by the coupler capacity. Though the reproducibility of couplers is good as shown in Fig. 21,
657 it is very tricky to control the coupler capacity accurately, thus speed, strain rate, and acceleration.
658 The coupler has the form of a cylinder with a notch at the center, so the coupler capacity is determined
659 by the notch depth. Therefore it is required to measure the coupler capacity by changing the notch
660 depth based on trial and error. An alternative would be electromagnetic devices which can replace the
661 coupler.

662

663 6. Conclusion

664 The stress–strain information from intermediate strain rates is essential input to obtain reliable crash
665 simulation results. However, the load applied is not easy to measure accurately at intermediate or high
666 strain rates, owing to the inertial effect. A new methodology was applied in this study, to obtain the
667 dynamic curve of a sheet metal specimen at an intermediate strain rate without measuring the loads.
668 This study utilized the VFM to obtain the dynamic parameters of the Johnson–Cook model with
669 acceleration information. A modified high-speed tensile equipment for steel sheet specimens was
670 devised to increase the acceleration magnitude, and the dynamic behavior was characterized using a
671 DIC technique with a high-speed camera. The stress–strain curve of a dual-phase DP980 steel sheet
672 at an intermediate strain rate of 300 s^{-1} was obtained from the acceleration information. The main
673 observations of this study are as follows:

674

675 (1) Before the actual experiments, FE simulations using Abaqus were performed to determine the
676 feasibility of the identification from the acceleration data. To check the minimum acceleration
677 magnitude required for the identification, the velocity boundary condition extracted from real
678 experimental results with the IFHS tester was adjusted to increase the acceleration magnitude. The
679 identification was successful with the current test configuration when the maximum acceleration
680 magnitude was approximately $1.0 \times 10^3 \text{ km/s}^2$.

681

682 (2) A modified high-speed tensile equipment for steel sheet specimens was developed to increase the
683 magnitude of the acceleration during the experiments. Various techniques have been applied to
684 increase the acceleration. It was found that the most reliable method to increase the stored elastic
685 strain energy, based on the mechanism, was to employ titanium frame bars of 20 mm diameter. The
686 higher strain energy increased the impact velocity, resulting in a higher acceleration. The maximum
687 acceleration magnitude increased from 568 km/s^2 to 899 km/s^2 .

688

689 (3) The identification of the Johnson–Cook parameter with the acceleration was not successful when
690 the energy frame bars with 25 mm diameter were used because of insufficient acceleration. However,
691 the stress–strain curve based on the acceleration data was very close to the curve obtained from the
692 load for the diameter of 20 mm at 300 s^{-1} strain rate, owing to the increased acceleration magnitude.
693 This indicates that the identification result obtained using the VFM with acceleration is reliable.

694

695 Acknowledgment

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